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Production Version of the Extended NASA-Langley
Vortex Lattice FORTRAN Computer Program -
Volume II - Source Code

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April 1982



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Space Administration

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PRODUCTION VERSION OF THE
EXTENDED NASA-LANGLEY VORTEX LATTICE
FORTRAN COMPUTER PROGRAM
VOL. II SOURCE CODE

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1 ABSTRACT / SUMMARY

This document presents the source code for the latest production version, MARK IV, of the NASA - Langley Vortex Lattice Computer Program. All viable subcritical aerodynamic features of previous versions have been retained. This version extends the previously documented program capabilities to four planforms, 400 panels, and enables the user to obtain vortex-flow aerodynamics on cambered planforms, flow field properties off the configuration in attached flow, and planform longitudinal load distributions.

2 INTRODUCTION

The NASA - Langley Vortex Lattice FORTRAN Program (VLM) is designed to estimate the subsonic aerodynamic characteristics of up to four complex planforms. The concepts embodied in this program are mostly detailed in references 1,2 and 3; this document is intended to serve as an update to these references for users and computer specialists who have an interest in implementing this program on their computers. Basically, the VLM Program is a segmented program designed to run on the Control Data Corporation (CDC) computers with the NOS operating system. This program requires a run-time field length of approximately 130K (octal) words of memory, and uses the Langley Research Center

Graphics Output System in FORTRAN (LRCGOSF), along with numerous CDC routines that provide random access of mass storage files. The User's Guide associated with this program is listed as reference 4.

Use of trade names or names of manufacturers in this report does not constitute an official endorsement of such products or manufacturers, either expressed or implied, by the National Aeronautics and Space Administration.

The main program, WINGAL, forms the "root node" to the tree. Each of the seven major branches, or nodes, has a specific function in the overall computational process, and some of these branches consist of more than one routine (this will be explained in the next section). The point of this tree diagram is that it constitutes the way the program is loaded into memory by the operating system, and hence, the way it executes for any given input set. The actual file of loader directives file based on this structure will be discussed in Section 4.

3.2 FUNCTION BY NODE

The root node consists of the main program (WINGAL) and four subroutines. The main program is used to declare all files and common blocks, as well as direct the overall processing done by this code. The four subroutines contained in this node are LOADING, FTLUP, INFSUB and READIN. LOADING is not called by any routine in VLM; its sole purpose is to force the loading of the CDC mass storage routines at this level so they are accessible to all higher level nodes that require them. FTLUP and INFSUB, however, are used by several of the higher nodes, and hence, must be in memory at all times. FTLUP is a linear interpolation routine that uses the CDC Fortran routine LOCF to determine the absolute location (address) of a variable. READIN is used to read in and print out the input data, with line numbers, for the user's reference. The common blocks declared in WINGAL serve as

the principal method of information transfer between the higher nodes; they are considered "global" and must also remain in memory at all times. The higher nodes in the tree are moved in and out of central memory by the loader when they are called by the main program, and this overlapping results in considerable reduction of central memory requirements. The function and associated routines of the higher nodes are as follows:

3.2.1 GEOMETRIC COMPUTATIONS

This node consists of two subroutines, GEOMETRY and PLNPLT. Subroutine GEOMETRY is called by WINGAL to determine, from the input data, the geometry of the configuration. GEOMETRY then makes a call to PLNPLT, which, in turn, produces the "printer plot" of the configuration.

3.2.2 AERODYNAMIC EQUATIONS MATRIX SOLVER

This node consists of six routines and performs the Given's Method of Matrix Solution to determine the circulation terms of the horseshoe vortices; i.e., these routines solve the matrix of the basic linear aerodynamic equations. The routines in this node are:

1. MATXSOL- Called by WINGAL, MATXSOL is the main routine in this node and generates the elements in the aerodynamic influence

coefficient matrix. MATXSOL then calls routine GIVENS to effect a solution.

2. GIVENS - Called by MATXSOL to partition the work storage arrays into rows and columns.
3. BLOCKR - Called by GIVENS to compute the size of, and the number of rows in each partition of the triangularized matrix.
4. TRIANG - Called by GIVENS to triangularize the augmented matrix using planar rotations.
5. SOLVER - Called by TRIANG to perform the back substitution on the matrix and overstore the results back onto mass storage.
6. BUFFIN - Called by TRIANG to transfer data into the work array from mass storage.

3.2.3 AERODYNAMIC ANALYSIS

This node consists of five routines and is used to compute the linear aerodynamic characteristics of the configuration. These routines are:

1. AERODYN - Called by WINGAL, AERODYN is the main routine in this node. AERODYN computes the linear aerodynamic lifting pressures and overall forces and moments.
2. FLOWFL - Determines the flow field characteristics off the wing. FLOWFL reads the field line definition when the flow field data is required. FLOWFL also uses routine FTLUP.
3. HEAPSRT - Called by FLOWFL to sort the X-Y values of the panel centers. FLOWFL compares these values to the X-Y locations of the flow line points in determining the relative position of the flow line with respect to any given planform. HEAPSRT uses a Heap Sort algorithm, which is described in reference 5.
4. SIFT - Called by HEAPSRT to swap values in the sorting process.
5. CDICLS - Computes the far-field induced drag for simple configurations (wing-body) with no dihedral.

3.2.4 NEAR-FIELD INDUCED DRAG SOLUTIONS

This node consists of the single routine, CDRAGNF, (called by WINGAL) that computes the near-field induced drag values. CDRAGNF uses routine FTLUP in determining the near-field chord force properties.

3.2.5 SIDE-EDGE SUCTION ANALYSIS

This node consists of two routines, TIPSUCT and WRTANS. The principal routine, TIPSUCT, called by WINGAL, computes the side-edge force and values for KV_{se} for each planform. WRTANS is then called by TIPSUCT to compute the values of K_p and KV_{le} for each planform. If the configuration is cambered/twisted so that the option for vortex flow computation on warped wings is exercised, WRTANS is not called and K_p and KV_{le} are not calculated. In this situation, TIPSUCT makes numerous calls to the LRCGOSF routines for producing the graphics output. This section of TIPSUCT will have to be revised considerably by users if the VLM program is installed on any other computer. The LRCGOSF routines used by VLM are listed in the Appendix A.

3.2.6 VORTEX FLOW ANALYSIS

This node consists of the single routine, VORTEX, called by WINGAL, to perform the vortex flow analysis for cambered configurations. VORTEX also uses routine FTLUP.

3.2.7 LONGITUDINAL LOAD DISTRIBUTION

This node consists of 18 routines, the principal one being CNLONG, that called by WINGAL. These routines compute the longitudinal load distribution of the configuration. Most of them are associated with the Delta Cp or Net Cp surface interpolation from constant Y to constant X values. They are (by name only) as follows:

INTERP	SUTS
IQHSCV	CURV1N
IQHSD	CURV12
IQHSE	CURVI
IQHSF	CEEZ
IQHSG	TERMS
IQHSH	SNHCSH
UERTST	INTRVL
UGETIO	

The routines beginning with the letters "IQ" provide smooth curve fitting over randomly distributed data points. IQHSCV calculates the interpolating function that is a fifth degree polynomial, and is continuous and has continuous first order partial derivatives. This

technique is detailed in reference 6. Routines IQHSD through UGETIO are support routines for this process. Routines SUTS through INTRVL are used to evaluate the the integral of the spline approximation to a function of a single variable. These routines were adapted from current routines on the NASA - Langley Research Center FORTRAN Math Library which is documented in reference 7. This documentation is available from the NASA - LaRC Analysis and Computation Division (ACD) User Support Office.

3.3 FILES

There are nine files associated with the VLM program and these are as follows:

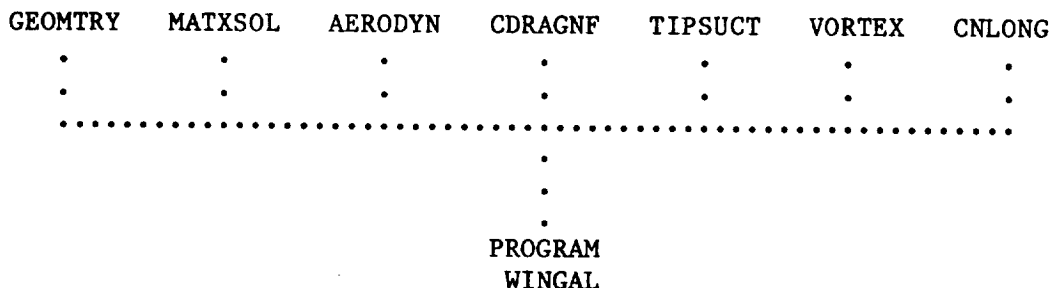
INPUT	TAPE6	TAPE20
OUTPUT	TAPE10	TAPE30
TAPE5	TAPE11	TAPE81

TAPE81 is equated to INPUT and TAPE6 to OUTPUT, and all formatted INPUT/OUTPUT is performed on these files. Routine READIN reads the input from TAPE81 and prints all of it except the title card over onto TAPE5. For the remainder of the program, TAPE5, serves as the primary input file. The remaining files are scratch files used by the program for auxillary storage. TAPE10 and TAPE11 are used in the matrix solver routines and are treated as random access backing store. The CDC routines READMS, WRITEMS, OPENMS, and CLOSEMS provide this capability.

TAPE10 is reused as a scratch file in the vortex analysis routine. There it is employed along with TAPE20 and TAPE30, and subsequently, TAPE30 is used to pass information from routine VORTEX back to routine TIPSUCT for the final coefficient reporting.

4 LOAD STRUCTURE

The tree diagram shown earlier can now be represented as follows:



where the nodes are main routines instead of functional descriptors. On CDC equipment, this method of loading is called Segmentation and is done by the CDC SEGRES Loader Program. The SEGRES program reads a directives file, shown in figure 1, and in effect structures the relocatable binary file into a set of movable pieces (here, each branch of the tree). When the main routine in these branches is called by WINGAL, the Loader brings into memory the entire branch, leaving the others out on mass storage. When a branch finishes processing and returns control back to the main program (WINGAL), that branch is moved back out to mass storage, and another one brought in and loaded into

the same space occupied by the previous one. Hence, the same space in memory is reused, which reduces the core requirements of the program. Complete details regarding the syntax of these directives and the actions of the SEGRES Loader Program are given in reference 8.

APPENDIX A

LRCGOSF ROUTINES USED BY VLM

PSEUDO
CALPLT
INFOPLT
NOTATE
PNTPLT

Documentation on these routines is given in reference 9, which is available from the NASA - LaRC ACD User Support Office.

APPENDIX B

LISTING OF VLM SOURCE CODE

```
C
C
PROGRAM WINGAL(INPUT,OUTPUT,TAPE81-INPUT,TAPE6=OUTPUT,  
$    TAPES,TAPE10,TAPE11,TAPE20,TAPE30)  
  
COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,  
$ STA(4), TBLSCW(100), YYCP(4),  
$ Q(400), PN(400), PV(400), ALPI(400), S(400), PSI(400),  
$ PHI(100), ZH(100), CP(400), STLOIND(4)  
  
COMMON /TOTTHREE/ CIR(400,2)  
  
COMMON /THREFOR/ CCAV(2,100), CLT, CLNT, NSSW, ALPO  
  
COMMON /ONETHRE/ TWIST(4), CREF, SREF, CAVE, CLDES, STRUE, AR,  
$ ARTTRUE, RTCDHT(4), CONFIG(2), NSSWSV(4),  
$ MSV(4), KBOT, PLAN, IPLAN, MACH,  
$ SSWWA(100), XL(4), XT(4), CLWB, CMCL, CLA(4), BLAIR(100),  
$ CLAMAR(4), CLWIN(4), CLWNG(4), XLOCIN,  
$ YINNER(4), YOUTER(4)  
  
INTEGER CONFIG  
  
COMMON /MAINONE/ ICODEOF, TOTAL, AANI(4), XS(4), YS(4), KFCTS(4),  
$ XREG(25,4), YREG(25,4), AREG(25,4), DIH(25,4), MCD(25,4),  
$ XX(25,4), YY(25,4), AS(25,4), ITWD(25,4), HMCD(25,4), AN(4),  
$ ZZ(25,4), IITPCOD, ICAMTST  
  
PRODUCTION CODE  
MARK IV VERSION  
SEPTEMBER, 1981  
  
PLEASE DIRECT ALL QUESTIONS, COMMENTS, ETC. TO:  
DR. JOHN E. LAMAR  
NASA - LANGLEY RESEARCH CENTER  
HAMPTON, VA. 23666      H.S. 287  
  
SOFTWARE SUPPORT ON THIS VERSION PROVIDED BY:  
HENRY E. HERBERT  
COMPUTER SCIENCES CORPORATION  
HAMPTON, VA. 23666  
  
--- NOTES TO THE USERS ---  
  
1. BOTH TOTAL RESULTS AND THOSE FROM THE LEADING
```

```

C      EDGE VORTEX SOLUTION WILL AGREE IF AND ONLY IF
C      ALL PANELS ARE OF UNIFORM WIDTH, AND CAN BE
C      CALCULATED FROM CLDES = 100. AND CLDES = 1.0
C
C      2. IF A WING HAS MORE THAN ONE STREAMWISE TIP, IT IS RECOM-
C      MENDED THAT THE WING BE INPUT AS TWO PLANFORMS TO PROVIDE
C      MORE MEANINGFUL SIDE EDGE RESULTS
C
C      3. STLOIND - "STREAMWISE LOAD INDICATOR" ARRAY; SET TO
C      0. IF THE LOADING ALONG THE ENTIRE OUTER STREAMWISE
C      EDGE OF THIS PLANFORM IS TO BE 0.0; OTHERWISE, SET TO
C      1.0 IF THIS LOADING IS TO BE NON-ZERO
C
C      COMMON/CCRRDD/ TSPAN(4), TSPAN, KBIT, CTILDA, XTILDA, DISTALE
C      DIMENSION INDEX(2)
C
C      VORTEX LATTICE AERODYNAMIC COMPUTATION
C      NASA-LRC PROGRAM NO. A2794
C
C      CALL OPENMS(11,INDEX,1,0)
C      CALL READIN
C
C      ICODEOF=TOTAL=0
C      CALL GEOMTRY
C      IF (M.EQ. -1) GO TO 70
C      IF (ICAMTST.EQ. 3) GO TO 70
C      IF (ICODEOF.GT.0) GO TO 70
C      IF (M.GT. 400) GO TO 40
C      NSW = 0
C      DO 15 IT = 1, IPLAN
C      NSW = NSW + NSSWSV(IT)
C      CONTINUE
C      IF (NSW.GT. 100) GO TO 30
C      ITSV=0
C      DO 20 IT=1, IPLAN
C      IF (AN(IT).LE.25.) GO TO 20
C      WRITE (6,100) IT,AN(IT)
C      ITSV=1
C      CONTINUE
C      IF (ITSV.GT.0) GO TO 60
C      GO TO 50

```

NOTES 7
NOTES 8
NOTES 9
NOTES 10
NOTES 11
NOTES 12
NOTES 13
NOTES 14
NOTES 15
NOTES 16
NOTES 17
NOTES 18
NOTES 19
NOTES 20
NOTES 21
NOTES 22
NOTES 23
WINGAL29
CCRRDD 2
CCRRDD 3
WINGAL31
WINGAL32
WINGAL33
WINGAL34
WINGAL35
WINGAL36
WINGAL37
WINGAL38
WINGAL39
WINGAL40
WINGAL41
WINGAL42
WINGAL43
WINGAL44
WINGAL45
WINGAL46
WINGAL47
WINGAL48
WINGAL49
WINGAL50
WINGAL51
WINGAL52
WINGAL53
WINGAL54
WINGAL55
WINGAL56
WINGAL57
WINGAL58
WINGAL59

```

30 WRITE (6,90) NSW
   GO TO 60
40 WRITE (6,80) M
   GO TO 60
50 CALL MATXSOL
   CALL AERODYN
   IF (ITPCOD.EQ.3) CALL CNLONG
   IF (CLOSE.EQ.100.) GO TO 55
   IF (IPTEST.EQ.1..OR.QTEST.EQ.1..OR.ITPCOD.EQ.2) GO TO 60
   CALL CDAGNF
   GO TO 57
55 CALL VORTEX
C
C TEST FOR ERROR CONDITIONS DETECTED IN VORTEX
C IF (M.EQ.-1) GO TO 70
C
57 CONTINUE
   IF (ITPCOD.EQ.1) CALL TIPSUCT
   TOTAL=TOTAL-1.
   GO TO 10
70 CONTINUE
C
C
C
80 FORMAT (1H1//10X,I6,93HHORSESHOE VORTICES LAIDOUT, THIS IS MORE THAN THE 400 MAXIMUM. THIS CONFIGURATION IS ABORTED.)
90 FORMAT (1H1//10X,I6,101H ROWS OF HORSESHOE VORTICES LAIDOUT. THIS IS MORE THAN THE 100 MAXIMUM. THIS CONFIGURATION IS ABORTED)
100 FORMAT (1H1//10X,8HPLANFORM,I6,4H HAS,I6,74H BREAKPOINTS. THE MAXIMUM DIMENSIONED IS 25. THE CONFIGURATION IS ABORTED.)
   END
   SUBROUTINE READIN
C
C THIS SUBROUTINE READS IN THE DATA OFF -TAPE81--, AND
C PRINTS IT TO -OUTPUT--,AND TO -TAPE5--
C
   DIMENSION ICARD(8)
   LINE = 0
   WRITE(6,100)
10 READ(81,200) ICARD
   IF (EOF(81)) 30, 20
20 LINE = LINE + 1
   WRITE(6,300) LINE, ICARD
   IF (LINE.EQ.1) GO TO 10
   WRITE(5,200) ICARD
   GO TO 10
C
30 ENDFILE 5
   REWIND 5
WINGAL60
WINGAL61
WINGAL62
WINGAL63
WINGAL64
WINGAL65
WINGAL66
WINGAL67
WINGAL68
WINGAL69
WINGAL70
WINGAL71
WINGAL72
WINGAL73
WINGAL74
WINGAL75
WINGAL76
WINGAL77
WINGAL78
WINGAL79
WINGAL80
WINGAL81
WINGAL82
WINGAL83
WINGAL84
WINGAL85
WINGAL86
WINGAL87
WINGAL88
WINGAL89
WINGAL90
READIN 2
READIN 3
READIN 4
READIN 5
READIN 6
READIN 7
READIN 8
READIN 9
READIN10
READIN11
READIN12
READIN13
READIN14
READIN15
READIN16
READIN17
READIN18
READIN19

```

```

C      100 FORMAT(1H1,30X,11HINPUT DATA ,/)
      200 FORMAT(8A10)
      300 FORMAT(110,1H,1X,8A10)
      RETURN
      END
      SUBROUTINE LOADING
C
C      THIS IS A DUMMY ROUTINE USED ONLY TO ENSURE THE
C      PROPER LOADING OF THE ITEMS NAMED BELOW.
C
      CALL STINDX(11,INDEX,1,0)
      CALL WRITMS(11,INDEX,1,0)
      RETURN
      END
      SUBROUTINE INFSUB (BOT,FVI,FWI,FUI)
      COMMON /INSUB23/ PSII,APHII,XXX,YYY,ZZZ,SNN,TOLRNC
      FC=COS(PSII)
      FS=SIN(PSII)
      FT=FS/FC
C
      FPC=COS(APHII)
      FPS=SIN(APHII)
      FPT=FPS/FPC
      F1=XXX+SNN*FT*FPC
      F2=YYY+SNN*FPC
      F3=ZZZ+SNN*FPS
      F4=XXX-SNN*FT*FPC
      F5=YYY-SNN*FPC
      F6=ZZZ-SNN*FPS
      FFA=(XXX**2+(YYY*FPS)**2+FPC**2*((YYY*FT)**2+(ZZZ/FC)**2-2.*XXX*YY
1Y*FT))-2.*ZZZ*FPC*(YYY*FPS+XXX*FT*FPS))
      FFB=(F1*F1+F2*F2+F3*F3)**.5
      FFC=(F4*F4+F5*F5+F6*F6)**.5
      FFE=F2*F2+F3*F3
      FFF=(F1*FPC*FT+F2*FPC+F3*FPS)/FFB-(F4*FPC*FT+F5*FPC+F6*FPS)/FFC
C
C      THE TOLERANCE SET AT THIS POINT IN THE PROGRAM MAY NEED TO BE
C      CHANGED FOR COMPUTERS OTHER THAN THE CDC 6000 SERIES
C
      IF (ABS(FFA).LT.(BOT*15.E-5)**2) GO TO 10
      FVONE=(XXX*FPS-ZZZ*FT*FPC)*FFF/FFA
      FVONE=(YYY*FT-XXX)*FFF/FFA*FPC
      FVONE=(ZZZ*FPC-YYY*FPS)*FFF/FFA
      GO TO 20

```

```

READIN20
READIN21
READIN22
READIN23
READIN24
READIN25
LOADING2
LOADING3
LOADING4
LOADING5
LOADING6
LOADING7
LOADING8
LOADING9
LOADIN10
INFSUB 2
INFSUB 3
INFSUB 4
INFSUB 5
INFSUB 6
INFSUB 7
INFSUB 8
INFSUB 9
INFSUB10
INFSUB11
INFSUB12
INFSUB13
INFSUB14
INFSUB15
INFSUB16
INFSUB17
INFSUB18
INFSUB19
INFSUB20
INFSUB21
INFSUB22
INFSUB23
INFSUB24
INFSUB25
INFSUB26
INFSUB27
INFSUB28
INFSUB29
INFSUB30
INFSUB31
INFSUB32
INFSUB33
INFSUB34
INFSUB35

```

[illegible]

```

C      EDGE VORTEX SOLUTION WILL AGREE IF AND ONLY IF
C      ALL PANELS ARE OF UNIFORM WIDTH, AND CAN BE
C      CALCULATED FROM CLDES = 100. AND CLDES = 1.0
C
C      2. IF A WING HAS MORE THAN ONE STREAMWISE TIP, IT IS RECOM-
C      MENDED THAT THE WING BE INPUT AS TWO PLANFORMS TO PROVIDE
C      MORE MEANINGFUL SIDE EDGE RESULTS
C
C      3. STLOIND - "STREAMWISE LOAD INDICATOR" ARRAY; SET TO
C      0. IF THE LOADING ALONG THE ENTIRE OUTER STREAMWISE
C      EDGE OF THIS PLANFORM IS TO BE 0.0; OTHERWISE, SET TO
C      1.0 IF THIS LOADING IS TO BE NON-ZERO
C
C      COMMON/CCRRDD/ TSPAN(4), TSPAN, KBIT, CTILDA, XTILDA, DISTALE
C
C      REAL MACH
C      INTEGER PRITCON
C      DIMENSION TEMP(8), INDEXES(4,3)
C      DATA INDEXES / 12*1 /
C      DATA NUMBER/5HFIRST,6HSECOND,5HTHIRD,6HFORTH/
C
C      PART ONE - GEOMETRY COMPUTATION
C
C      SECTION ONE - INPUT OF REFERENCE WING POSITION
C
C      IF(TOTAL.NE. 0.0)GO TO 5
C      DO 4 I=1,4
C      RTCDHT(I) = 0.0
C      XT(I) = 0.0
C      XL(I) = 0.0
C      4 CONTINUE
C      5 CONTINUE
C      YTOTL=1.E-10
C      AZY=1.E+13
C      PI = 4. * ATAN(1.)
C      PIT = PI / 2.
C      RAD = 180. / PI
C      IF (TOTAL.GT.0.) GO TO 70
C
C      SET PLAN EQUAL TO 1. FOR A WING ALONE COMPUTATION - EVEN FOR A
C      VARIABLE SWEEP WING
C
NOTES 7
NOTES 8
NOTES 9
NOTES 10
NOTES 11
NOTES 12
NOTES 13
NOTES 14
NOTES 15
NOTES 16
NOTES 17
NOTES 18
NOTES 19
NOTES 20
NOTES 21
NOTES 22
NOTES 23
GEOMTR12
CCRRDD 2
CCRRDD 3
GEOMTR14
GEOMTR15
GEOMTR16
GEOMTR17
GEOMTR18
GEOMTR19
GEOMTR20
GEOMTR21
GEOMTR22
GEOMTR23
GEOMTR24
GEOMTR25
GEOMTR26
GEOMTR27
GEOMTR28
GEOMTR29
GEOMTR30
GEOMTR31
GEOMTR32
GEOMTR33
GEOMTR34
GEOMTR35
GEOMTR36
GEOMTR37
GEOMTR38
GEOMTR39
GEOMTR40
GEOMTR41
GEOMTR42

```



```

C      SET PLAN EQUAL TO 2. FOR A WING - TAIL COMBINATION
C      SET PLAN = 3 OR 4 FOR CANARDS ETC.
C
C      SET TOTAL EQUAL TO THE NUMBER OF SETS
C      OF GROUP TWO DATA PROVIDED
C
C      READ(5,885)
C      $ PLAN, TOTAL, CREF, SREF, XLOCTN, CTILDA, XTILDA, DISTALE
C      IF(EOF(5)) 830,10
C      IPLAN=PLAN
C      XTILDA=XTILDA-XLOCTN
C
C      SET AAN(IT) EQUAL TO THE MAXIMUM NUMBER OF CURVES REQUIRED TO
C      DEFINE THE PLANFORM PERIMETER OF THE (IT) PLANFORM.
C
C      SET RTCDHT(IT) EQUAL TO THE ROOT CHORD HEIGHT OF THE LIFTING
C      SURFACE (IT), WHOSE PERIMETER POINTS ARE BEING READ IN, WITH
C      RESPECT TO THE WING ROOT CHORD HEIGHT
C
C      WRITE (6,860)
C      PRTCN = 10H
C      DO 60 IT=1,IPLAN
C      STLOIND(IT) = 0.0
C      READ(5,885)
C      $ AAN(IT), XS(IT), YS(IT), RTCDHT(IT), STLOIND(IT)
C      N=AAN(IT)
C      N1=N+1
C      MAK=0
C      IF(IPLAN.GT.1) PRTCN = NUMBER(IT)
C      WRITE (6,870) PRTCN,N,RTCDHT(IT),XS(IT),YS(IT)
C      WRITE (6,990)
C      DO 50 I=1,N1
C      READ(5,885)
C      $ XREG(I,IT), YREG(I,IT), DIH(I,IT), AMCD
C
C      IF (AMCD .NE. 2.) AMCD = 1.0
C      XREG(I,IT)=XREG(I,IT)-XLOCTN
C      MCD(I,IT)=AMCD
C      IF (I.EQ.1) GO TO 50
C      IF (MAK.NE.0.OR.MCD(I-1,IT).NE.2) GO TO 20
C      MAK=I-1
C      IF (ABS(YREG(I-1,IT))-YREG(I,IT)).LT.YTOL) GO TO 30
C      AREG(I-1,IT)=XREG(I-1,IT)-XREG(I,IT)/(YREG(I-1,IT)-YREG(I,IT))
C      ASWP=ATAN(AREG(I-1,IT))*RAD
C      GO TO 40
C      YREG(I,IT)=YREG(I-1,IT)
C      AREG(I-1,IT)=AZY
C      ASWP=90.

```

GEOMTR43

GEOMTR44

GEOMTR45

GEOMTR46

GEOMTR47

GEOMTR48

GEOMTR49

GEOMTR50

GEOMTR51

GEOMTR52

GEOMTR53

GEOMTR54

GEOMTR55

GEOMTR56

GEOMTR57

GEOMTR58

GEOMTR59

GEOMTR60

GEOMTR61

GEOMTR62

GEOMTR63

GEOMTR64

GEOMTR65

GEOMTR66

GEOMTR67

GEOMTR68

GEOMTR69

GEOMTR70

GEOMTR71

GEOMTR72

GEOMTR73

GEOMTR74

GEOMTR75

GEOMTR76

GEOMTR77

GEOMTR78

GEOMTR79

GEOMTR80

GEOMTR81

GEOMTR82

GEOMTR83

GEOMTR84

GEOMTR85

GEOMTR86

GEOMTR87

GEOMTR88

GEOMTR89

GEOMTR90

GEOMTR91

```

40      J=1-1
      C
      C      WRITE PLANFORM PERIMETER POINTS AND ANGLES
      C
      C      WRITE (6,960) J,XREG(J,IT),YREG(J,IT),ASWP,DIH(J,IT),MCD(J,IT)
      C      DIH(J,IT)=TAN(DIH(J,IT)/RAD)
      C      CONTINUE
      C      KFCIS(IT)=MAK
      C      WRITE (6,960) N1,XREG(N1,IT),YREG(N1,IT)
      C      CONTINUE
      C
      C      PART 1 - SECTION 2
      C      READ GROUP 2 DATA AND COMPUTE DESIRED WING POSITION
      C
      C      SET SAIL-IPLAN) EQUAL TO SWEEP ANGLE IN DEGREES
      C      FOR THE FIRST
      C      CURVE(S) THAT CAN CHANGE SWEEP FOR EACH PLANFORM
      C
      C      IF A PARTICULAR VALUE OF CL IS DESIRED AT WHICH THE LOADINGS ARE
      C      TO BE COMPUTED, SET CLDES EQUAL TO THIS VALUE
      C      SET CLDES EQUAL TO 11. FOR A DRAG POLAR AT CL VALUES OF-.1 TO 1.0
      C      GEOMT111
      C      GEOMT112
      C      GEOMT113
      C      GEOMT114
      C      GEOMT115
      C      IF PTEST IS SET EQUAL TO ONE THE PROGRAM WILL COMPUTE CLP
      C      IF QTEST IS SET EQUAL TO ONE THE PROGRAM WILL COMPUTE CMQ AND CLOGEOMT116
      C      DO NOT SET BOTH PTEST AND QTEST TO ONE FOR A SINGLE CONFIGURATION
      C      GEOMT117
      C      GEOMT118
      C      SET TWIST(1-IPLAN) EQUAL TO 0. FOR A FLAT PLANFORM
      C      AND TO 1.
      C      FOR A PLANFORM THAT HAS TWIST AND/OR CAMBER
      C
      C      SET ATPCOD TO ONE IF THE CONTRIBUTIONS TO LIFT, DRAG AND MOMENT
      C      FROM SEPERATED FLOW AROUND THE LEADING AND/OR SIDE EDGES IS
      C      DESIRED. OTHERWISE SET ATPCOD TO ZERO.
      C
      C      70 READ(5,950) CONFIG(1), CONFIG(2), SCW, VIC, MACH, CLDES,
      C      $ (SA(1),I=1,4),(TWIST(J),J=1,4),PTEST,
      C      $ QTEST,ATPCOD
      C
      C      IF (EOF(5)) 830, 71
      C      71 IF (VIC .GE. 10.) GO TO 72
      C      WRITE(6,1070)
      C      M=-1
      C      GO TO 2000
      C
      C      72 IF( CLDES .NE. 100. .OR. ATPCOD .NE. 1.) GO TO 73
      C      IF (CLDES .EQ. 100. .AND. ATPCOD .EQ. 1. .AND.
      C      $ SCW .GE. 2.) GO TO 73
      C      WRITE(6,1080)
      C
      C      GEOMTR92
      C      GEOMTR93
      C      GEOMTR94
      C      GEOMTR95
      C      GEOMTR96
      C      GEOMTR97
      C      GEOMTR98
      C      GEOMTR99
      C      GEOMT100
      C      GEOMT101
      C      GEOMT102
      C      GEOMT103
      C      GEOMT104
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      C      GEOMT106
      C      GEOMT107
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      C      GEOMT131
      C      GEOMT132
      C      GEOMT133
      C      GEOMT134
      C      GEOMT135
      C      GEOMT136
      C      GEOMT137
      C      GEOMT138
      C      GEOMT139
      C      GEOMT140

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M = -1
GO TO 2000
73 ICAMTST = 0
DO 74 IT = 1,4
  IF(TWIST(IT) .NE. 0.) ICAMTST = 1
74 CONTINUE
  IF(ATPCOD .EQ. 1.) ICAMTST = ICAMTST + 1
  IF(CLDES .NE. 100.) ICAMTST = ICAMTST + 1
  IF(ICAMTST .NE. 3) GO TO 75
  WRITE(6,1060)
  GO TO 2000
75 ITPCOD = ATPCOD
  IF (ATPCOD .NE. 2.) GO TO 90
  IF (CLDES .NE. 11. .OR. CLDES .NE. 100.) GO TO 90
  M = -1
  WRITE(6,1100) CLDES
  GO TO 2000
C
90 IF (ITPCOD .NE. 1) GO TO 110
C
C      READ IN THE LIMITS OF INTEGRATION ON
C      THE SIDE EDGES, AND ON THE LEADING/TRAILING
C      EDGES
C      (IF ATPCOD = 1)
C
  READ(5,885) (YINNER(IT),YOUTER(IT),IT=1,IPLAN)
  READ(5,885) (XL(IT),XT(IT),IT=1,IPLAN)
  DO 100 IT= 1,IPLAN
    XL(IT) = XL(IT) - XLOCTN
    XT(IT) = XT(IT) - XLOCTN
100 CONTINUE
  IF (SCW .NE. 0.) GO TO 110
  I = 1
  DO 107 ITT = 1,IPLAN
    IF (XL(ITT) .NE. XT(ITT)) I = I + 1
107 CONTINUE
  IF (I .EQ. 0) GO TO 110
  M = -1
  WRITE(6,1110)
  GO TO 2000
C
110 CONTINUE
  WRITE (6,890) CONFIG
  IF (PTST .EQ. 1. .AND. QTEST .EQ. 1.) GO TO 850
  IF (SCW.EQ.0.) GO TO 140
  DO 125 I = 1,100
    TALSCW(I) = SCW
125 CONTINUE
  GO TO 150

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GEOMT141
 GEOMT142
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 GEOMT189

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C
C   READ IN THE -STA- AND -TBLSCN- VALUES
140 ISTART = 0
IEND = 0
NSTA = 0
DO 145 ITT = 1, IPLAN
  READ(5,880) STA(ITT)
  NSTA = NSTA + STA(ITT)
  ISTART = IEND + 1
  IEND = NSTA
  READ(5,890) (TBLSCN(IKK), IKK=ISTART, IEND)
145 CONTINUE
C
C
150 DO 410 IT=1, IPLAN
  N=AN(IT)
  N1=N+1
  DO 160 I=1, N
    XREF(I)=XREG(I, IT)
    YREF(I)=YREG(I, IT)
    A(I)=AREG(I, IT)
    RSAR(I)=ATAN(A(I))
    IF (A(I).EQ.AZY) RSAR(I)=PIT
  CONTINUE
  XREF(N1)=XREG(N1, IT)
  YREF(N1)=YREG(N1, IT)
  IF (KFACTS(IT).GT.0) GO TO 170
  K=1
  SA(IT)=RSAR(I)*RAD
  GO TO 180
  K=KFACTS(IT)
  WRITE (6,920) K, SA(IT), IT
  SB=SA(IT)/RAD
  IF (ARS(SB-RSAR(K)).GT.(.1/RAD)) GO TO 210
  REFERENCE PLANKFORM COORDINATES ARE STORED UNCHANGED FOR WINGS
  WITHOUT CHANGE IN SWEEP
  DO 200 I=1, N
    X(I)=XREF(I)
    Y(I)=YREF(I)
    IF (RSAR(I).EQ.PIT) GO TO 190
    A(I)=TAN(RSAR(I))
    GO TO 200
  A(I)=AZY
190 SAR(I)=RSAR(I)
200 X(N1)=XREF(N1)
  Y(N1)=YREF(N1)
  GO TO 390
C
C   CHANGES IN WING SWEEP ARE MADE HERE
GEOMT190
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GEOMT238

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C
210 IF (MCD(K,IT).NE.2) GO TO 840
    KA=K-1
    DO 220 I=1,KA
        X(I)=XREF(I)
        Y(I)=YREF(I)
    220 SAR(I)=RSAR(I)
    C DETERMINE LEADING EDGE INTERSECTION BETWEEN FIXED AND VARIABLE
    C SWEEP WING SECTIONS
        SAR(K)=SB
        A(K)=TAN(SB)
        SAI=SB-RSAR(K)
        X(K+1)=XS(IT)+(XREF(K+1)-XS(IT))*COS(SAI)+(YREF(K+1)-YS(IT))*SIN(SGEOMT251
        Y(K+1)=YS(IT)+(YREF(K+1)-YS(IT))*COS(SAI)-(XREF(K+1)-XS(IT))*SIN(SGEOMT252
        1AI)
        1AI)
        IF (ABS(SB-SAR(K-1)).LT.(.1/RAD)) GO TO 230
        Y(K)=X(K+1)-X(K-1)-A(K)*Y(K+1)+A(K-1)*Y(K-1)
        Y(K)=Y(K)/(A(K-1)-A(K))
        X(K)=A(K)*X(K-1)-A(K-1)*X(K+1)+A(K-1)*A(K)*(Y(K+1)-Y(K-1))
        X(K)=X(K)/(A(K)-A(K-1))
        GO TO 240
    C ELIMINATE EXTRANEOUS BREAKPOINTS
    230 X(K)=XREF(K-1)
        Y(K)=YREF(K-1)
        SAR(K)=SAR(K-1)
        K=K+1
    240 K=K+1
    C SWEEP THE BREAKPOINTS ON THE VARIABLE SWEEP PANEL
    C (IT ALSO KEEPS SWEEP ANGLES IN FIRST OR FOURTH QUADRANTS)
    250 K=K+1
        SAR(K-1)=SAI+RSAR(K-1)
        IF (SAR(K-1).LE.PIT) GO TO 270
        SAR(K-1)=SAR(K-1)-3.1415927
        GO TO 260
    270 IF (SAR(K-1).GE.(-PIT)) GO TO 280
        SAR(K-1)=SAR(K-1)+3.1415927
        GO TO 270
    280 IF ((SAR(K-1)).LT..0) GO TO 290
        IF (SAR(K-1)-PIT) 320,300,300
    290 IF (SAR(K-1)+PIT) 310,310,320
    300 A(K-1)=AZY
        GO TO 330
    310 A(K-1)=-AZY
        GO TO 330
    320 A(K-1)=TAN(SAR(K-1))
    330 KK=MCD(K,IT)
        IF (KK.EQ. 1) GO TO 350
        Y(K)=YS(IT)+(YREF(K)-YS(IT))*COS(SAI)-(XREF(K)-XS(IT))*SIN(SAI)
        X(K)=XS(IT)+(XREF(K)-XS(IT))*COS(SAI)+(YREF(K)-YS(IT))*SIN(SAI)
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        GEOMT286
        GEOMT267

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440      GO TO 460
      IF(JP .EQ. 1) GO TO 460
      JPSV = JP
      IND=NIT-(JPSV-1)
      DO 450 JP=1,IND
      K2=NIT-JP+2
      K1=NIT-JP+1
      XX(K2,IT)=XX(K1,IT)
      YY(K2,IT)=YY(K1,IT)
      MMCD(K2,IT)=MMCD(K1,IT)
      AS(K2,IT)=AS(K1,IT)
      TTWD(K2,IT)=TTWD(K1,IT)
      YY(JPSV,IT)=YY(I,ITT)
      AS(JPSV,IT)=AS(JPSV-1,IT)
      TTWD(JPSV,IT)=TTWD(JPSV-1,IT)
      XX(JPSV,IT)=(YY(JPSV,IT)-YY(JPSV-1,IT))*AS(JPSV-1,IT)+XX(JPSV-1,IT)
1)
      MMCD(JPSV,IT)=MMCD(JPSV-1,IT)
      AN(IT)=AN(IT)+1.
      NIT=NIT+1
      CONTINUE
460      CONTINUE
470      C
      C
      C
      SEQUENCE WING COORDINATES FROM TIP TO ROOT

      N1=AN(IT)+1.
      DO 480 I=1,N1
      Q(I)=YY(I,IT)
      DO 520 J=1,N1
      HIGH=1.
      DO 490 I=1,N1
      IF ((Q(I)-HIGH).GE.0.) GO TO 490
      HIGH=Q(I)
      IH=I
      CONTINUE
490      IF (J.NE.1) GO TO 500
      ROTSV(IT)=HIGH
      KFX(IT)=IH
      Q(IH)=1.
      SPY(J,IT)=HIGH
      IF (IH.GT.KFX(IT)) GO TO 510
      IYL(J,IT)=1
      IYT(J,IT)=0
      GO TO 520
      IYL(J,IT)=0
      IYT(J,IT)=1
      CONTINUE
520      C
530      CONTINUE
      C

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C      SELECT MAXIMUM B/2 AS THE WING SPAN.      IF BOTH FIRST AND
C      SECOND PLANFORMS HAVE SAME SEMISPAN THEN THE SECOND PLANFORM IS
C      TAKEN TO BE THE WING.
C
      KBOT=1
      DO 535 IT = 1, IPLAN
      IF (ABS(BOTSV(IT)).GE.ABS(BOTSV(KBOT))) KBOT = IT
535    CONTINUE
      BOT=BOTSV(KBOT)
C
C      COMPUTE NOMINAL HORSESHOE VORTEX WIDTH ALONG WING SURFACE
C
      DO 570 IT = 1, IPLAN
      TSPAN(IT) = 0.0
      ISAVE = KFX(IT) - 1
      I = KFX(IT) - 2
540    IF (I.EQ.0) GOTO 550
      IF (TTWD(I,IT).EQ.TTWD(ISAVE,IT)) GOTO 560
550    CTWD = COS(ATAN(TTWD(ISAVE,IT)))
      TLGTH = (YY(ISAVE+1,IT)-YY(I+1,IT))/CTWD
      TSPAN(IT) = TSPAN(IT) + TLGTH
      IF (I.EQ.0) GO TO 570
      ISAVE=I
560    I=I-1
      GO TO 540
570    CONTINUE
      VI = TSPAN(KBOT) / VIC
      VSTOL=VI/2
C
C      ELIMINATE PLANFORM BREAKPOINTS WHICH ARE WITHIN (8/2)/2000 UNITS
C      Laterally
C
      DO 630 IT=1, IPLAN
      N=AN(IT)
      N1=N+1
      DO 630 J=1, N
      AA=ABS(SPY(J,IT)-SPY(J+1,IT))
      IF (AA.EQ.0..OR.AA.GT.ABS(TSPAN(KBOT)/2000.)) GOTO 630
      IF (AA.GT.YTOL) WRITE (6,1010) SPY(J+1,IT),SPY(J,IT)
      DO 620 I=1, N1
      IF (YY(I,IT).NE.SPY(J+1,IT)) GO TO 620
      YY(I,IT)=SPY(J,IT)
      CONTINUE
620    SPY(J+1,IT)=SPY(J,IT)
      CONTINUE
630
C      COMPUTE 2 COORDINATES
C
      DO 670 IT=1, IPLAN

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 GEOMT434


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640      JM=N1+AN(IT)+1.
        DO 640 JZ=1,N1
          ZZ(JZ,IT)=RTCDHT(IT)
          JZ=1
650      IF (JZ.GT.KFX(IT)) GO TO 660
          ZZ(JZ,IT)=ZZ(JZ-1,IT)+(YY(JZ,IT)-YY(JZ-1,IT))*TTWD(JZ-1,IT)
          GO TO 650
660      JM=JM-1
          IF (JM.EQ.KFX(IT)) GO TO 670
          ZZ(JM,IT)=ZZ(JM+1,IT)+(YY(JM,IT)-YY(JM+1,IT))*TTWD(JM,IT)
          GO TO 660
        CONTINUE
670      WRITE PLANFORM PERIMETER POINTS ACTUALLY USED IN THE COMPUTATIONS
        C
        C
        C
        WRITE (6,900)
        DO 690 IT=1,IPLAN
          N=AN(IT)
          N1=N+1
          IF (IPLAN.GT.1) WRITE(6,1000) NUMBER(IT)
          DO 680 KK=1,N
            TOUT=ATAN(TTWD(KK,IT))*RAD
            ADUT=ATAN(AS(KK,IT))*RAD
            IF (AS(KK,IT).EQ.AZY) ADUT=90.
            WRITE (6,910) KK,XX(KK,IT),YY(KK,IT),ZZ(KK,IT),ADUT,TOUT,NMCD(KK,IGEOMT460
            IT)
680      CONTINUE
        WRITE (6,910) N1,XX(N1,IT),YY(N1,IT),ZZ(N1,IT)
690      CONTINUE
        C
        C
        C
        PART ONE - SECTION THREE - LAY OUT YAWED HORSESHOE VORTICES

        STRUE=0.
        DO 695 IT = 1,IPLAN
          NSSWSV(IT) = 0
          MSV(IT) = 0
695      CONTINUE
          MSTOT = 0
          NSTOT = 0
          DO 780 IT=1,IPLAN
            N1=AN(IT)+1.
            I=0
            J=1
            YIN=BOTSV(IT)
            ILE=ITE-KFX(IT)
            DETERMINE SPANWISE BORDERS OF HORSESHOE VORTICES
            IXL=IXT=0
            I=I+1
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C      CPHI = COS(ATAN(TTWD(ILE,IT)))
C      IF (YIN-GE.(SPY(J,IT)+VSTOL*CPHI)) GO TO 710
C      BORDER IS WITHIN VORTEX SPACING TOLERANCE (VSTOL) OF BREAKPOINT
C      THEREFORE USE THE NEXT BREAKPOINT INBOARD FOR THE BORDER
      VBORD(I)=YIN
      GO TO 740
C      USE NOMINAL VORTEX SPACING TO DETERMINE THE BORDER
      VBORD(I)=SPY(J,IT)
C      COMPUTE SUBSCRIPTS ILE AND ITE TO INDICATE WHICH
C      BREAKPOINTS ARE ADJACENT AND WHETHER THEY ARE ON THE WING LEADING
C      EDGE OR THE TRAILING EDGE
      IF (J-GE.N1) GO TO 730
      IF (SPY(J,IT).NE.SPY(J+1,IT)) GO TO 730
      IXL=IXL+IYL(J,IT)
      IXT=IXT+IYT(J,IT)
      J=J+1
      GO TO 720
C      YIN=SPY(J,IT)
      IXL=IXL+IYL(J,IT)
      IXT=IXT+IYT(J,IT)
      J=J+1
C      CONTINUE
      IPHI=ILE-IXL
      IF (J-GE.N1) IPHI=1
      YIN=YIN-VI*COS(ATAN(TTWD(IPHI,IT)))
      IF (I-NE.1) GO TO 760
      ILE=ILE-IXL
      ITE=ITE+IXT
      GO TO 700
C      COMPUTE COORDINATES FOR CHORDWISE ROW OF HORSESHOE VORTICES
      YQ=(VBORD(I-1)+VBORD(I))/2.
      HW=(VBORD(I)-VBORD(I-1))/2.
      IM1 = I - 1 + NSWTOT
      ZH(IM1)=ZZ(ILE,IT)+(YQ-YY(ILE,IT))*TTWD(ILE,IT)
      PHI(IM1)=TTWD(ILE,IT)
      SS/WA(IM1)=AS(ILE,IT)
      XLE=XX(ILE,IT)+AS(ILE,IT)*(YQ-YY(ILE,IT))
      XTE=XX(ITE,IT)+AS(ITE,IT)*(YQ-YY(ITE,IT))
      XLOCAL=(XLE-XTE)/TBLSCW(IM1)
C      COMPUTE WING AREA PROJECTED TO THE X - Y PLANE
      STPUE=STRUE*XLOCAL*TBLSCW(IM1)*(HW*2.)*2.
      NSCW=TBLSCW(IM1)
      DD 770 JCW=1,NSCW
      AJCW=JCW-1
      XLEL=XLE-AJCW*XLOCAL
      NTS = JCW + MSTOT

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PN(NTS)=XLEL-.25*XLOCAL
P(NTS)=XLEL-.75*XLOCAL
PS(NTS)=-((XLE-PN(NTS))*AS(ITE,IT)+(PN(NTS)-XTE)*AS(ILE,IT))/(XLE-
1XTE)
S(NTS)=HW/CPHI
Q(NTS)=YQ
CONTINUE
MSV(IT)=MSV(IT)+NSCW
%STOT = %STOT + NSCW
770
C
C
C
TEST TO DETERMINE WHEN WING ROOT IS REACHED
IF (VBORD(I).LT.YREG(1,IT)) GO TO 750
780
C
C
C
NSSWSV(IT)=I-1
NSWTOT = NSWTOT + NSSWSV(IT)
CONTINUE
M = 0
DO 781 IT = 1,IPLAN
M = M + MSV(IT)
781 CONTINUE
C
C
C
COMPUTE ASPECT RATIO AND AVERAGE CHORD
BOT=-BOT
AR=4.*BOT*BOT/SPEF
ARTRUE=4.*BOT*BOT/STRUE
CAVE=STRUE/(2.*BOT)
BETA=(1.-MACH*MACH)**.5
C
C
C
C
READ IN THE -ALP- VALUES. IF TWIST(ITT) = 0.,
THEN BYPASS THE READ AND SIMPLY SET THEM TO 0.
DEGRAD = 1. / RAD
ISTART = 0
IEND = 0
INDXTBL = 0
DO 810 ITT = 1,IPLAN
JPANGE = NSSWSV(ITT)
DO 805 J = 1,JRANGE
INDXTBL = INDXTBL + 1
NUMGALP = TBLSCW(INDXTBL)
ISTART = IEND + 1
IEND = IEND + NUMGALP
IF (TWIST(ITT) .EQ. 0.) GO TO 795
READ(5,985) (ALP(ITOE),ITOE=ISTART,IEND)
IF (TWIST(ITT) .EQ. 1.) GO TO 805
DO 790 ITOE = ISTART,IEND
ALP(ITOE) = ALP(ITOE) + DEGRAD
CONTINUE
790
C
C
C
C
C
C

```



```

850      GO TO 2000
      ICODEOF=3
      WRITE (6,970) PTEST,QTEST
      GO TO 2000
C
C
C
860      FORMAT (1H1//63X,13HGEOMETRY DATA)
870      FORMAT (//45X,A10,22HREFERENCE PLANFORM HAS,13,7H CURVES//12X,19HGEOMT639
      1ROOT CHORD HEIGHT =,F12.5,4X,29HVARIBLE SWEEP PIVOT POSITION,4X,6GEOMT640
      2HX(S) =,F12.5,5X,6HY(S) =,F12.5//46X,40HBREAK POINTS FOR THE REFERGEOMT641
      3ENCE PLANFORM //
      880 FORMAT(16F5.1)
      885 FORMAT(8F10.6)
      890 FORMAT(1H1//,47X,16HCONFIGURATION : ,2A10)
900      FORMAT (22X,5HPOINT,6X,1HX,11X,1HY,11X,1HZ,10X,5HSWEEP,7X,8HDIHEDRGEOMT646
      1AL,4X,4HMOVE/68X,5SHANGLE,8X,5SHANGLE,6X,4HCODE/)
910      FORMAT (20X,15,3F12.5,2F14.5,16)
920      FORMAT (/40X,5HCURVE,13,9H IS SWEPT,F12.5,20H DEGREES ON PLANFORM,GEOMT649
      113)
930      FORMAT(1H1//,41X,
      $ 43HEND OF FILE ENCOUNTERED AFTER CONFIGURATION ,1X,2A10)
940      FORMAT (1H1//11X,45HTHE FIRST VARIABLE SWEEP CURVE SPECIFIED (K =GEOMT652
      1,13,44H ) DOES NOT HAVE AN M CODE OF 2 FOR PLANFORM,I4)
      950 FORMAT(2A10,8F5.2,7F2.0)
960      FORMAT (26X,15,2F12.5,2F16.5,4X,I4)
970      FORMAT (1H1//30X,38HERROR - PROGRAM CANNOT PROCESS PTEST =,F5.1,12GEOMT657
      1H AND QTEST =,F5.1)
980      FORMAT (//48X,35HBREAK POINTS FOR THIS CONFIGURATION//)
990      FORMAT (28X,5HPOINT,6X,1HX,11X,1HY,11X,5HSWEEP,10X,8HDIHEDRAL,7X,4GEOMT659
      1HMOVE/38X,3HREF,9X,3HREF,10X,5SHANGLE,11X,5SHANGLE,9X,4HCODE/)
1000     FORMAT(//52X,A10,22H PLANFORM BREAK POINTS/)
1010     FORMAT (//125X,34HTHE BREAKPOINT LOCATED SPANWISE AT,F11.5,3X,20HGEOMT662
      1HAS BEEN ADJUSTED TO,F9.5//)
1020     FORMAT(1H0,43X,I5,
      $ 41H HOPSESHOE VORTICES IN EACH CHORDWISE ROW )
1030     FORMAT (/23X,94HTABLE OF HORSESHOE VORTICES IN EACH CHORDWISE ROW
      1(FROM TIP TO ROOT BEGINNING WITH FIRST PLANFORM)//25F5.0/25F5.0)
C
1040     FORMAT(//,53X,
      $ 30HHOPSESHOE VORTEX SUMMARY TABLE ,/,
      $ 33X, 15, 37H HORSESHOE VORTICIES USED ON THE LEFT,
      $ 27H HALF OF THIS CONFIGURATION ,//,
      $ 50X,8HPLANFORM,7X,5HTOTAL,8X,8HSPANWISE,/)
C
1050     FORMAT (52X,I4,10X,I3,11X,I4)
1060     FORMAT(1H1, 9X, 22HF A T A L      E R R O R ,//,
      $ 10X, 46HYOU ARE REQUESTING A VORTEX LIFT SOLUTION FOR ,/,
      $ 10X, 46HA CAMBERED WING WITH -CLODES- NOT EQUAL TO 100. ,//,

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      $ 10X, 46MTHIS WILL NOT YIELD ACCURATE RESULTS. PLEASE  ,/,
      $ 10X, 46HRESUBMIT WITH -CLDES- = 100.
1070 FORMAT(1H1,9X,22HF A T A L E R R O R ,/,
      $ 10X,39HTHE VALUE OF -VIC- IS TOO SCALL. PLEASE ,/,
      $ 10X,30HMAKE CORRECTIONS AND RESUBMIT. )
C
1080 FORMAT(1H1,9X,22HF A T A L E R R O R ,/,
      $ 10X,34HTHE VALUE OF -SCW- MUST BE GREATER ,/,
      $ 10X,28HTHAN 2 FOR THIS COMBINATION. )
C
1090 FORMAT(1H1,9X,22HF A T A L E R R O R ,/,
      $ 10X,*YOUR INPUT VALUES FOR -STA- DID NOT *,/,
      $ 10X,*MATCH UP WITH THE NUMBER OF SPANWISE*,/,
      $ 10X,*STATIONS COMPUTED. REFER TO THE HORSESHOE *,/,
      $ 10X,*VORTEX SUMMARY TABLE ABOVE TO OBTAIN THE *,/,
      $ 10X,*PROPER VALUES FOR -STA- PER PLANFORM*,/)
C
1100 FORMAT(1H1,9X,22HF A T A L E R R O R ,/,
      $ 10X,*YOU HAVE REQUESTED THE FLOWFIELD OPTION, BUT*,/,
      $ 10X,*HAVE SPECIFIED -CLDES- INCORRECTLY (*F5.1,*) *,/,
      $ 10X, *PLEASE MAKE CORRECTIONS AND RESUBMIT. *,/)
C
1110 FORMAT(1H1,9X,22HF A T A L E R R O R ,/,
      $ 10X,*THE SIDE-EDGE FORCE IS NOT PROPERLY COMPUTED *)
2000 CONTINUE
      ENO
      SUBROUTINE PLANPLT(IPLAN,XX,YY,AN,ICON,ITHETA)
C
C      THIS ROUTINE PREPARES A PLOT DIAGRAM OF THE
C      INPUT PLANFORM CONFIGURATION(S) FOR THE LINE
C      PRINTER
C
C      IPLAN--NUMBER OF PLANFORMS TO BE PLOTTED
C      XX --X COORDINATE ARRAY DIMENSIONED 25 X 4
C      YY --Y COORDINATE ARRAY DIMENSIONED 25 X 4
C      AN --NUMBER OF POINTS IN PLANFORM LESS ONE
C      ICON --O REQUESTS CONTOUR DRAWINGS.
C      ITHETA--ANGLE OF ROTATION DESIRED IN DEGREES,
C              O FOR NO ROTATION
C
C      ROBERT GRAY COMPUTER SCIENCES CORP 1980
C
      DIMENSION XX(25,4), YY(25,4), AN(4)
      DIMENSION NARRAY (100)
      DIMENSION XXX(25,4), YYY(25,4)
      DIMENSION NXX(26,4), NYY(26,4), NCHAR(4)
      DIMENSION IXPTS(50)

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PLANPL71
PLANPL72

DATA NARRAY /100*1H /
DATA IPLUS /1H+/
DATA IBLANK /10H /
DATA NCHAR/1H0,1H#,1H$,1H%,
DATA YMIN /99999.0/, YMAX /-99999.0/
DATA XMIN /99999.0/, XMAX /-99999.0/

      ROTATE PLANFORM(S)

      THETA = ITHETA * (3.14159/180)
      CANS = COS (THETA)
      SANS = SIN (THETA)
      DO 5 I = 1, IPLAN
        M = AN(I) + 1
        DO 5 N = 1,M
          XXX(N,I) = XX(N,I) * CANS - YY(N,I) * SANS
          YYY(N,I) = XX(N,I) * SANS + YY(N,I) * CANS
        CONTINUE
      CONTINUE

      RESCALE ALL X AND Y COORDINATES
      Y IS 6 PER INCH ON THE LINE PRINTER, MAX 7 INCHES
      X IS 10 PER INCH, MAX 10 INCHES

      FIND MINIMUM AND MAXIMUM COORDINATES
      DO 10 I = 1,IPLAN
        M = AN(I) + 1
        DO 20 N = 1,M
          IF (YYY(N,I) .LT. YMIN) YMIN = YYY(N,I)
          IF (YYY(N,I) .GT. YMAX) YMAX = YYY(N,I)
          IF (XXX(N,I) .LT. XMIN) XMIN = XXX(N,I)
          IF (XXX(N,I) .GT. XMAX) XMAX = XXX(N,I)
        CONTINUE
      CONTINUE

      VALUES USED AS SUBSCRIPTS SHOULD BE GREATER
      THAN OR EQUAL TO 1

      IF (YMIN .GT. 0.0) YUP = 0.0 - YMIN
      IF (YMIN .LE. 0.0) YUP = ABS(YMIN)
      DO 70 I = 1,IPLAN
        M = AN(I) + 1
        DO 60 N = 1,M
          YYY(N,I) = YYY(N,I) + YUP + 1.0
        CONTINUE
      CONTINUE
      YMAX = YMAX + YUP + 1
      YMIN = YMIN + YUP + 1.0

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PLANP119
PLANP120
PLANP121

IF (XMIN .GT. 0.0) XUP = 0.0 - XMIN
IF (XMIN .LE. 0.0) XUP = ABS(XMIN)
DO 85 I = 1,IPLAN
M = AN(I) + 1
DO 90 N = 1,M
XXX(N,I) = XXX(N,I) + XUP + 1.0
CONTINUE
CONTINUE
XMAX = XMAX + XUP + 1.0
XMIN = XMIN + XUP + 1.0

C
C
      FIND Y SCALING FACTOR--MAXIMUM 42 LINES
YDIM = YMAX
YSCALE = 42.0/YDIM
      FIND X SCALING FACTOR--10 CHARACTERS PER INCH
      AND IN PROPORTION TO Y INCHES
XSCALE = YSCALE * 1.666
XDIM = XMAX * XSCALE
IF (XDIM .LE. 100.49) GO TO 30
      SCALED X DIMENSION IS GREATER THAN 100, RESCALE
RESCALE = 100.0/XDIM
YSCALE = YSCALE * RESCALE
XSCALE = XSCALE * RESCALE

C
C
      COMPUTE ALL RESEALED X AND Y COORDINATES,
      ROUNDING EACH
DO 40 I = 1,IPLAN
M = AN(I) + 1
DO 50 N = 1,M
RXX = XXX(N,I) * XSCALE * 10.0 + 5.0
NXX(N,I) = PXX/10.0
RYY = YYY(N,I) * YSCALE * 10.0 + 5.0
NYY(N,I) = RYY/10.0
CONTINUE
CONTINUE

C
C
      FIND INTEGER MAXIMA AND MINIMA
YMAX = YMAX * YSCALE * 10.0 + 5.0
MAXYY = YMAX/10.0
YMIN = YMIN * YSCALE * 10.0 + 5.0
MINYY = YMIN/10
XMAX = XMAX * XSCALE * 10.0 + 5.0
MAXXX = XMAX/10.0
XMIN = XMIN * XSCALE * 10.0 + 5.0
MINXX = XMIN/10.0

C
C
C
C
30
50
40
C
C
C

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C
C      CENTER THE DATA
      ICNTER = ((100 - (MAXXX - MINXX)) / 2) - MINXX
      DO 65 I = 1, IPLAN
        M = AN(I) + 1
        DO 65 N = 1, M
          NXX(N, I) = NXX(N, I) + ICNTER
        CONTINUE
      65
C
C      WRITE HEADING
      60 WRITE(6, 802) (I, NCHAR(I), I=1, IPLAN)
      802 FORMAT(1H1, 51X, 34HAPPROXIMATE PLANFORM CONFIGURATION, /
        *(4X, 9HPLANFORM, I2, 6H IS, A1))
C
C      DRAW THE INPUT PLANFORM CONFIGURATIONS
      FOR EACH SCAN LINE
      FOR EACH SCAN LINE
      DO 500 ISCAN = MINYY, MAXYY
        CARRIAGE CONTROL FOR THE FIRST PLANFORM ON A LINE
        IS BLANK, AFTERWARDS IT IS A PLUS
        ICC = IBLANK
      FOR EACH PLANFORM
        FIND POINTS ON THIS SCAN LINE
      DO 400 I = 1, IPLAN
        M = AN(I) + 1
        FIND BREAKPOINTS ON THIS LINE
      DO 110 N = 1, M
        IF (NYY(N, I) .EQ. ISCAN) NARRAY (NXX(N, I)) = NCHAR(I)
      CONTINUE
      110
C
C      INSURE A CLOSED PLANFORM
      NXX(M+1, I) = NXX(1, I)
      NYY(M+1, I) = NYY(1, I)
C
C      OFFSET SCAN LINE TO INSURE VERTICES ARE NOT INTERSECTED
      SCAN = ISCAN + 0.1
C
C      FIND POINTS AT WHICH SCAN LINE INTERSECTS LINE
      SEGMENTS
      ICOUNT = 0
      FOR EACH LINE SEGMENT...
      DO 300 N = 1, M
        IF THIS SEGMENT IS HORIZONTAL, IT MUST BE

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PLANP122
 PLANP123
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 PLANP170

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C      C      FILLED SEPARATELY.
C
      IF (ISCAN.NE. NYY(N,I)) GO TO 120
      IF (NYY(N+1,I).NE. NYY(N,I)) GO TO 120
      ILEFT = MINO(NXX(N,I),NXX(N+1,I))
      IRIGHT = MAXO(NXX(N,I),NXX(N+1,I))
      DO 130 J = ILEFT,IRIGHT
        NAPRAY(J) = NCHAR(I)
      CONTINUE
130    ICOUNT = ICOUNT + 1
        MAXNXX = MAXO(NXX(N,I),NXX(N+1,I))
        MINNXX = MINO(NXX(N,I),NXX(N+1,I))
        IXPTS(ICOUNT) = MINNXX
        ICOUNT = ICOUNT + 1
        IXPTS(ICOUNT) = MAXNXX
C
      RYMAX = AMAXO(NYY(N,I),NYY(N+1,I))
      RYMIN = AMINO(NYY(N,I),NYY(N+1,I))
      IF (SCAN.LT. RYMIN .OR. SCAN.GT. RYMAX) GO TO 300
C
      C      THE SCAN LINE WILL INTERSECT
      C      THIS SEGMENT
C
      A = SCAN - NYY(N+1,I)
      B = NXX(N+1,I) - NXX(N,I)
      C = NYY(N+1,I) - NYY(N,I)
      IF SEGMENT IS HORIZONTAL,
      IGNORE IT
      IF (NYY(N+1,I).EQ. NYY(N,I)) GO TO 300
      D = A * B/C
C
      C      THE POINT OF INTERSECTION IS
      C      (IPTINT,SCAN)
C
      RPTINT = (D + NXX(N + 1,I)) * 10.0 + 5.0
      IPTINT = RPTINT/10.0
      MAXNXX = MAXO(NXX(N,I),NXX(N+1,I))
      MINNXX = MINO(NXX(N,I),NXX(N+1,I))
C
      C      IF INTERSECTION IS TO THE LEFT OR
      C      RIGHT OF ENDPOINTS, IGNORE IT
C
      IF (IPTINT.GT. MAXNXX .OR. IPTINT.LT. MINNXX) GO TO 300
C
      C      STORE AND COUNT IPTINT
C
      ICOUNT = ICOUNT + 1
      IXPTS(ICOUNT) = IPTINT
      CONTINUE
300

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 PLANP219

[illegible]

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500 CONTINUE
C
  RETURN
  END
  SUBROUTINE MATXSOL
  DIMENSION YY(2), FV(2), FW(2), FVN(400)
  COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,
  $ STA(4), TBLSCW(100), YYCP(4),
  $ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
  $ PHI(100), ZH(100), CP(400), STLOIND(4)
C
  COMMON /TOTTHREE/ CIR(400,2)
C
  COMMON /INSUB23/ APSI, APHI, XX, YYY, ZZ, SNN, TOLC
  DIMENSION WA(20000), INDEX(2)
C
  PART 2 - COMPUTE CIRCULATION TERMS
C
  THE TOLERANCE SET AT THIS POINT IN THE PROGRAM MAY NEED TO BE
  CHANGED FOR COMPUTERS OTHER THAN THE CDC 6000 SERIES
C
  TOLC=(BOT*15.E-05)**2
  DO 10 NV=1,M
    CIR(NV,1)=12.5663704*ALP(NV)
    CIR(NV,2)=12.5663704
  IF (PTEST.NE.0.) CIR(NV,2)=-1.0964155*Q(NV)/BOT
  IF (QTEST.NE.0.) CIR(NV,2)=-1.0964155*PV(NV)*BETA
  CONTINUE
  IZZ=1
  NNV=TBLSCW(IZZ)
  REWIND 10
  DO 70 NV=1,M
    DO 20 I=1,M
      FVN(I)=0.
    IZ=1
    NNN=TBLSCW(IZ)
    DO 60 NN=1,M
      APHI=ATAN(PHI(IZZ))
      APSI=PSI(NN)
      XX=PV(NV)-PN(NN)
      YY(1)=Q(NV)-Q(NN)
      YY(2)=Q(NV)+Q(NN)
      ZZ=ZH(IZZ)-ZH(IZ)
      SNN=S(NN)
      DO 30 I=1,2
        YYY=YY(I)

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PLANP269
PLANP270
PLANP271
PLANP272
MATXSOL2
MATXSOL3
ALL 2
ALL 3
ALL 4
ALL 5
ALL 6
TOTTHRE 2
TOTTHRE 3
MATXSOL6
MATXSOL7
MATXSOL8
MATXSOL9
MATXSOL10
MATXSOL11
MATXSOL12
MATXSOL13
MATXSOL14
MATXSOL15
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MATXSOL36
MATXSOL37
MATXSOL38
MATXSOL39
MATXSOL40
MATXSOL41

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30      CALL INFSUB (BOT,FV(1),FW(1),FUI)
      APHI=-APHI
      APSI=-APSI
      CONTINUE
      IF (PTEST.NE.O.) GO TO 40
      FVN(NN)=FW(1)+FW(2)-(FV(1)+FV(2))*PHI(IZZ)
      GO TO 50
40      FVN(NN)=FW(1)-FW(2)-(FV(1)-FV(2))*PHI(IZZ)
50      IF (NN.LT.NNN.OR.NN.EQ.M) GO TO 60
      IZ=IZ+1
      NNN=NNN+1
      CONTINUE
      DUMB=-CIR(NV,1)
      DUMY=-CIR(NV,2)
      WRITE(10) (FVN(I),I=1,M)
      IF (NV.LT.NNV.OR.NV.EQ.M) GO TO 70
      IZZ=IZZ+1
      NNV=NNV+1
      CONTINUE
70      LWA = 20000
      CALL GIVENS(M,M,2,WA,LWA,CIR,10,11,IERR)
      CALL STINDEX(11,INDEX,1,0)
      END
      SUBROUTINE GIVENS(NR,NC,NS,V,KORE,B,LSEQ,LRAF,FLAG)

C SOLVE A LARGE SET OF LINEAR EQUATIONS OF THE FORM AX = B
C IN A SMALL CENTRAL MEMORY WORKING AREA USING GIVENS
C TRIANGULARIZATION. THE SET MAY BE OVERDETERMINED. THE
C A-MATRIX IS STORED BY ROWS ON A SEQUENTIAL FILE. THE WORK-
C ING AREA IS DYNAMICALLY MANAGED USING A RANDOM ACCESS
C FILE CREATED WITHIN THE MODULE.
C
C ON ENTRY
C NR = NUMBER OF POWS OF A.
C NC = NUMBER OF COLUMNS OF A.
C NS = NUMBER OF RIGHT HAND SIDES.
C W = WORKING AREA ARRAY.
C KORE = LENGTH OF THE WORKING AREA, GE 4 * (NC + NS)
C <IF NEGATIVE, KORE IS SET TO [FL - W(1)] >
C B = NP * NS MATRIX OF RIGHT HAND SIDES.
C LSEQ = LOGICAL UNIT NUMBER OF FILE HOLDING ROWS OF A.
C LRAF = LOGICAL UNIT NUMBER OF RANDOM ACCESS FILE.
C (GIVENS SUBINDEXES THIS FILE. THE USER
C MUST RESTORE THE MASTER INDEX UPON RETURN)
C
C UPON RETURN
C B HOLDS THE NS SOLUTION VECTORS, EACH STORED OVER THE
C FIRST NC WORDS OF THE CORRESPONDING COLUMN.
C FLAG = 0, EXECUTION PROCEEDED NORMALLY.

```

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MATXS042
MATXS043
MATXS044
MATXS045
MATXS046
MATXS047
MATXS048
MATXS049
MATXS050
MATXS051
MATXS052
MATXS053
MATXS054
MATXS055
MATXS056
MATXS057
MATXS058
MATXS059
MATXS060
MATXS061
MATXS062
MATXS063
MATXS064
GIVENS 2
GIVENS 3
GIVENS 4
GIVENS 5
GIVENS 6
GIVENS 7
GIVENS 8
GIVENS 9
GIVENS10
GIVENS11
GIVENS12
GIVENS13
GIVENS14
GIVENS15
GIVENS16
GIVENS17
GIVENS18
GIVENS19
GIVENS20
GIVENS21
GIVENS22
GIVENS23
GIVENS24
GIVENS25
GIVENS26
GIVENS27

```

```

C      = -, WORKING AREA TOO SMALL, -(FLAG) WORDS NEEDED.
C      = +, MATRIX SINGULAR, FLAG = NUMBER OF SINGULAR ROW
C
C C.W.BOLZ AND R.W.HAMM, COMPUTER SCIENCES CORP., 1974.
C
      DIMENSION M(KORE),R(400,2)
      COMMON/HOATAR/ KM,NBLK,NRAF,NROWS,NSEQ,TOL
      INTEGER FLAG
C INITIALIZE
      DO 4 I=1,KORE
      4  M(I)=0.0
      FLAG=0
      NSEQ=LSEQ
      NRAF=LRAF
      NCS=NC+NS
      TOL=1.0E-100
C SIZE THE WORKING AREA.
      NWA=4*(NC+NS)
      IF(KORE.GE.NWA) GO TO 1
      FLAG=-NWA
      RETURN
C
C PARTITION THE WORKING AREA. NROWS IS NUMBER OF ROWS TO BE
C PROCESSED PER REDUCTION PASS. KM IS PORTION OF WORKING
C AREA AVAILABLE FOR HOUSEKEEPING ARRAYS AND SCRATCH STORAGE
C NBLK IS NUMBER OF BLOCKS OF THE TRIANGULARIZED MATRIX.
      1 NROWS=KORE/(2*NCS)+1
      2 KM=KORE-NROWS*NCS
      NBLK=NC*NCS/(2*KM)+1
      KM=KM-3*NBLK-2
      IF(KM.GE.2*NCS-1) GO TO 3
      NROWS=OVERESTIMATED. REDUCE.
      NROWS=NROWS-1
      GO TO 2
C
C COMPUTE PARTITION (BLOCK) SIZES AND NUMBER OF ROWS IN EACH
      3 NB=NBLK
      CALL BLOCKR(W,NB,NC,NCS,KORE)
C
C NOW DYNAMICALLY ALLOCATE THE WORKING AREA...
      LB=1
      LI=LB+2*NBLK
      LR=LI+NBLK+2
      LW=LR+NROWS*NCS
C
C SUBINDEX THE PANDOM ACCESS FILE.
      CALL STINDX(NRAF,W(LI),NBLK+1,0)
C
C AND SOLVE USING GIVENS TRIANGULARIZATION.

```

GIVENS28
 GIVENS29
 GIVENS30
 GIVENS31
 GIVENS32
 GIVENS33
 GIVENS34
 GIVENS35
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 GIVENS70
 GIVENS71
 GIVENS72
 GIVENS73
 GIVENS74
 GIVENS75
 GIVENS76

```

      CALL TRIANG(W(LB),W(LR),V(LW),B,NR,NC,NS,NCS,NBLK,
      NROWS,KM,FLAG)
      RETURN
      END
      SUBROUTINE BLOCKR(LBLK,NB,NC,NCS,KORE)
      C COMPUTE SIZE OF, NUMBER OF ROWS IN, EACH PARTITION
      C OF THE TRIANGULARIZED MATRIX.
      C
      C R.W.HAMM, COMPUTER SCIENCES CORP., 1974
      C
      DIMENSION LBLK(2,NB)
      COMMON/HOATAR/ KM,NBLK,NRAF,NROWS,NSEO,TOL
      1 NT=NCS
      IC=NC
      NBLK=0
      MAXW=0
      C LOOP ON BLOCKS OF MATRIX.
      2 NBLK=NBLK+1
      KW=KM
      LNG=0
      IRW=0
      C COMPUTE ROWS WITHIN BLOCK. IRW = NO.OF ROWS, LNG = LENGTH
      C OF BLOCK, NT = LENGTH OF ROW I, KW = WORDS REMAINING.
      DO 3 I=1,IC
      KW=KW-NT
      IF(KW.LT.0) GO TO 4
      LNG=LNG+NT
      IRW=IRW+1
      NT=NT-1
      3 CONTINUE
      4 LBLK(1,NBLK)=LNG
      LBLK(2,NBLK)=IRW
      MAXW=MAXO(LNG,MAXW)
      C NEXT BLOCK
      IC=IC-IRW
      IF(IC.GT.0) GO TO 2
      C TEST IF WORKING AREA OVERRUN. MAXW = MAX.PARTITION SIZE.
      NWDOS=MAXW+NCS+NPOS+3*NBLK+2
      NDIF=KORE-NWDOS
      IF(NDIF) 5,7,6
      C WORKING AREA OVERRUN. REDUCE BLOCK SIZE AND REITERATE.
      5 KM=KM+NDIF
      GO TO 1

```

GIVEN577
 GIVEN578
 GIVEN579
 GIVEN580
 GIVEN581
 GIVEN582
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 GIVEN618
 GIVEN619
 GIVEN620
 GIVEN621
 GIVEN622
 GIVEN623
 GIVEN624
 GIVEN625

```

C UNDERESTIMATED. CAN WE INCREASE NROWS...
6 NROWS=NROWS+NDIF/NCS
  KM=MAXV
7 RETURN
  END
  SUBROUTINE TRIANG(LBLK,ROWS,W,B,NR,NC,NS,NCS,NB,
    NRW,KMAX,FLAG)
C
C TRIANGULARIZE THE AUGMENTED MATRIX USING PLANE
C ROTATIONS, THEN BACK-SUBSTITUTE FOR SOLUTIONS.
C
C R.W.HAMM AND C.W.BOLZ, COMPUTER SCIENCES CORP., 1974.
C
  DIMENSION LBLK(2,NB),ROWS(NCS,NRW),W(KMAX),B(400,2)
  COMMON/HDATA/ KM,NBLK,NRAF,NROWS,NSEQ,TOL
  INTEGER FLAG
C
C INITIALIZE SCRATCH FILE NRAF TO SIMPLIFY LATER LOGIC.
  DO 1 I=1,KM
    1 W(I)=0.0
    DO 2 N=1,NBLK
      LNG=LBLK(1,N)
      CALL WPITMS(NRAF,W,LNG,N)
    2 CONTINUE
    NCSP=NCS+1
    IR=0
C
C TRIANGULARIZATION LOOP. READ NROWS ROWS PER PASS, AUGMENT.
  REMIND NSEQ
  DO 15 IRSET=1,NR,NROWS
    DO 4 JRR=1,NROWS
      CALL BUFIN(NSEQ,ROWS(1,JRR),NC,IEOF)
      IF(IEOF.EQ.0) GO TO 5
      JR=JRR
      IS=IR+JR
      DO 3 J=1,NS
        NCJ=NC+J
        ROWS(NCJ,JR)=B(IS,J)
      3 CONTINUE
      4 IRR=IR+JR
      IR=IRR+1
      NRWZ=0
C
C READ IN A BLOCK AND SET ITS ROW INDICES.
    DO 13 I=1,NBLK
      LNG=LBLK(1,N)
      NPWA=NRWZ+1
      NRWZ=LBLK(2,N)+NRWZ
      IF(I*IR.LT.NPWA) GO TO 14

```

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GIVEN126
GIVEN127
GIVEN128
GIVEN129
GIVEN130
TRIANG 2
TRIANG 3
TRIANG 4
TRIANG 5
TRIANG 6
TRIANG 7
TRIANG 8
TRIANG 9
TRIANG10
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TRIANG40
TRIANG41
TRIANG42
TRIANG43
TRIANG44
TRIANG45

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```

C      CALL READMS(NRAF,W,LNG,N)
C      C LOOP THROUGH ROWS READ IN THIS PASS,
      LR=0
      DO 12 I=IR,IRR
        LR=LR+1
        JJ=1
C      C AND PROCESS ROWS OF THIS BLOCK.
        DO 11 J=NRWA,NRWZ
          IF(I-J) 12,9,6
          AB=ROWS(J,LR)
          IF(ABS(AB).LT.TOL) GO TO 8
          R=SQRT(AB*AB+W(JJ)*W(JJ))
          IF(R.LT.TOL) GO TO 8
          C=W(JJ)/R
          S=AB/R
          W(JJ)=R
          JP=J+1
          KK=JJ+1
6
C      C ELIMINATE ELEMENT J OF INPUT ROW.
          DO 7 K=JP,NC5
            T=C*W(KK)+S*ROWS(K,LR) - S*W(KK)
            ROWS(K,LR)=C*ROWS(K,LR) - S*W(KK)
            W(KK)=T
            KK=KK+1
7          CONTINUE
C      C
8          CONTINUE
          JJ=JJ+NCSP-J
          GO TO 11
C      C ELEMENTS J TO NCS OF ROWS TO WORKING AREA.
9          KK=JJ
          P=SIGN(1.0,ROWS(J,LR))
          DO 10 K=J,NC5
            W(KK)=P*ROWS(K,LR)
            KK=KK+1
10         CONTINUE
11         CONTINUE
12         CALL WRITMS(NRAF,W,LNG,N,1)
13         CONTINUE
14         IR=IRR
15         CONTINUE
C      C BACK-SUBSTITUTE FOR SOLUTION
      CALL SOLVER(LBLK,W,B,NR,NC,NS,NB,KMAX,FLAG)
      RETURN
      END
      SUBROUTINE SOLVER(LBLK,W,B,NR,NC,NS,NB,KMAX,FLAG)

```

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TRIANG46
TRIANG47
TRIANG48
TRIANG49
TRIANG50
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TRIANG87
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TRIANG89
TRIANG90
TRIANG91
TRIANG92
TRIANG93
SOLVER 2

```

```

C
C BACK SUBSTITUTION FOR X. STORE OVER B.
C
C C.W.BOLZ AND R.W.HAMM, COMPUTER SCIENCES CORP., 1974.
C
C DIMENSION LBLK(2,NB),W(KMAX),B(400,2)
COMMON/HDATAR/ KM,NBLK,NRAF,NROWS,NSEQ,TOL
INTEGER FLAG
C
C LE=0
C NK=NBLK+1
C NSP=NS+1
C MR=NC
C
C BLOCK LOOP. LAST BLOCK FIRST.
C DO 6 NN=1,NBLK
C   N=NK-NN
C   LNG=LBLK(1,N)
C   CALL READMS(NRAF,W,LNG,N)
C   NRW=LBLK(2,N)
C   KK=LNG-NS
C
C ROWS IN THIS BLOCK.
C DO 5 I=1,NRW
C   DO 1 JS=1,NS
C     KKS=KK+JS
C     1 B(MR,JS)=W(KKS)
C
C SOLUTIONS
C DO 4 JS=1,NS
C   TEPM=0.0
C   LN=LE
C   LR=NC
C   JJ=KK
C
C SUBSTITUTION
C 2 IF(LN.EQ.0) GO TO 3
C   TERM=TERM+W(JJ)*B(LR,JS)
C   LR=LR-1
C   LN=LN-1
C   JJ=JJ-1
C   GO TO 2
C
C TEST FOR SINGULAR ROW
C 3 IF(W(JJ).LT.TOL) GO TO 7
C 4 B(LP,JS)=(B(LP,JS)-TERM)/W(JJ)
C   MP=MP-1
C   LE=LE+1
C   KK=JJ-NSP
C   5 CONTINUE
C   6 CONTINUE
SOLVER 3
SOLVER 4
SOLVER 5
SOLVER 6
SOLVER 7
SOLVER 8
SOLVER 9
SOLVER 10
SOLVER 11
SOLVER 12
SOLVER 13
SOLVER 14
SOLVER 15
SOLVER 16
SOLVER 17
SOLVER 18
SOLVER 19
SOLVER 20
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SOLVER 41
SOLVER 42
SOLVER 43
SOLVER 44
SOLVER 45
SOLVER 46
SOLVER 47
SOLVER 48
SOLVER 49
SOLVER 50
SOLVER 51

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SOLVER52  
SOLVER53  
SOLVER54  
SOLVER55  
SOLVER56  
SOLVER57  
BUFFIN 2  
BUFFIN 3  
BUFFIN 4  
BUFFIN 5  
BUFFIN 6  
BUFFIN 7  
BUFFIN 8  
BUFFIN 9
```

```
RETURN  
C ERROR  
7 FLAG=LR  
RETURN  
END  
SUBROUTINE BUFFIN(LUN,A,N,IER)  
DIMENSION A(1)  
IER=N  
READ(LUN) (A(I),I=1,N)  
IF(EOF(LUN)) 1,2  
1 IER=0  
2 RETURN  
END
```

```

SUBROUTINE AERODYN
C
C
  DIMENSION CLCC(400,2), CH(2,100), SUM(3), YCP(4), CROLL(4),
  $ AC(4), CLCL(2,100), P(400), SHDAD(2,100),
  $ SLDT(100), SMLD(2,100), SCT(5), SAT(5)
  COMMON /ALL/ BOT, BOTSV(4), M, BETA, PIEST, QTEST,
  $ STA(4), TBLSCW(100), YYCP(4),
  $ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
  $ PHI(100), ZH(100), CP(400), STLPIND(4)
C
  COMMON /TOTTHREE/ CIR(400,2)
C
  COMMON /THREFOR/ CCAV(2,100), CLT, CLNT, NSSV, ALPD
C
  COMMON /ONETHRE/ T4IST(4), CREF, SREF, CAVE, CLDES, STRUE, AR,
  $ ARTPUE, RICOHT(4), CONFIG(2), NSSWSV(4),
  $ MSV(4), KBOT, PLAN, IPLAN, MACH,
  $ SSWWA(100), XL(4), XT(4), CLWB, CMCL, CLA(4), BLAIR(100),
  $ CLAMAR(4), CLWNG(4), CLWNG(4), XLOCIN,
  $ YINNER(4), YOUTER(4)
C
  INTEGER CONFIG
C
  COMMON /THRECDI/ SLOAD(3,100)
  COMMON /INSUB23/ APSI, API, XXX, YYY, ZZZ, SNN, TOLC
  COMMON /MAINONE/ ICODEOF, TOTAL, AAN(4), XS(4), YS(4), KECTS(4),
  $ XREG(25,4), YREG(25,4), AREG(25,4), DIH(25,4), MCD(25,4),
  $ XX(25,4), YY(25,4), AS(25,4), TTWD(25,4), MMCD(25,4), AN(4),
  $ ZZ(25,4), ITIPCOD, ICAMTST
C
  DIMENSION NUMBER(4)
  DATA NUMBER/5HFIRST, 6HSECOND, 5HTHIRD, 6HFORTH/
C
  PART 3 - COMPUTE OUTPUT TERMS
C
  CLSAVE=CLDES
  IF (CLDES.EQ.100.) CLDES=1.

```

AERODYN2
 AERODYN3
 AERODYN4
 AERODYN5
 AERODYN6
 AERODYN7
 ALL 2
 ALL 3
 ALL 4
 ALL 5
 ALL 6
 TOTTHRE 2
 TOTTHRE 3
 THREFOR2
 THREFOR3
 ONETHRE2
 ONETHRE3
 ONETHRE4
 ONETHRE5
 ONETHRE6
 ONETHRE7
 ONETHRE8
 ONETHRE9
 ONETHR10
 AERODY12
 AERODY13
 MAINONE2
 MAINONE3
 MAINONE4
 MAINONE5
 MAINONE6
 AERODY15
 AERODY16
 AERODY17
 AERODY18
 AERODY19
 AERODY20
 AERODY21
 AERODY22
 AERODY23

```

RAD=57.29578
ALREF=1
QINF=1.
TWST = 0.0
NSSW = 0
DO 5 IT = 1,IPLAN
TWST = TWST + TWIST(IT)
NSSW = NSSW + NSSWSV(IT)
CLWG(IT) = 0.0
CLAMAR(IT) = 0.0
CLWIN(IT) = 0.0
YCP(IT) = 0.0
CROLL(IT) = 0.0
YYCP(IT) = 0.0
CONTINUE
5
C
C
C
C
PART 3 - SECTION 1
COMPUTE LIFT AND PITCHING MOMENT HERE
IZ=1
NNN=IBLSCHW(IZ)
DO 10 I=1,M
P(I)=S(I)*COS(ATAN(PHI(IZ)))
IF (I.LT.NNN.OR.I.EQ.M) GO TO 10
IZ=IZ+1
NNN=NNN+IBLSCHW(IZ)
CONTINUE
10
IT = 1
SUM(1) = 0.0
SUM(2) = 0.0
SUM(3) = 0.0
MSUM = MSV(IT)
DO 20 I=1,M
SUM(1) = SUM(1) + CIR(I,1) * P(I)
SUM(2) = SUM(2) + CIR(I,2) * P(I)
SUM(3) = SUM(3) + CIR(I,2) * P(I) * Q(I)
IF (I.EQ.MSUM) GO TO 15
GO TO 20
15
CLWG(IT) = SUM(1) * 8. / SREF
CLWIN(IT) = SUM(2) * 8. / SREF
CPOLL(IT) = SUM(3) * 8. / (SREF*80T)
IT = IT + 1
MSUM = MSUM + MSV(IT)
CONTINUE
20
CLT=8.*SUM(1)/SREF
CLNT=3.*SUM(2)/SREF
IF (K80T.EQ.1) GO TO 30
CLWNGT = CLWG(K80T) - CLWG(K80T-1)
CLWING = CLWIN(K80T) - CLWIN(K80T-1)

```

```

30  CLWNGT = CLWNG(1)
35  CLWNG = CLWIN(1)
    CRL = 0.0
    DO 40 I=1,M
      CRL=CRL+(Q(I)*CIR(I,2)*2.*P(I))*2.
      CLCC(I,1)=CIR(I,1)*2/CAVE
      CLCC(I,2)=CIR(I,2)*2/CAVE
40  C
    C
    C
    COMPUTE CLP
    CLP=CRL/(SREF*BOT*0.08725)
    CLA(2)=CLNT
    DO 120 IAX=1,2
      SA=S3-SC=0.
      I=0
      IT = 0
      JB=NSSWSV(1)
      JA=1
50  CONTINUE
      DO 70 JSSW=JA,JB
        SD=SE=0.
        SLOAD(IXX,JSSW)=0
        NSCW=TBLSCH(JSSW)
        DO 70 JSCW=1,NSCW
          IF (TWST.EQ.0..AND.IXX.EQ.1) GO TO 60
          I=I+1
          SA=SA+CIR(I,IXX)*P(I)
          SB=SB+CIR(I,IXX)*Q(I)*P(I)
          SC=SC+CIR(I,IXX)*PN(I)*P(I)*BETA
          SLOAD(IXX,JSSW)=SLOAD(IXX,JSSW)+(BOT*CIR(I,IXX)*1.)/(2.*SUM
            1(IXX))
          AA = P(I) / S(I)
          SD = SD + CIR(I,IXX) * AA
          SE = SE + CIR(I,IXX) * PN(I) * BETA * AA
          IF (JSCW.NE.NSCW) GO TO 70
          SMOAD(IXX,JSSW)=SE
          SMLD(IXX,JSSW)=SD
          GO TO 70
60  SLOAD(1,JSSW)=SMOAD(1,JSSW)=SMLD(1,JSSW)=0.
70  CONTINUE
      IT = IT + 1
      IF (JSSW.GE.NSSW) GO TO 80
      JA = JA + NSSWSV(IT)
      JB = JB + NSSWSV(IT + 1)
      IF (IAX.EQ.1) GO TO 50
      IF (IT.EQ.1) GO TO 75
      SCT(IT) = SC
      SAT(IT) = SA

```

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AERODY73
AERODY74
AERODY75
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AEROD115
AEROD116
AEROD117
AEROD118
AEROD119
AEROD120
AEROD121

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```

CLAMAR(IT) = (SCT(IT) - SCT(IT-1)) / ((SAT(IT) - SAT(IT-1)) * CREF) AEROD122
GOTO 50 AEROD123
75 SCT(IT) = SC AEROD124
SAT(IT) = SA AEROD125
CLAMAR(IT) = SCT(IT) / (SAT(IT) * CREF) AEROD126
GOTO 50 AEROD127
80 CONTINUE AEROD128
IF (IXX.EQ.1) GO TO 100 AEROD129
IF (IPLAN.EQ.1) GO TO 90 AEROD130
SCT(IT) = SC AEROD131
SAT(IT) = SA AEROD132
CLAMAR(IT) = (SCT(IT) - SCT(IT-1)) / ((SAT(IT) - SAT(IT-1)) * CREF) AEROD133
GOTO 100 AEROD134
90 CLAMAP(1)=SC/(SA*CREF) AEROD135
CONTINUE AEROD136
100 IF (TWST.EQ.0..AND.IXX.EQ.1) GO TO 110 AEROD137
YCP(1)=S9/(SA*BOT) AEROD138
AC(1)=SC/(SA*CREF) AEROD139
GOTO 120 AEROD140
110 YCP(1)=AC(1)=0. AEROD141
CONTINUE AEROD142
120 CMCL=AC(2) AEROD143
CMO=(AC(1)-AC(2))*CLT AEROD144
AEROD145
AEROD146
AEROD147
AEROD148
AEROD149
AEROD150
AEROD151
AEROD152
AEROD153
AEROD154
AEROD155
AEROD156
AEROD157
AEROD158
AEROD159
AEROD160
AEROD161
AEROD162
AEROD163
AEROD164
AEROD165
AEROD166
AEROD167
AEROD168
AEROD169
AEROD170

PART 3 - SECTION 2
COMPUTE OTHER- AND PRINT ALL FINAL- OUTPUT DATA HERE

DO 140 IXX=1,2
JN=0
DO 140 JSSW=1,NSSW
CH(1)=JSSW=0
NSCW=TBLSW(JSSW)
DO 130 JSCW=1,NSCW
JN=JN+1
CH(1)=JSSW=(-2.0)*(PV(JN)-PN(JN))*BETA+CH(1)=JSSW
CONTINUE
CCAV(1)=CH(1)=JSSW/CAVE
CLCL(1)=JSSW=SLOAD(1)=JSSW/CCAV(1)=JSSW
CONTINUE
CLD=CLD=0
IF (CLD.EQ.1) CLD=1.
DO 150 I=1,M
CP(1)=CLCC(1)=CLCC(I,2)*(CLD-CLT)/CLNT)*CAVE/(2.0*(PN(I)-PV(I))*BETA)
CONTINUE
WRITE (6,240) CONFIG
IF (PTEST.NE.0.) WRITE (6,350)
IF (PTEST.NE.0.) WRITE (6,330)
IF (PTEST.EQ.0..AND.QTEST.EQ.0.) WRITE (6,340)

```



```

C
C
C
      COMPUTE INDUCED DRAG FOR FLAT WING-BODY WITH NO DIHEDRAL
      NSV = 1
      MTOT = 1
      DO 190 IT = 1, KBOT
      MTOT = MTOT + MSV(IT)
      NSV = NSV + NSSWSV(IT)
180  CONTINUE
      CALL CDICLS(AR,ARTRUE,NSSWSV(KBOT),MTOT,NSV,COI,CDIT)
      CLAPD=CLA(2)/57.29578
      ALPO=-(CLT/CLA(2))*57.29578
      ALPD=CLDES/CLAPD+ALPO
      ALPW=1./CLAPD
      CLWB=CLWING*ALPD/57.29578+CLWNGT
      COIW3=COI/(CLWB*CLWB)
      IF (IUTK.EQ.1) WRITE (6,250) HEAD,CDIT
      WRITE (6,260) CLDES,ALPD,CLWB,COI,CDIWB
      WRITE (6,290) CLA(2),CLAPD,CLT,ALPO,YCP(2),CMCL,CMO
      YTCP(1) = YCP(2)
      IF (IPLAN.EQ.1) GOTO 194
      DO 193 IT=1,IPLAN
      IF (IT.GT.1) GOTO 191
      CLTWST = CLWNG(1)
      CLALPHA = CLWIN(1)
      YTCP(1) = CPOLL(1) / CLALPHA
      GOTO 192
191  CLTWST = CLWNG(IT) - CLWNG(IT-1)
      CLALPHA = CLWIN(IT) - CLWIN(IT-1)
      YTCP(IT) = (CROLL(IT) - CROLL(IT-1)) / CLALPHA
192  CLAPDIT = CLALPHA / 57.29578
      ALPOIT = - (CLTWST / CLALPHA) + 57.29578
      WRITE(6,400) NUMBER(IT), CLALPHA, CLAPDIT, CLTWST,
      $      ALPOIT, YTCP(IT)
193  CONTINUE
194  CONTINUE
      WRITE (6,300) CLT
      NP=J=0
      DO 210 NV=1,NSSW
      BCLCC=BADLAE=BSLD=0.
      NSCW=TBLSWC(NV)
      NP=NP+1
      NR=NR+NSCW
      DO 200 I=NP,NR
      ADLAE=CLCC(I,2)*CLT/CLNT
      BSLO=CLCC(I,1)-ADLAE
      BCLCC=BCLCC+CLCC(I,1)
      BADLAE=BADLAE+ADLAE
      BSLD=BSLD+BSLO

```

```

AERD220
AERD221
AERD222
AERD223
AERD224
AERD225
AERD226
AERD227
AERD228
AERD229
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AERD231
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AERD253
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AERD255
AERD256
AERD257
AERD258
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AERD260
AERD261
AERD262
AERD263
AERD264
AERD265
AERD266
AERD267
AERD268

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```

200 CONTINUE
   SLDT(NV)=(SMQAD(1,NV)*SMQAD(2,NV)*(CLDES-CLT)/CLNT)/(SMLD(1,NV)+SMAEROD270
   LLD(2,NV)*(CLDES-CLT)/CLNT)
   J=J+NSC#
   YQ=Q(J)/BOT
   NVS = 1
   DO 205 IT = 1,IPLAN
     JJ = IT - 1
     IF(JJ.EQ.0) GOTO 204
     NVS = NVS + NSSWSV(JJ)
204 IF(NV.NE.NVS) GOTO 205
     WRITE(6,310) NUMBER(IT)
     GOTO 206
205 CONTINUE
206 CONTINUE
220 WRITE (6,320) NV,YQ,SLOAD(2,NV),CLCL(2,NV),CCAV(2,NV),8CLCC,8ADLAE
     1,8ASLD,SLOAD(3,NV),SLOT(NV)
     CONTINUE
     CLDES=CLSAVE
     IF(ITIPCOD.EQ.2) CALL FLOWFL

C
C
C
230 FORMAT(/12X,A6,39H PLANFORM HORSESHOE VORTEX DESCRIPTIONS/)
240 FORMAT(1H1,/,/,55X,16HAERODYNAMIC DATA ,/,/,54X,
     $ 16HCONFIGURATION : ,2A10)
250 FORMAT (1H1,18X,22HCOMPLETE CONFIGURATION,31X,25HHING-BODY CHARACTERISTICS/64X,4HLIFT,9X,33HINDUCED DRAG (FAR FIELD SOLUTION)//16XA8AEROD296
     2,21H CL COMPUTED ALPHA,19X,6HCL(WB),7X,13HCOI AT CL(WB),4X,15HCAEROD297
     $DI/(CL(WB))*2),/,88X,16H(1/PI*AR REF) = ,F8.5,1H)
260 FORMAT (11X,2F15.5,15X,3F15.5)
270 FORMAT (///4X,11H REF. CHORD,6X,25HC AVERAGE TRUE AREA ,2X,1AEROD300
     14HREFERENCE AREA,9X,3HB/2,8X,7HREF. AR,8X,7HTRUE AR,4X,11HMACH NUMAEROD301
     2AR/)
280 FORMAT (8F15.5)
290 FORMAT (///4X,38HCOMPLETE CONFIGURATION CHARACTERISTICS//36X,8HCLAEROD304
     1 ALPHA,8X,53HCL(1WIST) ALPHA AT CL=0 Y CP CM/CL
     2/27X,23HPER RADJAN PER DEGREE/24X,7F12.5)
300 FORMAT (//25X,18HADDITIONAL LOADING/24X,24HWITH CL BASED ON S(TRUEAEROD306
     1)71X,11H-AT CL DES-/67X,34HLOAD DUE ADD. LOAD AT BASIC LOAD/27AEROD308
     2SPAN LOAD AT X LOCATION OF/8H STATION/6X,5H 2Y/8X,9H SL COEF ,4XAEROD309
     38HCL RATIO/4X,7HC RATIO,7X,14HTO TWIST CL=F9.5,3X,7HAT CL=05X,2AEROD310
     4HDESIGNEO CL LOCAL CENT PR/)
310 FORMAT(/,47X,A6,32H PLANFORM SPAN LOAD DISTRIBUTION ,/)
320 FORMAT (4X,14,F12.5,5X,3F12.5,3X,3F12.5,3X,2F12.5)
330 FORMAT (/54X,24HCMQ AND CLQ ARE COMPUTED//)
340 FORMAT (/38X,57HSTATIC LONGITUDINAL AERODYNAMIC COEFFICIENTS ARE CAEROD315
     1COMPUTED//)
350 FORMAT (/59X,15HCLP IS COMPUTED//)
     AEROD316
     AEROD317

```



```

C      DIMENSION MMN(4),NUMBER(4)
C      DIMENSION INDEXES(400,2), YFLOW(75), XFLOW(75)
C
C      REAL MACH
C      DATA NUMBER/5HFIRST,6HSECOND,5HTHIRD,6HFORTH/
C
C      NUMSTA = 75
C      NUMSTOT = 75
C      DO 15 J = 1,9
C         DO 10 I = 1,NUMSTA
C            3BREAKPT(I,J) = 0.
C        10   CONTINUE
C       15   CONTINUE
C
C      TOLC = (BOT * 15.E - 05) ** 2
C
C      WRITE (6,630)
C      WRITE (6,620)
C
C      PI = 4. * ATAN(1.)
C      RAD = 180. / PI
C      CONST = 4. * PI
C
C      ALPW AND ALPO USED HEREIN IN RADIANS
C
C      ALPW = 1. / CLNT
C      ALPO = -(CLT / CLNT)
C      ALPW = ALPW * CLDES
C      ALPD = ALPW + ALPO
C
C      GAMP(J,1) IS THE BASIC LOAD INITIALLY
C      GAMP(J,2) IS THE ADDITIONAL LOAD AT CL = 1.
C      GAMP(J,3) IS THE TOTAL LOAD AT CLDES
C      GAMP(J,4) IS THE ADDITIONAL LOAD AT CLDES
C
C      DO 20 I = 1,M
C         GAMP(I,1) = CIR(I,1) + CIR(I,2) * ( - CLT / CLNT )
C         GAMP(I,2) = CIR(I,2) / CLNT
C       20   CONTINUE
C
C      DO 30 K = 1,M
C         GAMP(K,4) = GAMP(K,2) * CLDES
C         GAMP(K,3) = GAMP(K,1) + GAMP(K,4)
C       30   CONTINUE
C

```

```

C
WRITE(6,640) CONFIG
WRITE (6,670) CLDES,CLDES
NN = 1
LMA = TBLSCW(1)
MM(1) = 1
MM(2) = 1 + MSV(1)
MM(3) = 1 + MSV(1) + MSV(2)
MM(4) = 1 + MSV(1) + MSV(2) + MSV(3)
DO 60 I = 1,NSSW
J = NN
J1 = J + 1
DO 40 IT=1,IPLAN
IF(J.EQ.MMM(IT)) WRITE(6,720) NUMBER(IT)
CONTINUE
40 WRITE (6,680) I,Q(J),GAMP(J,1),GAMP(J,2),GAMP(J,3),GAMP(J,4)
C
IF(TBLSCW(I).EQ.1) GO TO 50
50 WRITE (6,690) (GAMP(L,1),GAMP(L,2),GAMP(L,3),GAMP(L,4),L = J1,LMA)
C
IF (I .EQ. NSSW) GO TO 60
LMA=TBLSCW(I)+LMA
50 NN=TBLSCW(I)+NN
60 CONTINUE
LL = 1
MM = 0
K = 0
C
DO 80 ITT = 1,IPLAN
JRANGE = NSSWSV(ITT)
DO 70 J = 1,JRANGE
K = K + 1
MM = MM + TBLSCW(K)
INDEXES(K,2) = MM
XCVLE(K) = .5 * BETA * (3. * PN(LL) - PV(LL))
XCVTE(K) = .5 * BETA * (3. * PV(MM) - PN(MM))
INDEXES(K,1) = LL
YFL(K) = Q(LL)
LL = LL + TBLSCW(K)
70 CONTINUE
80 CONTINUE
C
C SORT THE VALUES IN -YFL- AND REMOVE THE DUPLICATES. NOTE
C FOR DUPLICATES, THEIR XLE AND XTE VALUES ARE STORED IN THE
C NEXT AVAILABLE COLUMNS OF THE -BREAKPT- MATRIX
C

```

FLOWFL64
 FLOWFL65
 FLOWFL66
 FLOWFL67
 FLOWFL68
 FLOWFL69
 FLOWFL70
 FLOWFL71
 FLOWFL72
 FLOWFL73
 FLOWFL74
 FLOWFL75
 FLOWFL76
 FLOWFL77
 FLOWFL78
 FLOWFL79
 FLOWFL80
 FLOWFL81
 FLOWFL82
 FLOWFL83
 FLOWFL84
 FLOWFL85
 FLOWFL86
 FLOWFL87
 FLOWFL88
 FLOWFL89
 FLOWFL90
 FLOWFL91
 FLOWFL92
 FLOWFL93
 FLOWFL94
 FLOWFL95
 FLOWFL96
 FLOWFL97
 FLOWFL98
 FLOWFL99
 FLOWFL100
 FLOWFL101
 FLOWFL102
 FLOWFL103
 FLOWFL104
 FLOWFL105
 FLOWFL106
 FLOWFL107
 FLOWFL108
 FLOWFL109
 FLOWFL110
 FLOWFL111
 FLOWFL112

```

C      NUMSTA = K
C      CALL HEAPSR
C      NUMPNTS = NUMSTA
C      REMAIND = NUMSTOT - NUMPNTS
C      DELTAY = 2.*BOT/REMAIND
C
C      READ IN TOTAL NUMBER OF FIELD LINES
C
C      READ(5,610) TOTFL
C
C      NTOTFL = TOTFL
C      DO 160 NTOT = 1,NTOTFL
C      RSTA = NTOT
C      SKIPIT = .FALSE.
C
C      F I E L D   L I N E   D E S C R I P T I O N
C
C      XDOWN IS THE X LOCATION AT THE PLANE OF SYMMETRY
C
C      SWEP IS THE SWEEP ANGLE IN DEGREES
C
C      ZREF IS THE Z LOCATION AT THE PLANE OF SYMMETRY
C
C      DIHED IS THE DIHEDRAL ANGLE IN DEGREES
C
C      READ (5,610) XDOWN,SWEP,ZREF,DIHED
C      XDOWN=XDOWN-XLOCIN
C
C      TANFL = TAN(SWEP / RAD)
C      TOIHED = TAN(DIHED / RAD)
C      WRITE (6,650)
C      ALPWW = ALPO * RAD
C
C      WRITE(6,660) RSTA,XDOWN,SWEP,ZREF,DIHED,ALPWW,CIDES
C      WRITE (6,700)
C
C      NOTE THAT THE PLANFORM OF MAXIMUM
C      SEMI-SPAN MUST EXTEND TO THE PLANE
C      OF SYMMEITY.
C      DO 110 I = 1,NUMPNTS
C      OFFPLAN(I) = .TRUE.

```

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FLOWF162
FLOWF163
FLOWF164
FLOWF165
FLOWF166
FLOWF167
FLOWF168
FLOWF169
FLOWF170
FLOWF171
FLOWF172
FLOWF173
FLOWF174
FLOWF175
FLOWF176
FLOWF177
FLOWF178
FLOWF179
FLOWF180
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FLOWF196
FLOWF197
FLOWF198
FLOWF199
FLOWF200
FLOWF201
FLOWF202
FLOWF203
FLOWF204
FLOWF205
FLOWF206
FLOWF207
FLOWF208
FLOWF209
FLOWF210

YFLOW(I) = YFL(I)
XFLOW(I) = YFL(I) * TANFL + XDOWN
DO 100 J=1,4
  JT2 = J * 2
  JT2P1 = J * 2 + 1
  XLE = BREAKPT(I,JT2)
  XTE = BREAKPT(I,JT2P1)
  IF((ABS(XLE) .LT. ABS(XFLOW(I))) .AND.
    $ (ABS(XFLOW(I)) .LT. ABS(XTE))) OFFPLAN(I) = .FALSE.
100 CONTINUE
110 CONTINUE
115 J = 0
  K = NUMPTS + 1
  DO 120 I = K, NUMSTOT
    J = J + 1
    OFFPLAN(I) = .TRUE.
    YFLOW(I) = -BOT - J*DELTAY
    XFLOW(I) = YFLOW(I) * TANFL + XDOWN
120 CONTINUE

DO 150 II = 1, NUMSTOT
  IF (OFFPLAN(II)) GO TO 125
  ZFL = ZREF + YFLOW(II) * TOIHED
  WRITE(6,730) II, XFLOW(II), YFLOW(II), ZFL
  SKIPIT = .TRUE.
  GO TO 150
125 WOU = 0.
  ZFL = ZREF + YFLOW(II) * TOIHED
  UOU = 0
  VOU = 0
  DWOUA = 0
  DUOUA = 0
  XFLOW(II) = XFLOW(II) / BETA
  IZ = 1
  NNN = TBLSCW(IZ)
  DO 140 NN = 1, M
    SNN = S(NN)
    XXX = XFLOW(II) - PN(NN)
    ZZZ = ZFL - ZH(IZ)
    APHI = ATAN(PHI(IZ))
    APSI = PSI(NN)
    YA(1) = YFLOW(II) - Q(NN)
    YA(2) = YFLOW(II) + Q(NN)
  DO 130 I = 1, 2

```

C
C
C
C
C
C

```

YYY = YA(I)
CALL INFSUB (BOT,FV(I),FW(I),FU(I))
APHI = -APHI
APSI = -APSI

130 CONTINUE
WOU = WOU + (FW(1) + FW(2)) * GAMP(NN,3) / CONST
VOU = VOU + (FV(1) + FV(2)) * GAMP(NN,3) / CONST
UOU = UOU + (FU(1) + FU(2)) * GAMP(NN,3) / CONST
DWOUDA = DWOUDA + (FW(1) + FW(2)) * GAMP(NN,4) / (CONST * ALPH)
DUODA = DUODA + (FU(1) + FU(2)) * GAMP(NN,4) / (CONST * ALPH)
IF (NN.LT. NNN .OR. NN.EQ. M) GO TO 140
IZ = IZ + 1
NNN = NNN + TBLSCW(IZ)

140 CONTINUE
CA = COS(ALPD)
SA = SIN(ALPD)
EPSILN = (ATAN((WOU * CA - UOU * SA) / (1. - WOU * SA - UOU * CA)))FLOWF227
1) * RAD
DEPDAL = (WOU ** 2 + UOU ** 2 - (WOU + DUODA) * SA + (DWOUDA - UOFLWF229
1U) * CA + WOU * DUODA - UOU * DWOUDA) / (1. - 2. * (WOU * SA + UOFLWF230
2U * CA) + UOU ** 2 + WOU ** 2)
QRATI = 1. - 2. * (UOU * CA + WOU * SA) + UOU ** 2 + VOU ** 2 + WOFLWF232
1U ** 2
SMACH = MACH * MACH
QRATIO = QRATI * (1. + .2 * SMACH * (1. - QRATI)) ** 2.5
DWNWH = ATAN(WOU) * RAD
SIGMA = ATAN(UOU) * RAD
XFLOW(II) = XFLOW(II) * BETA

WRITE(6,710)II,XFLOW(II),YFLOW(II),ZFL,WOU,VOU,UOU,DWNWH,EPSILN
1,DEPDAL,QRATIO,SIGMA

C
150 CONTINUE
IF (SKIPIT) WRITE(6,740)
160 CONTINUE
RETURN

C
610 FORMAT (8F10.6)
620 FORMAT (1H)
630 FORMAT (1H)
IN F J P M A T I G N)
640 FORMAT (//,58X,16MCONFIGURATION : ,2A10)
650 FORMAT (1H)
113-S-EEP,DEGREES,6X,1HZ,4X,16HDIHEDRAL,DEGREES,2X,13HALPHA,DEGREESFLOWF254
2,3X,7H-ING CL)
660 FORMAT (15X,F12.0,4F12.5,6X,2F12.5)
670 FORMAT (30X,7HSTATION,4X,3H Y ,26X,7HGAMMA/U/,53X,41HBASIC ADDITIFLOWF257
1CNAL
TOTAL ADDITIONAL/,48X,31HAT CL = 0. AT CL = 1.0 AT CLFLOWF258
2 =,F5.2,2X,7HAT CL =,F5.2)

```



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680 FORMAT (29X,I5,5F12.5)
690 FORMAT (46X,4F12.5)
700 FORMAT (/ ,80X,8HODNNWASH,/ ,1X,5HFIELD,6X,1HX,11X,1HY,11X,1HZ,11X,10HDIEPSILON,FLOWF262
13HW/U,9X,3HV/U,9X,3HU/U,6X,6HANGLE,5X,8HEPSTLON,1X,10HDEGREEFLOWF263
2),2X,8HOC(LOCAL),5X,6HSIGMA,/ ,1X,5HPPOINT,74X,7HDEGREES,5X,7HDEGREEFLOWF264
3S,3X,RHD(ALPHA),4X,GHO(INF),5X,7HDEGREES,/ )
710 FORMAT(1X,I3,1H,,6F12.5,5F11.5)
720 FORMAT(/ /42X,32HINPUT VALUES OF CIRCULATION FOR ,A10,
      * 8HPLANFORM// )
730 FORMAT(1X,I3,1H,,3F12.5,12X,5H***** )
740 FORMAT(10H,5X,28H***** - COMPUTATION OMITTED ,
$ 41HQUE TO THE POINT PROJECTION LYING WITHIN ,
$ 26HTHE CONFIGURATION BOUNDARY ,/)
END
SUBROUTINE HEAPSRT
      THIS ROUTINE PERFORMS A HEAPSORT ON THE VALUES IN
      THE YFL ARRAY, AND MOVES THE CORRESPONDING XLE AND XTE
      VALUES WHEN A SWAP IS MADE. THIS ALGORITHM IS DESCRIBED
      IN N. WIRTH'S "ALGORITHMS + DATA STRUCTURES - PROGRAMS",
      BOOK, 1976, PRENTICE-HALL, INC.
      COMMON / HEAP / BREAKPT(75,9), INDEXL, INDEXR, NUMSTA, NUMSTOT
      DIMENSION YFL(75), XCVCLE(75), XCVCLE(75)
      EQUIVALENCE (BREAKPT(1,1),YFL(1)),
      $ (BREAKPT(1,2),XCVCLE(1)),
      $ (BREAKPT(1,3),XCVCLE(1))
      COL1 COL2 COL3 COL4 COL5 COL6 COL7 COL8 COL9
      YFL XLE1 XTE1 XLE2 XTE2 XLE3 XTE3 XLE4 XTE4
      N = NUMSTA
      INDEXL = (N / 2) + 1
      INDEXR = N
      10 CONTINUE
      IF (INDEXL .LE. 1) GO TO 20
      INDEXL = INDEXL - 1
      CALL SIFT
      GO TO 10
20 CONTINUE
      IF ( INDEXR .LE. 1) GO TO 30
      Y1 = YFL(1)
      X1 = XCVCLE(1)

```

```

X2 = XCVTE(1)
YFL(1) = YFL(INDEXR)
XCVLE(1) = XCVLE(INDEXR)
XCVTE(1) = XCVTE(INDEXR)
YFL(INDEXR) = Y1
XCVLE(INDEXR) = X1
XCVTE(INDEXR) = X2
INDEXR = INDEXR - 1
CALL SIFT
GO TO 20

30 CONTINUE
C REMOVE THE DUPLICATE -YFL- VALUES BY COMBINING THEIR
C XLE AND XTE VALUES IN -BREAKPT-
C
N = 1
KK = 4
TEMP1 = YFL(1)
DO 50 J=2,NUMSTA
IF(TEMP1 .EQ. YFL(J)) GO TO 40
KK = 4
N = N + 1
YFL(N) = YFL(J)
XCVLE(N) = XCVLE(J)
XCVTE(N) = XCVTE(J)
TEMP1 = YFL(J)
GO TO 50
40 CONTINUE
BREAKPT(N,KK) = XCVLE(J)
BREAKPT(N,KK+1) = XCVTE(J)
KK = KK + 2
50 CONTINUE
NUMSTA = N
RETURN
END

SURROUTINE SIFT
COMMON / HEAP / BREAKPT(75,9), INDEXL, INDEXR, NUMSTA, NUMSTOT
C
DIMENSION YFL(75), XCVLE(75), XCVTE(75)
EQUIVALENCE (BREAKPT(1,1),YFL(1)),
5 (BREAKPT(1,2),XCVLE(1)),
5 (BREAKPT(1,3),XCVTE(1))
C
C COL1 COL2 COL3 COL4 COL5 COL6 COL7 COL8 COL9
C YFL XLE1 XTE1 XLE2 XTE2 XLE3 XTE3 XLE4 XTE4
C
INDEXI = INDEXL
INDEXJ = 2 * INDEXI
Y1 = YFL(INDEXI)
X1 = XCVLE(INDEXI)

```

```

HEAPSR27
HEAPSR28
HEAPSR29
HEAPSR30
HEAPSR31
HEAPSR32
HEAPSR33
HEAPSR34
HEAPSR35
HEAPSR36
HEAPSR37
HEAPSR38
HEAPSR39
HEAPSR40
HEAPSR41
HEAPSR42
HEAPSR43
HEAPSR44
HEAPSR45
HEAPSR46
HEAPSR47
HEAPSR48
HEAPSR49
HEAPSR50
HEAPSR51
HEAPSR52
HEAPSR53
HEAPSR54
HEAPSR55
HEAPSR56
HEAPSR57
HEAPSR58
HEAPSR59
HEAPSR60
SIFT 2
HEAP 2
HEAP 3
HEAP 4
HEAP 5
HEAP 6
HEAP 7
HEAP 8
HEAP 9
HEAP 10
HEAP 11
SIFT 4
SIFT 5
SIFT 6
SIFT 7

```

```

X2 = XCVTE(INDEXI)
10 CONTINUE
  IF (INDEXJ .GT. INDEXR) GO TO 30
  IF (INDEXJ .EQ. INDEXR) GO TO 20
    JPI = INDEXJ + 1
    ARSYJ = ABS(YFL(INDEXJ))
    ABSYJPI = ABS(YFL(JPI))
    IF (ABSYJ .LT. ABSYJPI) INDEXJ = JPI
20 CONTINUE
  IF (ABS(YI) .GE. ABS(YFL(INDEXJ))) GO TO 30
  YFL(INDEXI) = YFL(INDEXJ)
  XCVLE(INDEXI) = XCVLE(INDEXJ)
  XCVTE(INDEXI) = XCVTE(INDEXJ)
  INDEXI = INDEXJ
  INDEXJ = INDEXI + 2
  GO TO 10

C
30 YFL(INDEXI) = YI
  XCVLE(INDEXI) = X1
  XCVTE(INDEXI) = X2
  RETURN
END

SUBROUTINE FTLUP (X,Y,M,N,VARI,VARD)
***DOCUMENT DATE 09-12-69 SUBROUTINE REVISED 07-07-69 *****FTLUP1 3
MODIFICATION OF LIBRARY INTERPOLATION SUBROUTINE FTLUP
DIMENSION VARI(1), VARD(1), V(3), YI(2)
DIMENSION II(43)

C
C INITIALIZE ALL INTERVAL POINTERS TO -1.0 FOR MONOTONICITY CHECK
DATA (II(J),J=1,43)/43*-1/
MA=IABS(M)

C
C ASSIGN INTERVAL POINTER FOR GIVEN VARI TABLE
C THE SAME POINTER WILL BE USED ON A GIVEN VARI TABLE EVERY TIME
LI=MOD(LOC(VARI(1)),43)+1
I=II(LI)
IF (I.GE.0) GO TO 50
IF (N.LT.2) GO TO 60

C
C MONOTONICITY CHECK
IF (VARI(2)-VARI(1)) 20,20,40
C ERROR IN MONOTONICITY
K=LOC(VARI(1))
PRINT 170, J,M,(VARI(J),J=1,N),(VARD(J),J=1,N)
STOP
C MONOTONIC DECREASING
20 DD 30 J=2,N
IF (VARI(J)-VARI(J-1)) 30,10,10
30 CONTINUE

```

SIFT 8
 SIFT 9
 SIFT 10
 SIFT 11
 SIFT 12
 SIFT 13
 SIFT 14
 SIFT 15
 SIFT 16
 SIFT 17
 SIFT 18
 SIFT 19
 SIFT 20
 SIFT 21
 SIFT 22
 SIFT 23
 SIFT 24
 SIFT 25
 SIFT 26
 SIFT 27
 SIFT 28
 SIFT 29
 FTLUP1 2
 FTLUP1 3
 FTLUP1 4
 FTLUP1 5
 FTLUP1 6
 FTLUP1 7
 FTLUP1 8
 FTLUP1 9
 FTLUP110
 FTLUP111
 FTLUP112
 FTLUP113
 FTLUP114
 FTLUP115
 FTLUP116
 FTLUP117
 FTLUP118
 FTLUP119
 FTLUP120
 FTLUP121
 FTLUP122
 FTLUP123
 FTLUP124
 FTLUP125
 FTLUP126
 FTLUP127
 FTLUP128

```

C
40      GO TO 60
      MONOTONIC INCREASING
      DO 50 J=2,N
      IF (VARI(J)-VARI(J-1)) 10,10,50
      CONTINUE
C
50      INTERPOLATION
      IF (I.LE.O) I=1
      IF (I.GE.N) I=N-1
      IF (N.LE.1) GO TO 70
      IF (MA.NE.O) GO TO 80
      ZERO ORDER
      Y=VARD(1)
      GO TO 160
C
80      LOCATE 1 INTERVAL (X(I),LE.X,LT.X(I+1))
      IF ((VARI(I)-X)*(VARI(I+1)-X)) 110,110,90
C
90      IN GIVES DIRECTION FOR SEARCH OF INTERVALS
      IN=SIGN(1.0,(VARI(I+1)-VARI(I))*(X-VARI(I)))
C
100     IF X OUTSIDE ENDPOINTS, EXTRAPOLATE FROM END INTERVAL
      IF ((I+IN).LE.O) GO TO 110
      IF ((I+IN).GE.N) GO TO 110
      I=I+IN
      IF ((VARI(I)-X)*(VARI(I+1)-X)) 110,110,100
      IF (MA.EQ.2) GO TO 120
C
110     FIRST ORDER
      Y=(VARD(I)*(VARI(I+1)-X)-VARD(I+1)*(VARI(I)-X))/(VARI(I+1)-VARI(I))
      GO TO 160
C
120     SECOND ORDER
      IF (N.EQ.2) GO TO 10
      IF (I.EQ.(N-1)) GO TO 140
      IF (I.EQ.1) GO TO 130
      PICK THIRD POINT
      SK=VARD(I+1)-VARI(I)
      IF ((SK*(X-VARD(I-1))).LT.(SK*(VARI(I+2)-X))) GO TO 140
      L=I
      GO TO 150
C
130     L=I-1
      GO TO 150
C
140     L=I-1
      V(1)=VARI(L)-X
      V(2)=VARI(L+1)-X
      V(3)=VARI(L+2)-X
      YY(1)=(VARD(L)*V(2)-VARD(L+1)*V(1))/(VARI(L+1)-VARI(L))
      YY(2)=(VARD(L+1)*V(3)-VARD(L+2)*V(2))/(VARI(L+2)-VARI(L+1))
      Y=(YY(1)*V(3)-YY(2)*V(1))/(VARI(L+2)-VARI(L))
      I(L+1)=I
      RETURN
C
150
160

```

```

FTLUP129
FTLUP130
FTLUP131
FTLUP132
FTLUP133
FTLUP134
FTLUP135
FTLUP136
FTLUP137
FTLUP138
FTLUP139
FTLUP140
FTLUP141
FTLUP142
FTLUP143
FTLUP144
FTLUP145
FTLUP146
FTLUP147
FTLUP148
FTLUP149
FTLUP150
FTLUP151
FTLUP152
FTLUP153
FTLUP154
FTLUP155
FTLUP156
FTLUP157
FTLUP158
FTLUP159
FTLUP160
FTLUP161
FTLUP162
FTLUP163
FTLUP164
FTLUP165
FTLUP166
FTLUP167
FTLUP168
FTLUP169
FTLUP170
FTLUP171
FTLUP172
FTLUP173
FTLUP174
FTLUP175
FTLUP176
FTLUP177

```

```

C
170      FORMAT (1H1.50H TABLE BELOW OUT OF ORDER FOR FTLUP AT POSITION
      175,/31H X TABLE IS STORED IN LOCATION ,06,/(8G15.8))
      END
      SUBROUTINE COICLS (AR,ARTRUE,ISEMSP,MTOT,NSV,COI,COIT)
      DIMENSION ETAN(51), GAMPR(51,1), ETA(41), GAMMA(41), VE(41), B(41)
      COICLS 3
      COICLS 4
      ALL 2
      ALL 3
      ALL 4
      ALL 5
      ALL 6
      COICLS 6
      COICLS 7
      COICLS 8
      COICLS 9
      COICLS 10
      COICLS 11
      COICLS 12
      COICLS 13
      COICLS 14
      COICLS 15
      COICLS 16
      COICLS 17
      COICLS 18
      COICLS 19
      COICLS 20
      COICLS 21
      COICLS 22
      COICLS 23
      COICLS 24
      COICLS 25
      COICLS 26
      COICLS 27
      COICLS 28
      COICLS 29
      COICLS 30
      COICLS 31
      COICLS 32
      COICLS 33
      COICLS 34
      COICLS 35
      COICLS 36
      COICLS 37
      COICLS 38
      COICLS 39
      COICLS 40
      COICLS 41
      COICLS 42

      COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,
      $ STA(4), TBLSCW(100), YYCP(4),
      $ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
      $ PHI(100), ZH(100), CP(400), STLOIND(4)

      COMMON /THRECDI/ SLOAD(3,100)
      DO 10 I=1,41
      DO 10 J=1,41
      FVN(I,J)=0
      SPAN=2.*BOT
      CAVB=SPAN/ARTRUE
      PI=.314159265E+01
      NST=ISEMSP+1
      NN=MTOT
      DO 20 N=1,ISEMSP
      NM=NSV-N
      NSCW=TBLSCW(NM)
      NN=NN-NSCW
      ETAN(N)=ASIN(-Q(NN)*2./SPAN)
      GAMPR(N,1)=SLOAD(3,NM)*CAVB/(2.*SPAN)
      CONTINUE
      ETAN(NST)=PI/2.
      GAMPR(NST,1)=0
      DO 30 NP=1,41
      ANP=NP
      ETA(NP)=(ANP-21.)*PI/42.

      DO 40 JK=21,41
      CALL FTLUP (ETA(JK),GAMMA(JK),1,NST,ETAN,GAMPR)
      CONTINUE
      DO 50 NY=22,41
      NY=42-NY
      ETA(NY)=SIN(ETA(NY))
      ETAN(NY)=-ETA(NY)
      GAMMA(NP)=GAMMA(NY)
      DO 90 NU=21,41
      ANU=1U
      DO 80 N=1,41
      AN=N
      NNUD=IABS(N-NU)
      VE(N)=COS(((AN-21.)*PI)/42.)
      IF (NNUD.NE.0) GO TO 60

```

```

      B(N)=(42.)/(4.0*COS((ANU-21.)*PI/42.))
      GO TO 80
      IF (MOD(NNUD,2).EQ.0) GO TO 70
      B(N)=VE(N)/((42.)*(ETA(N)-ETA(NU))**2)
      GO TO 80
      B(N)=0.0
      CONTINUE
      DO 90 NP=21,41
      NUST=IABS(NU-21)
      IF (NUST.EQ.0) GO TO 90
      IF (MOD(NUST,2).EQ.0) GO TO 90
      NPST=IABS(NP-20)
      IF (MOD(NPST,2).EQ.0) GO TO 90
      NPNUD=IABS(NP-NU)
      IF (NPNUD.EQ.0) GO TO 90
      IF (MOD(NPNUD,2).EQ.0) GO TO 90
      FVN(NU,NP)=2.0*B(NP)/21.*COS((ANU-21.)*PI/42.)
      IT=42-HU
      ITT=42-NP
      FVN(NU,ITT)=2.0*B(ITT)/21.*COS((ANU-21.)*PI/42.)
      FVN(IT,NP)=FVN(NU,ITT)
      FVN(IT,ITT)=FVN(NU,NP)
      CONTINUE
      CCC=0.0
      DO 100 N=1,41
      CCC=CCC+(GAMMA(N)*GAMMA(N))
      CCD=0.0
      DO 110 NUP=1,41
      DO 110 N=1,41
      CCD=CCD-2.0*FVN(NUP,N)*(GAMMA(NUP)*GAMMA(N))
      CONTINUE
      CDI=PI*AR/4.*(CCC+CCD)
      CDIT=1./(PI*AR)
      RETURN
      END
      SUBROUTINE CORAGNE
      DIMENSION GAM(1100),XC4(1100),YQ(1100),CCR(40),
      $ FW(2),FV(2),XXCC(40),CCC(400),CRR(400),
      $ YB(100),CRI(102),NMA(4),XCC4(400),CHD(100),
      $ XC4(100),YY(2),PPI(100),ZZH(100),Z(1100),
      $ PHII(1000),SA(100),SSA(1100),ALUP(400),ALLP(100),
      $ ALPPO(1100),ALD(40),YC(102),YQJ(100),BOTL(4),NSUMSV(4),
      $ MSUMSV(4),NMSUMS(4),NUMBER(4)
      COMMON /ALL/ POT, ROTSV(4), M, BETA, PTEST, QTEST,
      $ STA(4), T9LSCH(100), YYCP(4),
      $ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
      $ PHI(100), ZH(100), CP(400), STLOIND(4)

```

```

CDICLS43
CDICLS44
CDICLS45
CDICLS46
CDICLS47
CDICLS48
CDICLS49
CDICLS50
CDICLS51
CDICLS52
CDICLS53
CDICLS54
CDICLS55
CDICLS56
CDICLS57
CDICLS58
CDICLS59
CDICLS60
CDICLS61
CDICLS62
CDICLS63
CDICLS64
CDICLS65
CDICLS66
CDICLS67
CDICLS68
CDICLS69
CDICLS70
CDICLS71
CDICLS72
CDICLS73
CDICLS74
CDICLS75
CDICLS76
CDICLS77
CDICLS78
CDRAGNF2
CDRAGNF3
CDRAGNF4
CDRAGNF5
CDRAGNF6
CDRAGNF7
CDRAGNF8
CDRAGNF9
ALL 2
ALL 3
ALL 4
ALL 5
ALL 6

```

```

COMMON /ONETHRE/ TWIST(4), CREF, SREF, CAVE, CLDES, STRUE, AR,
$   ARTRUE, RTCDHT(4), CONFIG(2), NSSWSV(4),
$   MSV(4), KBOT, PLAN, IPLAN, MACH,
$   SSWA(100), XL(4), XT(4), CLWB, CMCL, CLA(4), BLAIR(100),
$   CLAMAR(4), CLWIN(4), CLWNG(4), XLOCTN,
$   YINNER(4), YOUTER(4)
C
INTEGER CONFIG
C
COMMON /TOTHREE/ CIR(400,2)
C
COMMON /INSUB23/ APSI,APHI,XX,YYY,ZZ,SNN,TOLCSO
COMMON /THREFOR/ CCAV(2,100), CLT, CLNT, NSSW, ALPD
C
COMMON/CCRDD/ TSPAN(4), TSPAN4, KBIT, CTILDA, XTILDA, DISTALE
C
C
C
DATA NUMBER/5HFIRST,6HSECOND,5HTHIRD,6HFORTH/
C
WRITE (6,250)
APSI=TOLCSO=TBLS=0.
PI=4.*ATAN(1.)
FPI=4.*PI
DO 5 IT=1,IPLAN
  BOTL(IT) = ABS(TSPAN(IT))
5  CONTINUE
  SNN = BOTL(KBOT) / ( 2. * NSSWSV(KBOT))
  DELTYB=2.*SNN
  NSMS = 0
  MSMS = 0
  NMSMS = 0
  NMAX = 0
  DO 7 IT=1,IPLAN
    NMA(IT) = BOTL(IT) / DELTYB
    NSMS = NSMS + NSSWSV(IT)
    NSUMSV(IT) = NSMS
    MSMS = MSMS + MSV(IT)
    MSUMSV(IT) = MSMS
    NMSMS = NMSMS + NYA(IT)
    NMSUMS(IT) = NMSMS
7  CONTINUE
    NMAX = NMSUMS(IPLAN)
    DO 10 I=1,M
      CRR(I)=CIR(I,1)+CIR(I,2)*(CLDES-CLT)/CLNT
10  CONTINUE
      SC#MIN=20.
      DO 20 I=1,NSSW
        SC#MIN=AMIN1(SC#MIN,TBLSW(I))
20  NSC#MIN=SC#MIN

```

ONETHRE2
 ONETHRE3
 ONETHRE4
 ONETHRE5
 ONETHRE6
 ONETHRE7
 ONETHRE8
 ONETHRE9
 ONETHR10
 TOTHRE 2
 TOTHRE 3
 CORAGN13
 THREFOR2
 THREFOR3
 CCRDD 2
 CCRDD 3
 CORAGN16
 CORAGN17
 CORAGN18
 CORAGN19
 CORAGN20
 CORAGN21
 CORAGN22
 CORAGN23
 CORAGN24
 CORAGN25
 CORAGN26
 CORAGN27
 CORAGN28
 CORAGN29
 CORAGN30
 CORAGN31
 CORAGN32
 CORAGN33
 CORAGN34
 CORAGN35
 CORAGN36
 CORAGN37
 CORAGN38
 CORAGN39
 CORAGN40
 CORAGN41
 CORAGN42
 CORAGN43
 CORAGN44
 CORAGN45
 CORAGN46
 CORAGN47
 CORAGN48

```

MM=NSCWMIN*MAX
DELTXX=1./SCWMIN
DO 100 LA=1,NSSV
  CHD(LA)=CCAV(2,LA)*CAVE/BETA
  DELTXX=1./TBLSCW(LA)
  XC=-.75*DELTXX
  ITBL=TBLSCW(LA)
  DO 30 LB=1,ITBL
    XC=XC+DELTXX
    XXCC(LB)=XC
    LC=L9+TBL
  ALO(LB)=ALP(LC)
  XLE=PN(LC)+CHD(LA)*(1.-.75/TBLSCW(LA))
  XCC=-.75*DELTXX
  KCODE=LB=0
  DO 90 K=1,NSCWMIN
    J=K+(LA-1)*NSCWMIN
    XCC(J)=XCC+CHD(LA)+XLE
    CALL FILUP (XCC,ALOP(J),+1,ITBL,XXCC,ALO)
    AXMN=K*DELTXX
    CAT=0.
    IF (KCODE.EQ.2) CAT=CCR(LB)-CUT
    KCODE=0
    LB=LB+1
    LC=LB+TBL
    CCR(LB)=CCR(LC)
    AXITBL=LB*DELTXX
    IF (AXMN-AXITBL) 50,60,70
    CUT=CCR(LB)*(AXMN-(LB-1)*DELTXX)/DELTXX
    KCODE=2
    GO TO 80
  KCODE=1
  CUT=CCR(LB)
  CAT=CAT+CUT
  IF (KCODE.GE.1) GO TO 90
  IF (LB.LT.ITBL) GO TO 40
  CCC(J)=CAT
  TBL=TRLS+TBLSCW(LA)
  CONTINUE
  II=1
  DO 150 I=1,IPLAN
    IJZ=NSWSV(I)
    IJZ=IJZ+1
    IF (STLCIND(I).EQ.1.) IJX = IJZ
    IC = NSU*SV(I)
    ID=IC+1
    IZ = NSU*SV(I)
    YCAT=0.

```

30

40

50

60

70

80

90

100

CDRAGN49

CDRAGN50

CDRAGN51

CDRAGN52

CDRAGN53

CDRAGN54

CDRAGN55

CDRAGN56

CDRAGN57

CDRAGN58

CDRAGN59

CDRAGN60

CDRAGN61

CDRAGN62

CDRAGN63

CDRAGN64

CDRAGN65

CDRAGN66

CDRAGN67

CDRAGN68

CDRAGN69

CDRAGN70

CDRAGN71

CDRAGN72

CDRAGN73

CDRAGN74

CDRAGN75

CDRAGN76

CDRAGN77

CDRAGN78

CDRAGN79

CDRAGN80

CDRAGN81

CDRAGN82

CDRAGN83

CDRAGN84

CDRAGN85

CDRAGN86

CDRAGN87

CDRAGN88

CDRAGN89

CDRAGN90

CDRAGN91

CDRAGN92

CDRAGN93

CDRAGN94

CDRAGN95

CDRAGN96

CDRAGN97


```

IAMH=NMA(I)
IF(I.EQ.1) GOTO 142
NST = NSUMSV(I-1)
NSTT = NMSUMS(I-1)
GOTO 143
142 NST = 0
143 NSTT = 0
143 CONTINUE
DO 140 LA=1,NSCWMIN
IF (STLOIND(I) .EQ. 1.) GO TO 144
YC(I)=-PI/2.
CRI(I)=0.
144 DO 120 J= 1,IUZ
L=J+1
IF (STLOIND(I) .EQ. 1.) L = J
LU = LA + (J-1 + NST) * NSCWMIN
ALLP(J)=ALOP(LU)
XC44(J)=XCC4(LU)
CRI(L)=CCC(LU)
IF (LA.NE.1) GO TO 120
JJ = J + NST
ZZH(J)=ZH(JJ)
SA(J)=SSWA(JJ)
PHI(J)=PHI(JJ)
YOC(J)=O(IJ)
IJ=IJ+TBLSCW(JJ)
IE=IUZ-J+1
ITL=TBLSCW(IJ)
ID=ID-ITL
IA=ID+ITL
IF (IA.GT.IC) YCAT=YCAT-S(ID)
IF (IA.GT.IC) GO TO 110
YCAT=YCAT-S(ID)-S(IA)
110 IZ=IZ-1
YB(IE)=YCAT
120 CONTINUE
DO 130 JP=1,IUZ
JZ=JP+1
IF (STLOIND(I) .EQ. 1.) JZ = JP
YC(JZ) = ASIN(YB(JP)/BOTL(I))
130 CONTINUE
YOB=-NMA(I)*2.*SNN-SNM
DO 140 K=1,IAMH
KP = LA + (K-1 + NSTT) * NSCWMIN
YOB=YOB+DELTYR
YOC = ASIN(YOB/BOTL(I))
CALL FTUP (YOB,YO(KP),+1,IUZ,YB,YOQ)
CALL FTUP (YOB,ALPPD(KP),+1,IUZ,YB,ALLP)
CALL FTUP (YOB,SSA(KP),+1,IUZ,YB,SA)
CDRAGN98
CDRAGN99
CDRAG100
CDRAG101
CDRAG102
CDRAG103
CDRAG104
CDRAG105
CDRAG106
CDRAG107
CDRAG108
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CDRAG111
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CDRAG128
CDRAG129
CDRAG130
CDRAG131
CDRAG132
CDRAG133
CDRAG134
CDRAG135
CDRAG136
CDRAG137
CDRAG138
CDRAG139
CDRAG140
CDRAG141
CDRAG142
CDRAG143
CDRAG144
CDRAG145
CDRAG146

```

```

140 CALL FTUP (YOB,XC4(KP),+1,IUZ,YB,XC44)
150 CALL FTUP (YOB,Z(KP),+1,IUZ,YB,ZZH)
CALL FTUP (YOB,PHI(KP),+1,IUZ,YB,PHI)
CALL FTUP (YOB,GAM(KP),+1,IUX,YC,CRI)
IF (YOB.GT.YB(IUZ)) GAM(KP)=CRI(IUX)
CONTINUE
CONTINUE
CORAG=CTHRUST-CSUCT=0.
CONST=16.*SNN*BOT/SREF
DO 190 LI=1,NMAX
LA=(LI-1)*NSCWMIN+1
LB=LI*NSCWMIN
CDRAGIT=CTT=0.
DO 180 NV=LA,LB
CPT=COS(ATAN(PHII(NV)))
VELIN=0.
DO 170 NN=1,MH
XX=XC4(NV)-XC4(NN)
YY(1)=YQ(NV)-YQ(NN)
YY(2)=YQ(NV)+YQ(NN)
ZZ=Z(NV)-Z(NN)
APHI=ATAN(PHII(NN))
DO 160 I=1,2
YY=YY(I)
CALL INFSUB (BOT,FV(I),FW(I),FUI)
APHI=APHI
CONTINUE
160 VELIN=((FW(1)+FW(2))-((FV(1)+FV(2))*PHII(NV))*GAM(NN)/FPI+VELIN
CONTINUE
170 CTT=CTT+GAM(NV)*(ALPD/57.29578+ALPPD(NV))*CPT/(2.*BOT)
CORAGIT=CDRAGIT+VELIN*GAM(NV)*CPT/(2.*BOT)
CTT=CTT-CDRAGIT
SWLE=ATAN(SSA(LA))
CST=CTT/COS(SWLE)
CCC(LI)=CDRAGIT
CPR(LI)=CTT
XCC4(LI)=CST
CDRAG=CDRAG+CDPAGIT*CONST
CTHRUST=CTHRUST+CTT*CONST
CSUCT=CSUCT+CST*CONST
CONTINUE
190 TBLE=II=0
LI=0
LBLE=0
DO 220 I=1,IPLAN
IA=NMMA(I)
IF(I.EQ.1) GOTO 225
NST = NSUMSV(I-1)
NSTT = NM*SUMS(I-1)
CORAG147
CORAG148
CORAG149
CORAG150
CORAG151
CORAG152
CORAG153
CORAG154
CORAG155
CORAG156
CORAG157
CORAG158
CORAG159
CORAG160
CORAG161
CORAG162
CORAG163
CORAG164
CORAG165
CORAG166
CORAG167
CORAG168
CORAG169
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CORAG186
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CORAG189
CORAG190
CORAG191
CORAG192
CORAG193
CORAG194
CORAG195

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```

225      GOTO 226
226      NST = 0
      NSTT = 0
      CONTINUE
      DO 200 J=1,IAMH
      JJ = J + NSTT
      LA = 1 + (J-1 + NSTT) * NSCWMIN
      GAM(J)=CCC(JJ)
      XC4(J)=CRR(JJ)
      Z(J)=XCC4(JJ)
      PHII(J)=YQ(LA)
      CONTINUE
      IUZ=NSSWSV(II)
      DO 210 LBLAIR=1,IUZ
      LI=LI+1
      LU=1+TBLE
      LBLR=LBLR+1
      YAR=C(LU)
      II=II+1
      TBLE=TBLE+TBLSCW(II)
      Y008=Y88/80T
      CALL FTLUP (Y88,CDRAGIT,+1,IAMH,PHII,GAM)
      CALL FTLUP (Y88,CTT,+1,IAMH,PHII,XC4)
      CALL FTLUP (Y88,CST,+1,IAMH,PHII,Z)
      LI = LBLAIR + NST
      SWALE=ATAN(SSWA(LL))*57.29578
      DO 300 IT = 1,IPLAN
      IF(IT.GT.1) GOTO 310
      MOD = 1
      GOTO 320
310      MOD = NSUMSV(IT-1) + 1
320      IF(II.EQ.MOD) WRITE(6,240) NUMBER(IT)
330      CONTINUE
      WRITE (6,260) LI,Y008,SWALE,CDRAGIT,CTT,CST
      BLAIR(LBLR)=CST
      CONTINUE
210      CONTINUE
220      COCL2=CDRAG/CLOES**2
230      WRITE (6,270) COCL2,CTHRUST,CSUCT
      C
      C
      C
240      FORMAT(/37X,20HCONTRIBUTION OF THE ,A10,
      *35HPPLANFORM TO THE CHORD OR DRAG FORCE/)
250      FORMAT (///30X,73HINDUCED DRAG, LEADING EDGE THRUST AND SUCTION C/CDRAG240
      10EFFICIENT CHARACTERISTICS/40X,53HCOMPUTED AT THE DESIRED CL FROM CDRAG241
      2A NEAR FIEL SOLUTION/75X,20HSECTION COEFFICIENTS/48X,11HL, E. SWCDRAG242
      3EEP/25X,7HSTATION,9X,5H 2Y/8,5X,5HANGLE,7X,9HCOII C/28,5X,7HCT C/2CDRAG243
      48,5X,7HCS C/28)
      CDRAG196
      CDRAG197
      CDRAG198
      CDRAG199
      CDRAG200
      CDRAG201
      CDRAG202
      CDRAG203
      CDRAG204
      CDRAG205
      CDRAG206
      CDRAG207
      CDRAG208
      CDRAG209
      CDRAG210
      CDRAG211
      CDRAG212
      CDRAG213
      CDRAG214
      CDRAG215
      CDRAG216
      CDRAG217
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      CDRAG220
      CDRAG221
      CDRAG222
      CDRAG223
      CDRAG224
      CDRAG225
      CDRAG226
      CDRAG227
      CDRAG228
      CDRAG229
      CDRAG230
      CDRAG231
      CDRAG232
      CDRAG233
      CDRAG234
      CDRAG235
      CDRAG236
      CDRAG237
      CDRAG238
      CDRAG239
      CDRAG240
      CDRAG241
      CDRAG242
      CDRAG243
      CDRAG244

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COMMON /INSUB23/ APSI,APHI,XX,YYY,ZZ,SNN,TOLCSQ
DIMENSION CLLL(26),CDDI(26),CMMN(26),ALPH(26),CDCLZ(26),CDCLF(26),
1 COCLV(26)
REAL KVSE(4)
DIMENSION CENTR(4), TIPSUM(4)
DATA XTLEG / 60*0.0/
DATA ZLEGSV, CIRSUM, YLEGSV
$ / 200*0.0, 200*0.0, 200*0.0, 200*0.0 /
J = 0
DO 1 ITT = 1,IPLAN
  CENTR(ITT) = 0.0
  KVSE(ITT) = 0.0
  TIPSUM(ITT) = 0.0
  IF(XL(ITT).EQ. XT(ITT)) J = J + 1
1 CONTINUE
C
IF(J.EQ. IPLAN)GO TO 540
C
BLAMAR = 1.0/BETA
NSSW = 0
DO 2 ITT = 1,IPLAN
  XT(ITT) = XT(ITT) * BLAMAR
  XL(ITT) = XL(ITT) * BLAMAR
  NSSW = NSSW + NSSWSV(ITT)
2 CONTINUE
C
THE TOLERANCE SET AT THIS POINT IN THE PROGRAM MAY NEED TO BE
CHANGED FOR COMPUTERS OTHER THAN THE CDC 6000 SERIES
C
TOLC=.0100*BOT
TOLCSQ=TOLC*TOLC
C
TIPSI = 0.0
C
TIPSU=PITCH*0.
C
GEOMETRY FOR TIP TRAILING LEGS
C
IM = 0
IMM = 0
NSSL1 = 0
NSSL2 = 0
EPS = 1.E-6
CTSW = 0.0
CMZ = 0.0
CCIRS = 0.0
C

```

TIPSUC12
 TIPSUC13
 TIPSUC14
 TIPSUC15
 TIPSUC16
 TIPSUC17
 TIPSUC18
 TIPSUC19
 TIPSUC20
 TIPSUC21
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 TIPSUC54
 TIPSUC55
 TIPSUC56
 TIPSUC57
 TIPSUC58
 TIPSUC59
 TIPSUC60

```

C      DO 535 ITT = 1,IPLAN
        NSCW = MSV(ITT)/NSSWSV(ITT)
        NSSW2 = NSSW2 + NSSWSV(ITT)
        IF(ITT .EQ. IPLAN .AND. XL(ITT) .EQ. XT(ITT)) GO TO 475
        I=IM+1
        IMP = 0
        J=IM+2
        IUU = 2
        APHI=ATAN(PHI(IM+1))
        SA = SIN(APHI)
        CA=COS(APHI)
        TLX1=PN(I)-S(I)*TAN(PSI(I))*CA
        TLX2=PN(J)-S(J)*TAN(PSI(J))*CA
        CLFTLG=TLX1-TLX2
        XTLEG(1)=TLX1/2.+TLX2/2.
        YLEG=O(I)-S(I)*CA
        YLEGSV(1,ITT) = YLEG
        ZLEG=ZH(IM+1)-S(I)*SA
        ZLEGSV(1,ITT) = ZLEG
        IF (XL(ITT).EQ.XT(ITT)) GO TO 100
        DO 30 NV=2,NSCW
            NVT=Nv-1
            XTLFG(NV)=XTLEG(NVT)-CLFTLG
            NCTL=0
            NA=1
            NB=NSCW
            DO 70 NV=NA,NB
                C
                C
                C
                THE RATIO OF W/U IS INITIALIZED TO -1 BECAUSE IN THE TERM
                -U*ALPHA/U,USED IN THIS SUMMATION, ALPHA IS SET TO 1 RADIAN
                SO THAT THE RESULTING TIP SUCTION CAN BE USED DIRECTLY TO FIND
                KV SIDE EDGE
                WVOU(NV)=-1.
                IZ=1
                NNN=TRLSCW(IZ)
                DO 60 NN=1,M
                    AP-I=ATAN(PHI(IZ))
                    APSI=PSI(NN)
                    XX=XTLEG(NV)-PN(NN)
                    YY(1)=YLEG-Q(NN)
                    YY(2)=YLEG+Q(NN)
                    ZZ=ZLEG-ZH(IZ)
                    SNI=S(NN)
                    DO 50 I=1,2
                        YYY=YY(I)

```

```

50 CALL INFSUB (BOT,FV(I),FW(I),FUI)
   API = -API
   APSI = -APSI
   CONTINUE
   WVDU(NV)=WVDU(NV)+(FW(1)+FW(2))*CIR(NN,2)/12.5663704
   IF (NN.LT.NN.OR.NN.EQ.M) GO TO 60
   IZ=IZ+1
   NNN=NNN+1BLSCW(IZ)
   CONTINUE
70 CONTINUE
   NCTL=NCTL+1
   IF (NCTL-2) 80,100,150

C      GEOMETRY FOR SPANWISE BOUND VORTICES
C
80 NA=NSCW+1
   NB=2*NSCW
   JA=IMM+1
   YLEG=O(JA)
   ZLEG=ZH(IM+1)
   DO 90 J=1,NSCW
     JK=IMM+J
     NV=J+NSCW
     XTLEG(NV)=PN(JK)
     GO TO 40

C      GEOMETRY ALONG RIGHT TRAILING LEGS
C
100 NA=2*NSCW+1
   NB=3*NSCW
   CCIR3=0.
   JK=IMM+1
   API=ATAN(PHI(IM+1))
   SA=SIN(API)
   CA=COS(API)
   YLEG=Q(JK)+S(JK)*CA
   YLEGSV(IU,ITT) = YLEG
   ZLEG=ZH(IM+1)+S(JK)*SA
   ZLEGSV(IU,ITT) = ZLEG
   IF (XL(ITT).EQ.XT(ITT)) GO TO 150
   TLX1=PN(JK)+S(JK)*TAN(PHI(JK))*CA
   JK=JK+1
   TLX2=PN(JK)+S(JK)*TAN(PHI(JK))*CA
   CRTLG=TLX1-TLX2
   XTLEG(NA)=TLX1/2.+TLX2/2.
   NAA=NA+1
   IF(ITT.EQ.1) GO TO 130
   LIM = ITT - 1
   DO 125 ITLEG = 1,LIM

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TIPSU110
TIPSU111
TIPSU112
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TIPSU202
TIPSU203
TIPSU204
TIPSU205
TIPSU206
TIPSU207

L = NSSWSV(ITLEG) + 1
DO 120 I = 2,L
  IQ = I - 1
  IF((ABS(YLEGSV(I,ITLEG)-YLEG).LT.TOLC)
    .AND.
    (ABS(ZLEGSV(I,ITLEG)-ZLEG).LT.TOLC))
    $ CCIRS = CCIRS + CIRSUM(IQ,ITLEG)
    $ IF(CCIRS .NE. 0)GO TO 125
  120 CONTINUE
  125 CONTINUE
C
  DO 140 NV=NA,NB
    NVT=NV-1
    XTLEG(NV)=XTLEG(NVT)-CRITLG
    GO TO 40
  C
  C
  150 CONTINUE
  IF (CCIRS.NE.0.) GO TO 160
  GO TO 270
  160 IJ=2*NSCW+1
    XLT=XTLEG(IJ)+CLFTLG/2.
    XRT=XTLEG(IJ)+CRTTLG/2.
    XLL=XLT+CLFTLG/4.
    XRL=XRT+CRTTLG/4.
    IF (XLL.GE.XL(ITT).AND.XLT.LE.XT(ITT)) GO TO 170
    IF (XLL.LE.XL(ITT).AND.XLT.GE.XT(ITT)) GO TO 190
    IF (XLL.GT.XL(ITT).AND.XLT.GE.XL(ITT)) GO TO 200
    IF (XLL.LE.XT(ITT)) GO TO 200
    IF (XLL.GT.XL(ITT).AND.XLT.LT.XL(ITT)) GO TO 180
    CON4=(XT(ITT)-XLL)/(XLT-XLL)
    GO TO 210
    CON4=(XL(ITT)-XT(ITT))/(XLL-XLT)
    GO TO 210
    CON4=(XL(ITT)-XLT)/(XLL-XLT)
    GO TO 210
    CON4=1.
    GO TO 210
    CON4=0.0
    200 CONTINUE
    210 IF (XRL.GE.XL(ITT).AND.XRT.LE.XT(ITT)) GO TO 220
    IF (XPL.LE.XL(ITT).AND.XRT.GE.XT(ITT)) GO TO 240
    IF (XPL.GT.XL(ITT).AND.XPT.GE.XL(ITT)) GO TO 250
    IF (XPL.LE.XT(ITT)) GO TO 250
    IF (XPL.GT.XL(ITT).AND.XRT.LT.XL(ITT)) GO TO 230
    CON45=(XT(ITT)-XRL)/(XRT-XRL)
    GO TO 260
    CON45=(XL(ITT)-XT(ITT))/(XRL-XRT)
    GO TO 260
    220
    260

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230  CON5=(XL(IIT)-XRT)/(XRL-XRT)
240  GO TO 260
250  CON5=1.
260  GO TO 260
270  CON5=0.0
280  CONTINUE
290  TIPSU = CCIRS*0.25*(CON4*WVOU(1)*CLFTLG-CON5*WVOU(IJ)*CRTILG)
300  , *2./SREF*BETA
310  IF(TIPSI .GT. 0 .AND. SSWHA(IM+1) .LT. 0) GO TO 270
320  IF(TIPSI .GT. 0) TIPSUM(IIT) = TIPSUM(IIT) + TIPSU
330  TIPSU = TIPSU + TIPSU
340  PITCH=PITCH+CCIRS*0.25*(-CON4*WVOU(1)*CLFTLG*XTLEG(1)+CON5*WVOU(IJ)*CRTILG)
350  1)*CRTILG*XTLEG(IJ))*2./SREF*CREP)*BETA**2
360  CIRCUS=CCIRS
370  DD 460 NPOS=1, NSCW
380  JK=IM+NPOS
390  JN=2*NSCW+NPOS
400  NPIS=NSCW+NPOS
410  CIRCUS=CIRCUS+CIR(JK,2)
420  IF (XL(IIT).EQ.XT(IIT)) GO TO 460
430  XLLEG=XTLEG(NPOS)
440  XRLEG=XTLEG(JN)
450  XLL=XTLEG(NPOS)+CLFTLG/2.
460  XLT=XTLEG(NPOS)-CLFTLG/2.
470  XRL=XTLEG(JN)+CRTILG/2.
480  XRT=XTLEG(JN)-CRTILG/2.
490  IF (XLL.GE.XL(IIT).AND.XLT.LE.XT(IIT)) GO TO 280
500  IF (XLL.LE.XL(IIT).AND.XLT.GE.XT(IIT)) GO TO 300
510  IF (XLL.GT.XL(IIT).AND.XLT.GE.XL(IIT)) GO TO 310
520  IF (XLL.LE.XT(IIT)) GO TO 310
530  IF (XLL.GT.XL(IIT).AND.XLT.LT.XL(IIT)) GO TO 290
540  CON1=(XT(IIT)-XLL)/(XLT-XLL)
550  XLLEG=XT(IIT)+CON1*CLFTLG/2.
560  GO TO 320
570  CON1=(XL(IIT)-XT(IIT))/(XLL-XLT)
580  XLLEG=(XL(IIT)+XT(IIT))/2.
590  GO TO 320
600  CON1=(XL(IIT)-XLT)/(XLL-XLT)
610  XLLEG=XLT+CON1*CLFTLG/2.
620  GO TO 320
630  CON1=1.
640  GO TO 320
650  CON1=0.0
660  CONTINUE
670  IF (VPOS.EQ.NSCW.AND.CON1.EQ.1.) GO TO 360
680  IF (XKL.GE.XL(IIT).AND.XRT.LE.XT(IIT)) GO TO 330
690  IF (XRL.LE.XL(IIT).AND.XRT.GE.XT(IIT)) GO TO 350
700  IF (XPL.GT.XL(IIT).AND.XRT.GE.XL(IIT)) GO TO 370
710  IF (XRL.LE.XT(IIT)) GO TO 370

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IF (XRL.GT.XL(ITT).AND.XRT.LT.XL(ITT)) GO TO 340
CON2=(XT(ITT)-XRL)/(XRT-XRL)
XRLEG=XT(ITT)+CON2*CRTLGLG/2.
GO TO 380
330 CON2=(XL(ITT)-XT(ITT))/(XRL-XRT)
    XRLEG=(XL(ITT)+XT(ITT))/2.
    GO TO 380
340 CON2=(XL(ITT)-XRT)/(XRL-XRT)
    XRLEG=XRT+CON2*CRTLGLG/2.
    GO TO 380
350 CON2=1.
    GO TO 380
360 CON1=.75
    CON2=.75
    GO TO 380
370 CON2=0.0
380 IF (XRL.GT.XLL) GO TO 390
    XSIGN=-1.0
    XBLL=XLL
    XBLL=XRL
    GO TO 400
390 XBLL=XRL
    XBLL=XLL
    XSIGN=1.
    BVDLG=ABS(XBLL-XBLL)
    IF (XBLL.GE.XL(ITT)) GO TO 440
    IF (XBLL.LE.XT(ITT)) GO TO 440
    IF (XBLL.GE.XL(ITT).AND.XBLL.LE.XT(ITT)) GO TO 430
    IF (XBLL.LE.XL(ITT).AND.XBLL.GE.XT(ITT)) GO TO 420
    IF (XBLL.GT.XL(ITT).AND.XBLL.GE.XT(ITT)) GO TO 410
    CON3=(XT(ITT)-XBLL)/(XBLL-XBLL)
    XTLEG(NPIS)=XT(ITT)+CON3*BVDLG/2.
    CON3=CON3*XSIGN
    GO TO 450
410 CON3=(XL(ITT)-XBLL)/(XBLL-XBLL)
    XTLEG(NPIS)=XBLL+CON3*BVDLG/2.
    CON3=CON3*XSIGN
    GO TO 450
420 CON3=1.*XSIGN
    GO TO 450
430 CON3=(XL(ITT)-XT(ITT))/(XBLL-XBLL)
    XTLEG(NPIS)=(XL(ITT)+XT(ITT))/2.
    CON3=CON3*XSIGN
    GO TO 450
440 CON3=0.0
450 TIPS1=(CIRCUS*(WVOU(NPOS)*CLFTLG*CON1-CON2*WVOU(JN)*CRTTLG)
    , +CIR(JK,2)*(WVOU(NPIS)*CON3*BVDLG))*2./SREF*BETA
    IF (TIPS1.GT.0.AND.SS44A(IM+1).LT.0) GO TO 460
    IF(TIPS1.GT.0)TIPSUM(ITT)=TIPSUM(ITT)+TIPS1
TIPSU257
TIPSU258
TIPSU259
TIPSU260
TIPSU261
TIPSU262
TIPSU263
TIPSU264
TIPSU265
TIPSU266
TIPSU267
TIPSU268
TIPSU269
TIPSU270
TIPSU271
TIPSU272
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TIPSU300
TIPSU301
TIPSU302
TIPSU303
TIPSU304
TIPSU305

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TIPSU = TIPSU + TIPSU
PITCH=PITCH+(CIRCUS*(-WVOU(NPOS)*CLFTLG*CON1+XLLEG+WVOU(JN)*CON2*CTIPSU307
IRTTLG*XRLEG)-CIR(JK,2)*(WVOU(NPIS)*CON3*BVDLG*XTLEG(NPIS)))*2./(SR(TIPSU308
2EF*CREF)*BETA**2
460 CONTINUE
IM=IM+1
IMM=IMM+TBLSCW(IM)
IMP = IMP + 1
CIRSUM(IMP,ITT) = CIRCUS - CCIRS
IUU = IMP + 2
465 IF(IM.EQ.NSSW2)GO TO 475
IF(XL(ITT).EQ.XT(ITT))GO TO 100
C
IF(TBLSCW(IM+1).NE.TBLSCW(IM))GO TO 471
NCTL = 1
CLFTLG = CRTTLG
DO 470 NV = 1,NSCW
NY = NV + 2*NSCW
XTLEG(NV) = XTLEG(NY)
WVOU(NV) = WVOU(NY)
470 CONTINUE
GO TO 80
471 NSCW = TRLSCW(IM+1)
I = IMM + 1
J = I + 1
APHI = ATAN(PHI(IM+1))
GO TO 25
C
475 CONTINUE
KVSE(ITT) = 2.0 * ABS(TIPSU - CTSW)
IF(KVSE(ITT).LT.EPS)GO TO 510
CENTR(ITT) = (PITCH - CMW) * CREF/ABS(TIPSU - CTSW)
C
510 CMW = PITCH
CTSW = TIPSU
NSSW1 = NSSW2(ITT)
C
535 CONTINUE
C
C
C
540 IF(CLDOS.EQ.100.) GO TO 541
CALL WRTANS(KVSE, CENTR, TIPSUM)
GO TO 575
541 RAD=4.*ATAN(1.)/180.
WRITE(6,690)
DO 580 IK=1,IPLAN
WRITE(6,700) IK

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```

XL(IK) = XL(IK) * BETA
XT(IK) = XT(IK) * BETA
WRITE(6,710) XL(IK), XT(IK),
$      KVSE(IK), CENTR(IK), BOTSV(IK)
580 CONTINUE
WRITE(6,560)
DO 600 IA=1,26
READ(30) WVOU(IA), XTLEG(IA), CIRSUM(IA)
600 CONTINUE
DO 605 IA=1,26
READ(30) YLEGSV(IA), ZLEGSV(IA), TBLSCW(IA)
605 CONTINUE
DO 610 IA=1,26
READ(30) Q(IA), PN(IA), PV(IA)
610 CONTINUE
DO 620 IA=1,26
READ(30) ALP(IA), S(IA), PSI(IA)
620 CONTINUE
REWIND 30
CLOZ = 0.0
CDMINZ = XTLEG(1)
CLOF = 0.0
CDMINF = ZLEGSV(1)
CDMINV = 100.0
CLOV = 0.0
DO 545 IALPH = 1,26
ALPH(IALPH) = (IALPH-6)*2.
ALPHA = ALPH(IALPH)*RAD
SA=SIN(ALPHA)
SA2=SA*ABS(SA)
CA=COS(ALPHA)
CLVSE=CDVSE=CMVSE=0.
DO 550 I=1,IPLAN
CLVSE = KVSE(I) * SA2 * CA + CLVSE
CDVSE = KVSE(I) * SA2 * SA + CDVSE
CMVSE = KVSE(I) * SA2 * CENTR(I)/CREF + CMVSE
550 CONTINUE
WRITE(6,570) ALPH(IALPH), WVOU(IALPH), XTLEG(IALPH), CIRSUM(IALPH),
1      YLEGSV(IALPH), ZLEGSV(IALPH), TBLSCW(IALPH),
2      Q(IALPH), PN(IALPH), PV(IALPH), CLVSE, CDVSE, CMVSE
IF (XTLEG(IALPH).GE.CDMINZ) GO TO 900
CDMINZ = XTLEG(IALPH)
CLOZ = WVOU(IALPH)
900 CONTINUE
IF (ZLEGSV(IALPH).GE.CDMINF) GO TO 910
CDMINF = ZLEGSV(IALPH)
CLOF = YLEGSV(IALPH)
910 CONTINUE
WRITE(30) CLVSE, CDVSE, CMVSE

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TIPSU355
TIPSU356
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TIPSU398
TIPSU399
TIPSU400
TIPSU401
TIPSU402
TIPSU403

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545 CONTINUE
END FILE 30
REWIND 30
WRITE(6,800)
DO 810 IALPH=1,26
READ (30) CLVSE,CDVSE,CMVSE
CLL=Q(IALPH)+CLVSE
CDD=PN(IALPH)+CDVSE
CMM=PV(IALPH)+CMVSE
CLLA=Q(IALPH)+ALP(IALPH)
CDDA=PN(IALPH)+S(IALPH)
CMAA=PV(IALPH)+PSI(IALPH)
CLLL(IALPH) = CLL+ALP(IALPH)
CDDD(IALPH) = CDD+S(IALPH)
CMM(IALPH)= CMM+PSI(IALPH)
WRITE(6,570) ALPH(IALPH),ALP(IALPH),S(IALPH),PSI(IALPH),CLL,CDD,
1CMM,CLLA,CDDA,CMAA,CLLL(IALPH), CDDD(IALPH), CMM(IALPH)
IF (CDDD(IALPH).GE.COMINV) GO TO 920
COMINV = CDDD(IALPH)
CLOV = CLL(IALPH)
920 CONTINUE
810 CONTINUE
PIAR = 1./((3.14159265*AR)
WRITE (6,831) PIAR,CLOZ,COMINZ,CLOF,COMINF,CLOV,COMINV
DO 820 IALPH=1,26
IF(WVOU(IALPH).EQ.CLOZ) CCLZ(IALPH) = 100.
IF(YLEGSV(IALPH).EQ.CLOF) CDCLF(IALPH) = 100.
IF(CLL(IALPH).EQ.CLOV) CDCLV(IALPH) = 100.
IF(WVOU(IALPH).NE.CLOZ)CCLZ(IALPH) = (XTLEG(IALPH)-COMINZ)/((WVOU
1(IALPH) - CLOZ)*2)
IF(YLEGSV(IALPH).NE.CLOF) CDCLF(IALPH)= (ZLEGSV(IALPH)-COMINF)/
1((YLEGSV(IALPH)- CLOF)*2)
IF(CLL(IALPH).NE.CLOV) CDCLV(IALPH)= (CDDD(IALPH)- COMINV)/
1((CLLL(IALPH)- CLOV)*2)
WRITE(6,830) ALPH(IALPH), WVOU(IALPH),CDCLZ(IALPH),YLEGSV(IALPH),
1CDCLF(IALPH), CLL(IALPH), CDCLV(IALPH)
920 CONTINUE
CALL PSEUDO
PLT CL VS ALPHA FOR ZERO SUCTION, FULL SUCTION, AND VORTEX LIFT
CALL INFOPLT( 0,26, ALPH, 1,YLEGSV,1,-10., 40.0,-.4,1.6,1.0,-5,
15HALPHA,-2, 2HCL,12,10.,10.,0.75,1.50)
CALL INFOPLT( 0,26, ALPH, 1,YLEGSV,1,-10., 40.0,-.4,1.6,1.0,-5,
15HALPHA,-2, 2HCL, 0,10.,10.,0.75,1.50)
CALL INFOPLT( 0,26, ALPH, 1, WVOU, 1,-10., 40.0,-.4,1.6,1.0,-5,
15HALPHA,-2, 2HCL, 0,10.,10.,0.75,1.50)
CALL INFOPLT( 0,26, ALPH, 1, WVOU, 1,-10., 40.0,-.4,1.6,1.0,-5,
15HALPHA,-2, 2HCL,11,10.,10.,0.75,1.50)

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CALL INFOPLT( 0.26, ALPHA, 1, CLLL, 1, -10., 40.0, -.4, 1.6, 1.0, -5,
15HALPHA, -2, 2HCL, 0.10, 10., 0.75, 1.50)
CALL NOTATE ( 7.0, 2.7, 0.15, 12HZERO SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.77, 11, 2)
CALL NOTATE ( 7.0, 2.4, 0.15, 12HFULL SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.47, 12, 2)
CALL NOTATE ( 7.0, 2.1, 0.15, 16HPOTENTIAL+VORTEX, 0.0, 16)
CALL PNTPLT ( 6.7, 2.17, 13, 2)
CALL INFOPLT( 1.26, ALPHA, 1, CLLL, 1, -10., 40.0, -.4, 1.6, 1.0, -5,
15HALPHA, -2, 2HCL, 13, 10., 10., 0.75, 1.50)

PLOT CL VS CD FOR ZERO SUCTION, FULL SUCTION, AND VORTEX LIFT
TIPSU453
TIPSU454
TIPSU455
TIPSU456
TIPSU457
TIPSU458
TIPSU459
TIPSU460
TIPSU461
TIPSU462
TIPSU463
TIPSU464
TIPSU465
TIPSU466
TIPSU467
TIPSU468
TIPSU469
TIPSU470
TIPSU471
TIPSU472
TIPSU473
TIPSU474
TIPSU475
TIPSU476
TIPSU477
TIPSU478
TIPSU479
TIPSU480
TIPSU481
TIPSU482
TIPSU483
TIPSU484
TIPSU485
TIPSU486
TIPSU487
TIPSU488
TIPSU489
TIPSU490
TIPSU491
TIPSU492
TIPSU493
TIPSU494
TIPSU495
TIPSU496
TIPSU497
TIPSU498
TIPSU499
TIPSU500
TIPSU501

CALL INFOPLT( 0.26, WVOU, 1, XTLEG, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 0.10, 10., 0.75, 1.5)
CALL INFOPLT( 0.26, YLEGSV, 1, ZLEGSV, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 0.10, 10., 0.75, 1.5)
CALL INFOPLT( 0.26, CLLL, 1, CDD, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 0.10, 10., 0.75, 1.5)
CALL INFOPLT( 0.26, WVOU, 1, XTLEG, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 11, 10., 10., 0.75, 1.5)
CALL INFOPLT( 0.26, YLEGSV, 1, ZLEGSV, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 12, 10., 10., 0.75, 1.5)
CALL NOTATE ( 7.0, 2.7, 0.15, 12HZERO SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.77, 11, 2)
CALL NOTATE ( 7.0, 2.4, 0.15, 12HFULL SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.47, 12, 2)
CALL NOTATE ( 7.0, 2.1, 0.15, 16HPOTENTIAL+VORTEX, 0.0, 16)
CALL PNTPLT ( 6.7, 2.17, 13, 2)
CALL INFOPLT( 1.26, CLLL, 1, CDD, 1, -.4, 1.6, 0.4, 1.0, -2, 2HCL, -2,
2HCD, 13, 10., 10., 0.75, 1.5)

PLOT CL VS CM FOR ZERO SUCTION, FULL SUCTION, AND VORTEX LIFT
TIPSU453
TIPSU454
TIPSU455
TIPSU456
TIPSU457
TIPSU458
TIPSU459
TIPSU460
TIPSU461
TIPSU462
TIPSU463
TIPSU464
TIPSU465
TIPSU466
TIPSU467
TIPSU468
TIPSU469
TIPSU470
TIPSU471
TIPSU472
TIPSU473
TIPSU474
TIPSU475
TIPSU476
TIPSU477
TIPSU478
TIPSU479
TIPSU480
TIPSU481
TIPSU482
TIPSU483
TIPSU484
TIPSU485
TIPSU486
TIPSU487
TIPSU488
TIPSU489
TIPSU490
TIPSU491
TIPSU492
TIPSU493
TIPSU494
TIPSU495
TIPSU496
TIPSU497
TIPSU498
TIPSU499
TIPSU500
TIPSU501

CALL INFOPLT( 0.26, YLEGSV, 1, TBLSCW, 1, -.4, 1.6, -0.10, 0.10, 1.0, -2,
2HCL, -2, 2HCD, 12, 10., 10., 0.75, 1.50)
CALL INFOPLT( 0.26, YLEGSV, 1, TBLSCW, 1, -.4, 1.6, -0.10, 0.10, 1.0, -2,
2HCL, -2, 2HCD, 0.10, 10., 0.75, 1.50)
CALL INFOPLT( 0.26, WVOU, 1, CIRS, 1, -.4, 1.6, -0.10, 0.10, 1.0, -2,
2HCL, -2, 2HCD, 11, 10., 10., 0.75, 1.50)
CALL INFOPLT( 0.26, WVOU, 1, CIRS, 1, -.4, 1.6, -0.10, 0.10, 1.0, -2,
2HCL, -2, 2HCD, 0.10, 10., 0.75, 1.50)
CALL INFOPLT( 0.26, CLLL, 1, CMH, 1, -.4, 1.6, -0.10, 0.10, 1.0, -2,
2HCL, -2, 2HCD, 0.10, 10., 0.75, 1.50)
CALL NOTATE ( 7.0, 2.7, 0.15, 12HZERO SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.77, 11, 2)
CALL NOTATE ( 7.0, 2.4, 0.15, 12HFULL SUCTION, 0.0, 12)
CALL PNTPLT ( 6.7, 2.47, 12, 2)
CALL NOTATE ( 7.0, 2.1, 0.15, 16HPOTENTIAL+VORTEX, 0.0, 16)

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C 710 FORMAT(23X, F10.5, 10H (LEADING), 2X, F10.5, 11H (TRAILING),
$ 4X, F10.5, 2X, F10.5, 2X, F10.5)

C 800 FORMAT(1H1///4X118HS E P A R A T I N D
1 F L O W C O N T I A L C O N T I P S U 551
2 D/45X80HP L U S P O T E N T I A L C O N T I P S U 552
3 R I B U T I O N S/15X109HA U G M E N T E D L E A D T I P S U 553
4 I N G E D G E + S I D E E D G E L E A D I N G E D G E + A U G M E N T E D L E + S E T I P S U 554
5 + A U G M E N T E D / T I P S U 555
6 T I P S U 556
7 U G C M A U G C L C D C D C M C M C L A U G C D A T I P S U 557
8 C M C L C D C D C M C M C L A U G C D A T I P S U 558
575 CONTINUE T I P S U 559
END T I P S U 560
SUBROUTINE WRTANS(KVSE, CENTR, TIPSUM)
COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,
$ STA(4), TBLSCW(100), YYCP(4),
$ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
$ PHI(100), ZH(100), CP(400), STLOIND(4)
COMMON /ONETHRE/ TWIST(4), CREF, SREF, CAVE, CLOS, STRUE, AR,
$ ARTRUE, RTCDHT(4), CONFIG(2), NSSWSV(4),
$ MSV(4), KBOT, PLAN, IPLAN, MACH,
$ SSWWA(100), XL(4), XT(4), CLWB, CMCL, CLA(4), BLAIR(100),
$ CLAMAP(4), CLWIN(4), CLWNG(4), XLOCTN,
$ YINNER(4), YOUTER(4)
INTEGER CONFIG
COMMON /THREFOR/ CCAV(2,100), CLT, CLNT, NSSW, ALPD
REAL KP(4), KVLE(4), KVSE(4)
DIMENSION CENTR(4), CENT(4), TIPSUM(4), CENTY(4)
LCH=0
LAMAR=NSSWSV(1)
CONV=3.1415926536/180.
CINV=1./((3.1415926536*AR)
DELTA=2.*CONV
CONST=16.*BOT/SREF
ALPHA=ALPD*CONV
S22=ALPHA**2
EPS = 1.E-6
DO 10 I=1,IPLAN
CENT(ITT) = 0.0
XT(ITT) = XT(ITT) * BETA
XL(ITT) = XL(ITT) * BETA
CENTY(ITT) = 0.0
KP(ITT) = 0.0
TIPSU551
TIPSU552
TIPSU553
TIPSU554
TIPSU555
TIPSU556
TIPSU557
TIPSU558
TIPSU559
TIPSU560
TIPSU561
TIPSU562
TIPSU563
TIPSU564
TIPSU565
WRTANS 2
ALL 2
ALL 3
ALL 4
ALL 5
ALL 6
ONETHRE2
ONETHRE3
ONETHRE4
ONETHRE5
ONETHRE6
ONETHRE7
ONETHRE8
ONETHRE9
ONETHR10
THREFOR2
THREFOR3
WRTANS 6
WRTANS 7
WRTANS 8
WRTANS 9
WRTANS10
WRTANS11
WRTANS12
WRTANS13
WRTANS14
WRTANS15
WRTANS16
WRTANS17
WRTANS19
WRTANS19
WRTANS20
WRTANS21
WRTANS22

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WRTANS23
WRTANS24
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WRTANS26
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WRTANS62
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WRTANS64
WRTANS65
WRTANS66
WRTANS67
WRTANS68
WRTANS69
WRTANS70
WRTANS71

KVLE(ITT) = 0.0
10 CONTINUE
C
  KP(1) = CLWIN(1)
  IF(IPLAN.EQ.1) GO TO 20
  DO 15 ITT = 2,IPLAN
    KP(ITT) = CLWIN(ITT) - CLWIN(ITT-1)
  15 CONTINUE
  20 CONTINUE
C
  IEDGE = 1
  INDEX = 0
  DO 50 ITT = 1,IPLAN
    MSPAN = NSSWSV(ITT)
    PITCHMO = 0.0
    FORCE = 0.0
    ROLLMO = 0.0
    DO 30 ISPAN = 1,MSPAN
      INDEX = INDEX + 1
      IF(Q(IEDGE).GT.YINNER(ITT)) GO TO 25
      IF(Q(IEDGE).LT.YOUTER(ITT)) GO TO 25
      PITCHMO = PITCHMO + (PN(IEDGE) +
        (PN(IEDGE)-PN(IEDGE+1))/4.0)* BETA *
        $
        $ BLAIR(INDEX) * S(IEDGE) * CONST
      FORCE = FORCE + BLAIR(INDEX) * S(IEDGE) * CONST
      ROLLMO = ROLLMO + Q(IEDGE) * BLAIR(INDEX) *
        $
        $ S(IEDGE) * CONST
      IEDGE = IEDGE + TBLSCW(INDEX)
    25 CONTINUE
    30 IF(FORCE.EQ.0.0) GO TO 50
    KVLE(ITT) = FORCE / S22
    IF(PITCHMO.NE.0.0) CENTY(ITT) = PITCHMO / FORCE
    IF(ROLLMO.NE.0.0) CENTY(ITT) = ROLLMO / FORCE
    IF(KVLE(ITT).GT.EPS) GO TO 50
    CENTY(ITT) = 0.0
    CENTY(ITT) = 0.0
  50 CONTINUE
C
  100 WRITE (6,190)
  DO 110 IK=1,IPLAN
    CENTPM=CLAYAR(IK)*CREF
    WRITE (6,200) IK
    SPANJFK = YICP(IK) * BOT
    WRITE(6,210) KP(IK), CENTPM, SPANOFK
    CENTYSP = CENTY(IK) * BOT
    WRITE(6,220) YINNER(IK), YOUTER(IK),

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      S      KVL(1K), CENT(1K), CENTYSP
      WRITE(6,230) XL(1K), XT(1K),
      S      KVS(1K), CENTR(1K), BUTSV(1K)
      WRITE(6,290) TIPSUM(1K)
110 CONTINUE
120 CONTINUE
      DO 160 IK=1,IPLAN
      IF (LCH.EQ.1) GO TO 130
      WRITE (6,240) IK
130 WRITE (6,250)
      ALPHA=0.0
      DO 160 J=1,26
      V=SIN(ALPHA)
      C=COS(ALPHA)
      C2=C**2
      S2=V**2
      IF (LCH .EQ. 1) GO TO 140
      C      INDIVIDUAL PLANFORM CHARACTERISTICS
      C
      CLP = KP(1K) * V * C2
      CLVL = CLP + KVL(1K) * S2 * C
      CLSL = CLP + KVS(1K) * S2 * C
      CLTOT = CLVL + KVS(1K) * S2 * C
      CMP = CLAMAR(1K) * KP(1K) * V * C
      CMPL = CMP + CENT(1K) * KVL(1K) * S2/CREF
      CMPS = CMP + KVS(1K) * CENTR(1K) * S2/CREF
      CMTOT = CMPL + KVS(1K) * CENTR(1K) * S2/CREF
      GO TO 150
      C      TOTAL PLANFORM CHARACTERISTICS
      C
140 CLP = SKP * V * C2
      CLVL = CLP + SKVL * S2 * C
      CLSL = CLP + SKVS * S2 * C
      CLTOT = CLVL + SKVS * S2 * C
      CMP = V * C * SCMP
      CMPL = CMP + S2 * SCMPL/CREF
      CMPS = CMP + S2 * SCMPS/CREF
      CMTOT = CMPL + CMPS - CMP
      C
150 COI=CLTOT*TAN(ALPHA)
      COII=(CLTOT**2)*CINV
      ALPHA=ALPHA/CONV
      CNIT=CLTOT/C
      WRITE (6,260) ALPHA,CNIT,CLP,CLVL,CLSL,CLTOT,CMP,CMPL,CMPS,CMTOT,
10I,COII
      ALPHA=ALPHA*DELTA
      IF (IPLAN.EQ.1) GO TO 170
      IPLAN=1
      LCH=1
160

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WRTANS72
 WRTANS73
 WRTANS74
 WRTANS75
 WRTANS76
 WRTANS77
 WRTANS78
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 WRTAN114
 WRTAN115
 WRTAN116
 WRTAN117
 WRTAN118
 WRTAN119
 WRTAN120

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C
WRITE (6,280)
SKP = 0.0
SKVLE = 0.0
SKVSE = 0.0
SCMP = 0.0
SCMPL = 0.0
SCMPS = 0.0

K = IFIX(PLAN)
DO 165 IIT = 1,K
  SKP = SKP + KP(IIT)
  SKVLE = SKVLE + KVLE(IIT)
  SKVSE = SKVSE + KVSE(IIT)
  SCMP = SCMP + KP(IIT) * CLAMAR(IIT)
  SCMPL = SCMPL + KVLE(IIT) * CENT(IIT)
  SCMPS = SCMPS + KVSE(IIT) * CENT(IIT)
165 CONTINUE

C
GO TO 120
170 WRITE (6,270)
  IPLAN=PLAN
  RETURN
C
C
C
180 FORMAT(8F10.5)
190 FORMAT(1H1,/,/,41X,
  $ 50HKP , KV AND RESPECTIVE CENTROIDS FOR EACH PLANFORM )
C
200 FORMAT(/,/,55X,18HPLANFORM NUMBER ,I2,/,
  $ 92X,8HLOCATION,/,
  $ 77X,5HVALUE,3X,9HCHORDWISE,4X,8HSPANWISE,/,
  $ 21X,21HLIMITS OF INTEGRATION )
C
210 FORMAT(65X,7HKP ,3(F10.5,2X))
C
220 FORMAT(10X,F10.5,2X,7H(INNER),6X,F10.5,2X,7H(OUTER),
  $ 11X,7HKV LE ,3(F10.5,2X))
C
230 FORMAT(10X,F10.5,2X,9H(LEADING),4X,F10.5,2X,10H(TRAILING) ,
  $ 8X,7HKV SE ,3(F10.5,2X))
C
240 FORMAT (1H1,/,/,43X,40HPERFORMANCE CHARACTERISTICS FOR PLANFORM,IWRAN164
12)
250 FORMAT (/7X,5HALPHA,6X,2HCN,8X,3HCLP,4X,9HCLP+CLVLE,1X,9HCLP+CLVSWRAN166
1E,4X,2HCL,8X,3HCHMP,4X,9HCHMP+CHVLE,1X,9HCHMP+CHVSE,4X,2HCM,8X,2HCD,3WRAN167
2X,13HCL**2/(PI*AR)/)
260 FORMAT(3X,12F10.5)
WRAN121
WRAN122
WRAN123
WRAN124
WRAN125
WRAN126
WRAN127
WRAN128
WRAN129
WRAN130
WRAN131
WRAN132
WRAN133
WRAN134
WRAN135
WRAN136
WRAN137
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WRAN152
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WRAN155
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WRAN160
WRAN161
WRAN162
WRAN163
WRAN164
WRAN165
WRAN166
WRAN167
WRAN168
WRAN169

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270 FORMAT (///,50X,21HTHIS CASE IS FINISHED)
280 FORMAT (1H1,///,48X,33HTOTAL PERFORMANCE CHARACTERISTICS)
290 FORMAT(1H0,32X,
      $ 47HSUM OF THE POSITIVE SIDE EDGE CONTRIBUTIONS = ,F10.5)
C
      END
      SURROUTINE VORTEX
      DIMENSION GAM(1000),XC4(1000),YQ(1000),CCR(40),
      $ FW(2),FV(2),XCC(400),CCC(400),CCU(400),YR(100),
      $ BOTL(4),NUMBER(4),NMA(4),NMA(4),
      $ CRI(102),NMA(4),XCC(400),CHD(100),XC4(100),YY(2),
      $ PPHI(100),ZZH(100),Z(1000),PHI(1000),SA(100),
      $ SSA(1000),ALOP(400),ALLP(100),ALPPD(1000),ALDI(40),
      $ YC(102),YQ(100),CCR(40),CRJ(102),YCHLO(4),YCHHI(4),
      $ VELIN(26),CDRAGIT(26),CLL(26),GAD(1000),CUTE(26),
      $ CMM(26),CDRAG(26),CLIFT(26),CPITCH(26),
      $ CSS(26),CSUCT(26),CLAUG(26),CDAUG(26),CMAUG(26),CLV(26)
C
      COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,
      $ STA(4), TBLSCW(100), YYCP(4),
      $ Q(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
      $ PHI(100), ZH(100), CP(400), STLOIND(4)
C
      --- NOTES TO THE USERS ---
C
      1. BOTH TOTAL RESULTS AND THOSE FROM THE LEADING
      EDGE VORTEX SOLUTION WILL AGREE IF AND ONLY IF
      ALL PANELS ARE OF UNIFORM WIDTH, AND CAN BE
      CALCULATED FROM CLDES = 100. AND CLDES = 1.0
C
      2. IF A WING HAS MORE THAN ONE STREAMWISE TIP, IT IS RECOM-
      MENDED THAT THE WING BE INPUT AS TWO PLANFORMS TO PROVIDE
      MORE MEANINGFUL SIDE EDGE RESULTS
C
      3. STLOIND - "STREAMWISE LOAD INDICATOR" ARRAY; SET TO
      0. IF THE LOADING ALONG THE ENTIRE OUTER STREAMWISE
      EDGE OF THIS PLANFORM IS TO BE 0.0; OTHERWISE, SET TO
      1.0 IF THIS LOADING IS TO BE NON-ZERO
C
      COMMON /ONETHRE/ TWIST(4), CREF, SKEF, CAVE, CLDES, STRUE, AR,
      $ ARTRUE, RTCDHT(4), CONFIG(2), NSSWSV(4),
      $ WTV(4), KROT, PLAN, IFLAN, MACH,
      $ SSJVAL(100), YL(4), YI(4), CLWR, CMCL, CIA(4), RLAI(100),
      $ CLAMAR(4), CLAIR(4), CLAWC(4), YLCTN,
      $ YIWR(4), YIWR(4)
C
      INTERG CONFIG

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WRTAN170
WRTAN171
WRTAN172
WRTAN173
WRTAN174
WRTAN175
VORTEX 2
VORTEX 3
VORTEX 4
VORTEX 5
VORTEX 6
VORTEX 7
VORTEX 8
VORTEX 9
VORTEX10
VORTEX11
VORTEX12
VORTEX13
ALL 2
ALL 3
ALL 4
ALL 5
ALL 6
VORTEX15
NOTES 2
NOTES 3
NOTES 4
NOTES 5
NOTES 6
NOTES 7
NOTES 8
NOTES 9
NOTES 10
NOTES 11
NOTES 12
NOTES 13
NOTES 14
NOTES 15
NOTES 16
NOTES 17
NOTES 18
NOTES 19
NOTES 20
NOTES 21
NOTES 22
NOTES 23
VORTEX17
ONETHRE2
ONETHRE3
ONETHRE4
ONETHRE5
ONETHRE6
ONETHRE7
ONETHRE8
ONETHRE9

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```

20      GO TO 1,NSSW
      SCWMIN=AMIN1(SCWMIN,TBLSCW(I))
      NSCWMIN=SCWMIN
      MM=MIN(CWMIN,NMAX)
      DELTXC=1./SCWMIN
      DO 100 LA=1,NSSW
        CHD(LA)=CCAV(2,LA)*CAVE/RETA
        DELTY=1./TBLSCW(LA)
        XC=-.75*DELTX
        ITBL=TBLSCW(LA)
        DO 30 LR=1,ITBL
          XC=XC+DELTX
          XXCC(LR)=XC
          LC=LR+ITBL
          ALP(LR)=ALP(LC)
          XLF=PN(LC)+CHD(LA)*(1.-.75/TBLSCW(LA))
          XCC=-.75*DELTXC
          WCC=LC=0
          DO 90 K=1,NSCWMIN
            J=X+(LA-1)*NSCWMIN
            XCC=XCC+DELTXC
            XCC4(J)=-XCC*CHD(LA)+XLF
            CALL FTLP (XCC,ALDP(J),+1,ITBL,XXCC,ALD)
            AXN=X*DELTXC
            CAT=CAT+0.
            IF (XCODE.EQ.2) CAT=CCP(LR)-CLT
            IF (XCODE.EQ.2) CAT=CCP(LR)-CUTA
            WCC=0
            LR=LR+1
            LC=LR+ITBL
            CCP(LR)=CIR(LC,2)
            CCP(LA)=CIR(LC,1)
            AXITBL=L3*DELTX
            IF (AXN-AXITBL) 50,60,70
            CUT=CCP(LR)*(AXN-(LR-1)*DELTX)/DELTX
            CUTA=CCP(LR)*(AXN-(LR-1)*DELTX)/DELTX
            WCC=2
            GO TO 80
          60      WCC=1
          70      CUT=CCP(LR)
            CUTA=CCP(LR)
            CAT=CAT+CUT
            CAT=CATA+CUTA
            IF (XCODE.EQ.1) GO TO 85
            IF (LR.IT.ITBL) GO TO 40
            85      CC4(J)=CAT
            CC4(J)=CATA
            90      CONTINUE
            TBLSCW=ITBL+TBLSCW(LA)

```

```

VORTX66
VORTX67
VORTX68
VORTX69
VORTX70
VORTX71
VORTX72
VORTX73
VORTX74
VORTX75
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VORTX77
VORTX78
VORTX79
VORTX80
VORTX81
VORTX82
VORTX83
VORTX84
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VORTX87
VORTX88
VORTX89
VORTX90
VORTX91
VORTX92
VORTX93
VORTX94
VORTX95
VORTX96
VORTX97
VORTX98
VORTX99
VORTF100
VORTF101
VORTF102
VORTF103
VORTF104
VORTF105
VORTF106
VORTF107
VORTF108
VORTF109
VORTF110
VORTF111
VORTF112
VORTF113
VORTF114

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100 CONTINUE
  II=1
  IC = 0
  IZ = 0
  NST = 0
  NSTI = 0
  DO 150 I=1,IPLAN
    IUZ=NSWSV(I)
    IUX=IUZ+1
    IF (STLOIND(I) .EQ. 1.) IUX = IUZ
    IC = IC + MSV(I)
    ID=IC+1
    IZ = IZ + NSWSV(I)
    YCAT=0.
    IAMB=NMA(I)

C
105 CONTINUE
  DO 140 LA=1,NSCW*IN
    IF (STLOIND(I) .EQ. 1.) GO TO 107
    VC(1)=-PI/2.
    CRJ(1)=0.
    CRJ(1)=0.
    DO 120 J = 1,IUZ
      L=J+1
      IF (STLOIND(I) .EQ. 1.) L = J
      LU = LA + (J-1 + NST) * NSCW*IN
      ALLP(J)=ALOP(LU)
      XC44(J)=XC4(LU)
      CRJ(L)=CCC(LU)
      CRJ(L)=CCU(LU)
      IF (LA.NE.1) GO TO 120
      JJ = J + NST
      ZZH(J)=ZH(JJ)
      SA(J)=SSAWA(JJ)
      PP+I(J)=PPI(JJ)
      YCC(J)=O(IJ)
      II=II+PPLSCW(JJ)
      IF=IIZ-J+1
      ITL=TRLSCW(IZ)
      ID=ID+ITL
      IA=I0+ITL
      IF (IA.GT. IC) YCAT=YCAT-S(ID)
      IF (IA.GT. IC) GO TO 110
      YCAT=YCAT-S(ID)-S(IA)
      IZ=IZ-1
      YB(IE)=YCAT
110 CONTINUE
      DO 130 J0=1,IUZ
        J3=J0+1

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VORTF115
VORTF116
VORTF117
VORTF118
VORTF119
VORTF120
VORTF121
VORTF122
VORTF123
VORTF124
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VORTF126
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VORTF136
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VORTF142
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VORTF144
VORTF145
VORTF146
VORTF147
VORTF148
VORTF149
VORTF150
VORTF151
VORTF152
VORTF153
VORTF154
VORTF155
VORTF156
VORTF157
VORTF158
VORTF159
VORTF160
VORTF161
VORTF162
VORTF163

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130 IF (FTLIND(I) .EQ. 1.) J7 = JP
    YC(J7) = ASIN(YR(JP)/ROT(I))
    CONTINUE
    YOR = NMA(I)*2.*SNN-SNN
    DO 140 K=1,IAVM
        K0 = LA + (K-1) + NSTI)*NSCWMIN
        YOR = YOR + DELTA
        YOC = ASIN(YOR/ROT(I))
        CALL FTLP (YOR,YC(KP),+1,IUZ,YB,YOO)
        CALL FTLP (YOR,ALPD(KP),+1,IUZ,YR,ALLP)
        CALL FTLP (YOR,SSA(KP),+1,IUZ,YR,SA)
        CALL FTLP (YOR,AC4(KP),+1,IUZ,YR,XC44)
        CALL FTLP (YOR,Z(KP),+1,IUZ,YB,77H)
        CALL FTLP (YOR,PHI(KP),+1,IUZ,YB,PHI)
        CALL FTLP (YOC,GAM(KP),+1,IUX,YC,CRI)
        CALL FTLP (YOC,GAC(KP),+1,IUX,YC,CRJ)
        IF (YOR.GT.YR(IUZ)) GAM(KP)=CRI(IUX)
        IF (YOR.GT.YR(IUZ)) GAC(KP)=CRJ(IUX)
    CONTINUE
140 NST = NSSWV(I) + NST
    NSTI = NMA(I) + NSTI
    CONTINUE
    CO 151 IALPH=1,26
    CDPAC(IALPH)=CLIFT(IALPH)=CPITCH(IALPH)=CSUCT(IALPH)=0.
    CLAPC(IALPH)=CDALC(IALPH)=CVAUG(IALPH)=0.
151 CONTINUE
    CONST=4.*SNN/SPEF
    C
    C
    CO 1512 J = 1,4
        DO 1510 IALOM = 1,26
            COS(IALPH,J) = C.O
            CLC(IALPH,J) = 0.0
            CPH(IALPH,J) = 0.0
            CST(IALPH,J) = 0.0
1510 CONTINUE
1512 CONTINUE
    C
    C
    REWIND 10
    DO 150 LI=1,NMAX
        IROT = 1
        KIPOT = YPLAN - 1
        DO 1515 JTT = 1,4*IPOT
            IF (LT.GT. NVALSUM(IJTT)) IROT = IROT + 1
1515 CONTINUE
            LA=(LI-1)*NSCWMIN+1
            LA=LI+NSCWMIN
            DO 152 IALPH=1,26

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VORTF164
VORTF165
VORTF166
VORTF167
VORTF168
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VORTF184
VORTF185
VORTF186
VORTF187
VORTF188
VORTF189
VORTF190
VORTJ 1
VORTJ 2
VORTJ 3
VORTJ 4
VORTJ 5
VORTJ 6
VORTJ 7
VORTJ 8
VORTJ 9
VORTJ 10
VORTJ 11
VORTJ 12
VORTF191
VORTF192
VORTF193
VORTF194
VORTF195
VORTF196
VORTF197
VORTF198
VORTF199
VORTF200

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152 CONTINUE
COPAGIT(IALPH)=CLL(IALPH)=CMP(IALPH)=CSS(IALPH)=CLV(IALPH)=0.
AAP=ATAN(ALPPD(LA))
OLYLF=(YC4(LA)+0.25*(XC4(LA)-XC4(LA+1)))*BETA
S4LE=ATAN(SSA(LA))
DO 170 NV=LA,LR
CPT=COS(ATAN(PHII(NV)))
DO 175 IALPH=1,26
ALPHA=(IALPH-6)*2.*PI/180.
CMM=GAM*(LA)*SIN(ALPHA)+GAD(LA)
IF(GAM*.FO.O.) GO TO 153
CUTE(IALPH)=ABS(GAMM)/GAMM
GO TO 154
153 CUTE(IALPH)=1.
154 VELIN(IALPH)=0.
155 CONTINUE
IF (ICOUNT .GT. 1) GO TO 171
DO 170 NV=1,M
XX=XC4(NV)-XC4(NN)
YY(1)=YQ(NV)-YQ(NN)
YY(2)=YQ(NV)+YQ(NN)
ZZ=Z(NV)-Z(NN)
APHI=ATAN(PHII(NN))
DO 170 I=1,2
YY=YY(I)
CALL INFSUR (BUT,FV(I),FW(I),FUI)
APHI=--APHI
CONTINUE
DO 175 IALPH=1,26
ALPHA=(IALPH-6)*2.*PI/180.
GAMM=GAM*(NN)*SIN(ALPHA)+CAD(NN)
VELIN(IALPH)=((FV(1)+FV(2))-((FV(1)+FV(2))*PHII(NV))*GAMM/FPI+VELIN)
1(IALPH)
165 CONTINUE
170 CONTINUE
DO 177 IALPH=1,26
WRITE(20)VELIN(IALPH)
CONTINUE
GO TO 172
171 DO 156 IALPH=1,26
READ(20)VELIN(IALPH)
156 CONTINUE
172 CONTINUE
AAP=ATAN(ALPPD(NV))
DO 175 IALPH=1,26
ALPHA=(IALPH-6)*2.*PI/180.
GAMM=GAM*(NV)*SIN(ALPHA)+GAD(NV)
CIV=(C75*(ALPHA+AAPP)+VELIN(IALPH)*SIN(AAPP))*CPT
CL=(C1*(ALPHA+AAPP)-VELIN(IALPH)*COS(AAPP))*CPT
VORTE201
VORTE202
VORTE203
VORTE204
VORTE205
VORTE206
VORTE207
VORTE208
VORTE209
VORTE210
VORTE211
VORTE212
VORTE213
VORTE214
VORTE215
VORTE216
VORTE217
VORTE218
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VORTE242
VORTE243
VORTE244
VORTE245
VORTE246
VORTE247
VORTE248
VORTE249

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SDW=SIGN(ALPHA+AAPP)
CQV=COS(ALPH4+AAPP)
CQU=TAN(ALPHA+AAP)/COS(SWLF)
COA=L./COS(ALPHA+AAP)*COS(SWLF))
IF(YO(NV).LI.YCHHI(IPOT),OR,YO(NV).GT.YCHLO(IPOT)) GO TO 173
C
LEADING EDGE VORTX FLOW ANALYSIS EMPLOYED HERE
C
CN1=CUV*CQW*CON*CODE(CUTE(IALPH))
CN2=CUV*CV/COS(SWLF)*CODE(IALPH)
CN3=CUV*CW*CD*COD(AAP)*COA*CLXLE*CODE(IALPH)
CN4=CN1*CN2/(COS(ALPHA+AAP)*COS(SWLF))
CSS(IALPH)=COS(IALPH)+GAMM*CN4*2.
CLV(IALPH)=CLV(IALPH)+GAMM*CN2*2.
GO TO 174
C
ATTACHED FLOW ANALYSIS EMPLOYED HERE
C
ONE HUNDRED PERCENT LEADING EDGE SUCTION
C
173 IF(ICOUNT.EQ.1) GO TO 177
CN1=-CUV*CON
CN2=CUV*CON*TAN(ALPHA+AAP)
CN3=CUV*CON*SIN(AAP)*CLXLF/COS(ALPHA+AAP)
GO TO 174
C
ATTACHED FLOW ANALYSIS EMPLOYED HERE
C
ZERO PERCENT LEADING EDGE SUCTION
C
177 CN1=0.
CN2=G.
CN3=G.
C
174 CONTINUE
CORAGIT(IALPH)=CORAGIT(IALPH)+GAMM*(CUV*SNW+CN1)*2.
CLL(IALPH)=CLL(IALPH)+GAMM*(CUV*CON+CN2)*2.
CM(IALPH)=CM(IALPH)+GAMM*(CUV*COS(AAPP)*YC4(NV)*BETA+CN3)*2.
175 CONTINUE
180 CONTINUE
PO 145 JALPH = 1,26
WRITE(IC) CLL(IALPH), CM(IALPH), CORAGIT(IALPH)
IF (ICOUNT .EQ. 3) WRITE(IC) CLV(IALPH)
CLV(IALPH,IPOT) = COS(IALPH,IPOT) + COPAGIT(IALPH) * CONST
CLC(IALPH,IPOT) = CLS(IALPH,IPOT) + CLL(IALPH) * CONST
CPH(IALPH,IPOT) = CPH(IALPH,IPOT) + CM(IALPH) * CONST/CPREF
IF (ICOUNT .EQ. 3)
* CST(IALPH,IPOT) = CST(IALPH,IPOT) + CSS(IALPH) * CONST
185 CONTINUE
190 CONTINUE

```

```

END FILE 10
DO 194 IALPH = 1,26
DO 194 ITT = 1,IPLAN
  CDRAG(IALPH) = CDRAG(IALPH) + CDG(IALPH,ITT)
  CLIFT(IALPH) = CLIFT(IALPH) + CLF(IALPH,ITT)
  CPITCH(IALPH) = CPITCH(IALPH) + CPH(IALPH,ITT)
  IF (ICOUNT.EQ. 3)
    CSUCT(IALPH) = CSUCT(IALPH) + CST(IALPH,ITT)
194 CONTINUE
C
IF(ICOUNT.EQ.1) END FILE 20
IPUT=0
BEGIN 10
IF(ICOUNT.NE.3) GO TO 193
CSUCMIN=1.
DO 192 IJ=1,26
  CSUCMIN=AMIN1(CSUCMIN,CSUCT(IJ))
  IF(CSUCMIN.EQ.CSUCT(IJ)) IALPSV=IJ
192 CONTINUE
193 DO 225 IALP=1,26
  ALPHA=(IALP-6)*2.
  WRITE(6,265) ALPHA
  IF(ICOUNT.NE.3.OP.DISTALF.EQ.0.) GO TO 191
  AAL=ALPHA*PI/180.
  JSIGN=1
  IF(IALP.LE.IALPSV) JSIGN=-1
  TERM=CSUCT(IALP)*CTILDA/DISTALF*JSIGN
  CLAUS(IALP)=TERM*COS(AAL)
  CDALG(IALP)=TERM*SIN(AAL)
  CMAUG(IALP)=TERM*XTILDA/CRFF
191 TPLE=II=0
  LJ=0
  IPUT=IPUT+1
  NSTT = C
  DO 220 I = 1,IPLAN
    IALPM=NMA(I)
    DO 200 J=1,IAMP
      JJ = J + NSTT
      LA = 1 + (J - 1 + NSTT) * NSCWMIN
      DO 195 IALPH=1,26
        READ(10) CLL(IALPH),CMM(IALPH),CDRAGIT(IALPH)
        IF(ICOUNT.EQ.3) READ(10) CLV(IALPH)
195 CONTINUE
        CCC(JJ) = CLL(IPUT)/CAVE
        CCU(JJ) = CMM(IPUT)/(CAVE*CRFF)
        XCC4(JJ) = CDRAGIT(IPUT)/CAVE
        G4R(JJ) = CLV(IPUT)/CAVE
        G4X(JJ)=CCC(JJ)

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VDRTE299
VDRTE300
VDRTE301
VDRTE302
VDRTE303
VDRTE304
VDRTE305
VDRTE306
VDRTE307
VDRTE308
VDRTE309
VDRTE310
VDRTE311
VDRTE312
VDRTE313
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VDRTE340
VDRTE341
VDRTE342
VDRTE343
VDRTE344
VDRTE345
VDRTE346
VDRTE347

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200 X04(J)=XCC(JJ)
    Z(J)=XCC4(JJ)
    S04(J)=GAT(JJ)
    PHI(J)=YQ(LLA)
    CONTINUE
    IU7=NSCW(I)
    GO 210 LBLAIR=1,IU7
    IF(LBLAIR.EQ. 1)WRITE(6,240)NUMBER(I)
    LI=LI+1
    LU=1+TRLE
    YAR=Z(LU)
    II=II+1
    IELF=TRLE+TRLSW(II)
    YOCR=YAR/RO7
    CALL FTLP (YAR,CLLO,+1,IAMM,PHII,GAM)
    CALL FTLP (YAR,CMMO,+1,IAMM,PHII,XC4)
    CALL FTLP (YAR,CDRAGO,+1,IAMM,PHII,Z)
    CALL FTLP (YAR,CLLV,+1,IAMM,PHII,SSA)
    WRITE (6,260) LI,YOCR,CLLO,CDRAGO,CMMO,CLLV
    CONTINUE
210 P-STY = NMA(I) + NSTY
    WRITE(4,271) NUMBER(I), CLF(IALP,I), CDG(IALP,I), CPH(IALP,I)
    IF (ICOUNT.EQ. 3)WRITE(6,285) CST(IALP,I)
    CONTINUE
220
221 LOTS(6,270) CLIFT(IALP),CDPAG(IALP),CPITCH(IALP)
    WRITE(30) CLIFT(IALP),CDPAG(IALP),CPITCH(IALP)
    RE-IND IC
    IF(ICOUNT.EQ.3) WRITE(6,285) CSUCT(IALP)
225 CONTINUE
    IF(ICOUNT.NE.3) GO TO 310
    P-300 IALP=1,24
    WRITE(30)CLAUG(IALP),CDALG(IALP),CMALG(IALP)
300 CONTINUE
310 ICCL-I=ICOUNT+1
    IF(ICOUNT.EQ.2) GO TO 5
    IF (ICOUNT.GT. 2) GO TO 400
    GO 220 ITT=1,IPLAN
    YGLO(ITT) = YINREP(ITT)
    YGHI(ITT) = YOUTEP(ITT)
320 CONTINUE
    GO TO 5
    C
    C
400 CONTINUE
    END FILE 30
    REWIND 30
    C

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VDPTEF348
VDPTEF340
VDPTEF350
VDPTEF351
VDPTEF352
VDPTEF353
VDPTEF354
VDPTEF355
VDPTEF356
VDPTEF357
VDPTEF358
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VDPTEF368
VDPTEF369
VDPTEF370
VDPTEF371
VDPTEF372
VDPTEF373
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VDPTEF375
VDPTEF376
VDPTEF377
VDPTEF378
VDPTEF379
VDPTEF380
VDPTEF381
VDPTEF382
VDPTEF383
VDPTEF384
VDPTEF385
VDPTEF386
VDPTEF387
VDPTEF388
VDPTEF389
VDPTEF390
VDPTEF391
VDPTEF392
VDPTEF393
VDPTEF394
VDPTEF395
VDPTEF396

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C C      VORTF397
C C      VORTF398
C C      VORTF399
C      240 FORMAT(/,37X,22HDISTRIBUTIONS FOR THE ,410,9H PLANFORM ,/)
C      250 FORMAT(1H1////////////////////)
C          1    40X=AEPODYNAMIC CHARACTERISTICS FOR CAMBERED AND TWISTEDVORTF402
C          20 WINGS*/473X*WITH VORTEX LIFT AT VARIOUS ANGLES OF ATTACK*)
C          255 FORMAT(/36X*PLANFORM *12* HAS LEADING EDGE VORTEX FLOW ASSUMED FRVORTF403
C              1PM *,F12.5,* TO *,F12.5*/,45X,
C              2*AND ATTACHED FLOW ELSEWHERE ACROSS THE SPAN*)
C          260 FORMAT(17X,12,10X,5(EQ.5,11X))
C          265 FORMAT(1H1,////,50X*ANGLE OF ATTACK = *F9.5,1X*DEGREES*//60X*SECTIONVORTF409
C              1ONAL CHARACTERISTICS*/14X*STATION*,12X,*2Y/R*, 14X9HCL*C/CAVE,
C              211X9HCO*C/CAVE,4X21H(CM*C**2)/(CAVE*CREFF),4X,17HCL VORT LE*C/CAVF,VORTF410
C              3 /OXYLOHARMUT C.G./)
C      270 FORMAT(/,,50X,21HTOTAL CHARACTERISTICS ,//,20X,
C          $ 4HCL =,F12.5,17X,4HCD =,F12.5,17X,4HCM =,F12.5)
C      271 FORMAT(/,,50X,A6,25H PLANFORM CHARACTERISTICS ,//,20X,
C          $ 4HCL =,F12.5,17X,4HCD =,F12.5,17X,4HCM =,F12.5)
C      275 FORMAT(/,,40X*ZERO PERCENT LEADING EDGE SECTION ASSUMED*)
C      280 FORMAT(/,,36X*ONE HUNDRED PERCENT LEADING EDGE SECTION ASSUMED*)
C      285 FORMAT(/,,50X,11HC SECTION =,F12.5)
C      290 FORMAT(/,,1CX,3CH*** THE VALUE OF -VIC- IS TOO ,
C          $ 39HS SMALL. PLEASE CORRECT AND RESUBMIT. ***)
C C      RETURN
C C      END
C      SUBROUTINE CNLONG
C C          THIS OVERLAY COMPUTES XNUM LOCAL CN VALUES
C C          AND TOTAL CN FOR EACH PLANFORM.
C C          ROBERT GRAY COMPUTER SCIENCES CORP. 1980
C C          COMMON /ALL/ BOT, BOTSV(4), M, BETA, PTEST, QTEST,
C              $ STA(4), TBLSCW(100), YYCP(4),
C              $ O(400), PN(400), PV(400), ALP(400), S(400), PSI(400),
C              $ PHI(100), ZH(100), CP(400), STLOIND(4)
C C          COMMON /TOTTHREE/ CIR(400,2)
C C          COMMON /THREFOR/ CCAV(2,100), CLT, CLNT, NSSW, ALPD
C C          COMMON /ONETHRE/ TWIST(4), CREFF, SREF, CAVE, CLOS, STRUE, AR,
C              $ ARTTRUE, RTCDHT(4), CONFIG(2), NSSWSV(4),
C              $ MSV(4), KBOT, PLAN, IPLAN, MACH,
C              $ ONETHRE3
C              $ ONETHRE4

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ONETHRE6
ONETHRE6
ONETHRE7
ONETHRE8
ONETHRE9
ONETHR10
MAINONE3
MAINONE3
MAINONE4
MAINONE5
MAINONE6
MAINONE6
CCRRDD 2
CCRRDD 3
CNLONG16
CNLONG17
CNLONG18
CNLONG19
CNLONG20
CNLONG21
CNLONG22
CNLONG23
CNLONG24
CNLONG25
CNLONG26
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CNLONG34
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CNLONG36
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CNLONG43
CNLONG44
CNLONG45
CNLONG46
CNLONG47
CNLONG48
CNLONG49
CNLONG50
CNLONG51

$ SSWWA(100), XL(4), XT(4), CLWB, CNCL, CLA(4), BLAIR(100),
$ CLAMAR(4), CLWIN(4), CLWNG(4), XLOCIN,
$ YINNER(4), YOUTER(4)

INTEGER CONFIG

COMMON /MAINONE/ ICODEOF, TOTAL, AAN(4), XS(4), YS(4), KFCTS(4),
$ XREG(25,4), YREG(25,4), AREG(25,4), DIH(25,4), MCD(25,4),
$ XX(25,4), YY(25,4), AS(25,4), TTWD(25,4), MHCD(25,4), AN(4),
$ ZZ(25,4), ITIPCOD, ICAMIST

COMMON/CCRRDD/ TSPAN(4), TSPANA, KBIT, CTILDA, XTILDA, DISTALE

DIMENSION OCP(1,25),YL(25),XXL(25),
$ WK1(49),XOVERL(25),CNL(25),YLOBL2(25),IENDSW(2),
$ RPTS(2),CPX(400)
DATA XNUM /20./
DATA YNUM /10./
DATA IST /1/
DATA CN /0.0/

WRITE(6,805)

FIND X COORDINATES FOR ALREADY COMPUTED
DELTA CP'S

DO 5 I = 1,M
CPX(I) = PN(I) * BETA
5 CONTINUE

COMPUTE CN FOR EACH PLANFORM

DO 1000 IP = 1,IPLAN

WRITE(6,806) IP
WRITE(6,807)

-IST- STARTING POSITION IN ARRAYS
CPX,Q, AND CP FOR EACH PLANFORM
-MAX- NUMBER OF POINTS IN ARRAYS CPX, Q,
AND CP USED BY EACH PLANFORM
-N- NUMBER OF BREAKPOINTS USED TO DEFINE
EACH PLANFORM

IF (IP .GT. 1) IST = IST + MSV(IP - 1)

```

```

C
C
N = AN(IP) + 1
MAX = MSV(IP)
XFORE = -999999
XAFT = 999999
YMIN = 999999
YROOT = -999999

C
C
C      FIND LEADING AND TRAILING POINTS
DO 10 I = 1, N
  IF (XFORE .LT. XX(I,IP)) XFORE = XX(I,IP)
  IF (XAFT .GT. XX(I,IP)) XAFT = XX(I,IP)
  IF (YMIN .GT. YY(I,IP)) YMIN = YY(I,IP)
  IF (YROOT .LT. YY(I,IP)) YROOT = YY(I,IP)
10 CONTINUE

C
C
C      FIND TOTAL LENGTH -RL-
RL = ABS(XFORE - XAFT)

C
C
C      FIND THE X INCREMENT
XINC = RL/XNUM

C
C
C      USE MID POINT OF EACH X SEGMENT
C      FOR EACH LOCAL VALUE OF X
XNEXT = XFORE - (XINC / 2.0)

C
C
C      FOR EACH VALUE OF XXL...
NUMX = XNUM
N = AN(IP)

C
DO 100 I = 1, NUMX
  XXL(I) = XNEXT
  FIND INTERCEPT POINT AT THIS XXL
  NPTS = 0
  DO 30 J = 1, N
    RXMAX = AMAX1(XX(J,IP), XX(J+1,IP))
    RXMIN = AMIN1(XX(J,IP), XX(J+1,IP))
    IF (XXL(I) .LE. RXMIN .OR. XXL(I) .GE. RXMAX) GO TO 30
    PERPENDICULAR TO XXL WILL INTERCEPT
    NPTS = NPTS + 1
  IF (NPTS .LE. 2) GO TO 25
  WRITE (6,900) XXL(I)
  GO TO 95
25
C      FIND Y COORDINATE FOR INTERCEPT
  RYMAX = AMAX1(YY(J,IP), YY(J+1,IP))
  RYMIN = AMIN1(YY(J,IP), YY(J+1,IP))
  A = ABS(RXMIN - RXMAX)
  AP = ABS(XXL(I) - RXMAX)

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 CNLONG99
 CNLONG100

```

      B = ABS(RYMAX - RYMIN)
      BP = (B * AP) / A
      RPTS(NPTS) = RYMAX - BP
30 CONTINUE
      IF(NPTS .EQ. 0) GO TO 90

      SET BL AND COMPUTE Y COORDINATES
      ON NOTCHED PORTION OF THE WING
      YL GOES FROM THE LEADING EDGE TO THE INNER
      EDGE (I.E., FROM YLE TO RYMAX); OTHERWISE,
      FROM YLE TO YROOT.
      RYMAX = YROOT
      YLE = RPTS(1)
      IF (NPTS .EQ. 1) GO TO 35
      RYMIN = AMIN1(RPTS(1),RPTS(2))
      RYMAX = AMAX1(RPTS(1),RPTS(2))
      YLE = RYMIN
      BL = (RYMAX - YLE) * 2
      WRITE(6,800)XXL(I),BL
      YINC = (BL / 2.) / (YNUM - 1)
      NUMY = YNUM
      YNEXT = 0.0
      DO 40 J = 1,NUMY
      YL(J) = YLE + YNEXT
      YNEXT = YNEXT + YINC
40 CONTINUE

      INTERPOLATE FOR NEW DELTA CP'S
      CALL INTERP(CPX(IST),Q(IST),CP(IST),MAX,XXL(I),1,YL,NUMY,
      & DCP,1,ITER)

      DO 45 J = 1,NUMY
      WRITE(6,801) XXL(I),YL(J),DCP(J)
45 CONTINUE

      COMPUTE CNL AT THIS XXL
      = BL/ROT TIMES INTEGRAL FROM 2 * YLE / BL TO
      2 * YROOT(OR INNER EDGE OF WING) / BL OF DELTA CP
      TIMES D(2Y/BL)

      COMPUTE 2Y/BL FOR EACH YL
      DO 50 J = 1,NUMY
      YLORL2(J) = (YL(J) / BL) * 2.
50 CONTINUE

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CNLON101
 CNLON102
 CNLON103
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CNLONI97
CNLONI98
CNLONI99

IENDSW(1) = 1
IENDSW(2) = 1
XLOW = YLOBL2(1)
XUP = YLOBL2(NUMY)
IF (NPTS.EQ. 2) XUP = 2 * RYMAX / BL
SIGMA = 0.0
IW = 0
CALL SUTS(NUMY,YLOBL2,DCP,XLOW,XUP,SIGMA,IENDSW,
$ END,IW,CNL(I),WK1,IER)

C
CNL(I) = CNL(I) * (BL/BOT)
WRITE(6,802) CNL(I)

C
GO TO 95
90 WRITE(6,901) I,XXL(I)

C
95 XOVERL(I) = XXL(I) / RL
XNEXT = XNEXT - XINC
100 CONTINUE

C
      COMPUTE CN =
      LENGTH* (B/2)/SREF * INTEGRAL FROM LEADING X/L
      TO TRAILING X/L OF CNL TIMES D(X/LENGTH)

      XLOW = XAFT / RL
      XUP = XFORE / RL
      SIGMA = 0.0
      IW = 0
      IENDSW(1) = 1
      IENDSW(2) = 1

C
      ROUTINE SUTS REQUIRES THAT VARIABLES
      BE ENTERED IN INCREASING ORDER

      LIM = NUMX / 2
      DO 110 J = 1,LIM
      K = (NUMX + 1) - J
      RKEEP = XOVERL(J)
      XOVERL(J) = XOVERL(K)
      XOVERL(K) = RKEEP

      RKEEP = CNL(J)
      CNL(J) = CNL(K)
      CNL(K) = RKEEP
      110 CONTINUE

C
      CALL SUTS(NUMX,XOVERL,CNL,XLOW,XUP,SIGMA,IENDSW,
$ IEND,IW,TCN,WK1,IER)
      TCN = TCN * RL * BOT/SREF
      CN = TCN + CN

```

```

C      WRITE(6,803) IP,TCN
C      1000 CONTINUE
C
C      WRITE(6,804) CN
C
C      800 FORMAT(5X,F12.5,42X,F12.5)
C      801 FORMAT(5X,F12.5,5X,F12.5,5X,F12.5)
C      802 FORMAT(72X,6HCNL = ,F12.5)
C      803 FORMAT(110,17HCN FOR PLANFORM , I1,3H = ,F12.5)
C      804 FORMAT(110,11HTOTAL CN = ,F12.5)
C      805 FORMAT(111,50X,30HLONGITUDINAL LOAD DISTRIBUTION)
C      806 FORMAT(110,57X,17HPLANFORM NUMBER , I1)
C      807 FORMAT(110,10X,11X,16X,11Y,13X,12HINTERPOLATED,9X,5HBL(X),/,
C          $      44X,8HDELTA CP,/)
C      900 FORMAT(11H ,27HTON MANY INTERCEPTS AT X = , F12.5,
C          $      2X,17HTWO ARE PERMITTED)
C      901 FORMAT(11H ,27HNO INTERCEPT FOR X STATION , I2,
C          $      4HX = ,F12.7)
C
C      9000 CONTINUE
C      END
C      SUBROUTINE INTERP(CPX,Q,CP,MAX,XXL,NX,YL,NUMY,DCP,
C          $      ND,IER)
C
C      THIS SUBROUTINE CALLS ROUTINE -IQHSCV- TO INTERPOLATE
C      VALUES FOR DELTA CP (DCP).
C
C      CPX-- X COORDINATES
C      Q--- Y COORDINATES
C      CP-- DELTA CP VALUES AT CPX,Q
C      MAX-- NUMBER OF POSITIONS TO BE USED IN CPX,Q, AND CP
C      XXL-- ARRAY CONTAINING NEW X COORDINATES
C      NX-- NUMBER OF ELEMENTS IN XXL
C      YL-- ARRAY CONTAINING NEW Y COORDINATES
C      NUMY-- NUMBER OF ELEMENTS IN YL
C      DCP-- OUTPUT ARRAY DIMENSIONED NX BY NY CONTAINING
C          $      INTERPOLATED DELTA CP'S
C      ND-- ROW DIMENSION OF ARRAY DCP
C      IWK-- INTEGER WORK ARRAY OF LENGTH
C          $      311 * MAX + 1 * NUMY
C      IWK-- REAL WORK ARRAY OF LENGTH 6 * MAX
C      IER-- ERROR RETURN
C          $      = 129, MAX IS LESS THAN 4 OR NX OR NUMY
C          $      IS LESS THAN 1
C          $      = 130, DATA POINTS ARE COLINEAR
C          $      = 131, SOME DATA POINTS ARE IDENTICAL

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CNLON199
CNLON200
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CNLON202
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CNLON221
INTERP 2
INTERP 3
INTERP 4
INTERP 5
INTERP 6
INTERP 7
INTERP 8
INTERP 9
INTERP10
INTERP11
INTERP12
INTERP13
INTERP14
INTERP15
INTERP16
INTERP17
INTERP18
INTERP19
INTERP20
INTERP21
INTERP22
INTERP23
INTERP24
INTERP25
INTERP26
INTERP27

```

```

C      DIMENSION CPX(1),Q(1),CP(1),XXL(1),YL(1),DCP(1,NUMY)
C      DIMENSION IWK(12425),WK(2400)
C
C      CALL IQHSCV(CPX,Q,CP,MAX,XXL,NX,YL,NUMY,
C      $          DCP,ND,IWK,WK,IER)
C
C      RETURN
C
C      END
C      SUBROUTINE IQHSCV (XD,YD,ZD,ND,XI,NXI,YI,NYI,ZI,IZI,IWK,WK,IER)
C      INTEGER
C      REAL
C      INTEGER
C      1    IL1,IL2,ITI,IXI,IYI,IZ,JIGOMN,JIGOMX,JIGIMN,
C      2    JIGIMX,JIGP,JNGP,JWIGPO,JWIGP,JWIPL,JWIPT,
C      3    JWIVL,JWIWP,JWNGPO,JWNGP,JWWPD,NDPO,NGPO,NGPI,
C      ITPV
C      INTEGER
C      REAL
C      COMMON /IBCDPT/
C      IER = 0
C      NDPO = ND
C      NXIO = NXI
C      NYIO = NYI
C      IF (NDPO.LT.4) GO TO 30
C      IF (NXIO.LT.1.OF.NYIO.LT.1) GO TO 30
C      IWK(1) = NDPO
C      IWK(3) = NXIO
C      IWK(4) = NYIO
C      JWIPT = 16
C      JWIVL = 6*NDPO+1
C      JWNGPO = JWIVL-1
C      JWIVL = 24*NDPO+1
C      JWIVP = 3C*NDPO+1
C      JWIGPO = 31*NDPO
C      JWPGO = 5*NDPO+1
C      CALL IQMSG (NDPO,XD,YD,NT,IWK(JWIPT),NL,IWK(JWIPL),IWK(JWIVL),
C      1 IWK(JWIWP),WK,IER)
C      IF (IEP.GE.128) GO TO 9000
C      IWK(5) = NT
C      IWK(6) = NL
C      IF (NT.EQ.0) GO TO 9005
C      CALL IQSHM (XD,YD,NT,IWK(JWIPT),NL,IWK(JWIPL),NXIO,NYIO,XI,YI,
C      1 IWK(JWNGPO+1),IWK(JWIGPO+1))
C      CALL IQHSE (NDPO,XD,YD,ZD,NT,IWK(JWIPT),WK,WK(JWWPD))
C      ITPV = 0
C      JIGOMX = 0
C      JIGIMN = NXIO*NYIO+1

```

```

NGNP = NT+2*NL
DO 25 JNGP=1,NNGP
  ITI = JNGP
  IF (JNGP.LE.NT) GO TO 5
  IL1 = (JNGP-NT+1)/2
  IL2 = (JNGP-NT+2)/2
  IF (IL2.GT.NL) IL2 = 1
  ITI = IL1*(NT+NL)+IL2
  JNGP = JNGPO+JNGP
  NGPO = IWK(JWNGP)
  IF (NGPO.EQ.0) GO TO 15
  JIGOMN = JIGOMX+1
  JIGOMX = JIGOMX+NGPO
  DO 10 JIGP=JIGOMN,JIGOMX
    JWIGP = JWIGPO+JIGP
    IZ = IWK(JWIGP)
    IYI = (IZ-1)/NXIO+1
    IXI = IZ-NXIO*(IYI-1)
    CALL IQHSF (XD,YD,ZD,NT,IWK(JWIPT),NL,IWK(JWIPL),WK,ITI,
      XI(IXI),YI(IYI),ZI(IXI,IYI))
    CONTINUE
  10 CONTINUE
  JWNGP = JWNGPO+2*NNGP+1-JNGP
  NGP1 = IWK(JWNGP)
  IF (NGP1.EQ.0) GO TO 25
  JIGLMX = JIGLMN-1
  JIGLMN = JIGLMN-NGP1
  DO 20 JIGP=JIGLMN,JIGLMX
    JWIGP = JWIGPO+JIGP
    IZ = IWK(JWIGP)
    IYI = (IZ-1)/NXIO+1
    IXI = IZ-NXIO*(IYI-1)
    CALL IQHSF (XD,YD,ZD,NT,IWK(JWIPT),NL,IWK(JWIPL),WK,ITI,
      XI(IXI),YI(IYI),ZI(IXI,IYI))
    CONTINUE
  20 CONTINUE
  25 CONTINUE
  GO TO 9005
  30 IER = 129
  9000 CONTINUE
  CALL UERTST (IER,6HIQHSCV)
  9005 RETURN
  END
INTEGER FUNCTION IQHSD (X,Y,I1,I2,I3,I4)
INTEGER
  I1,I2,I3,I4
REAL
  X(1),Y(1)
  IDX
  ALSQ,A2SQ,A3SQ,A4SQ,B1SQ,B2SQ,B3SQ,B4SQ,C1SQ,
  C2SQ,C3SQ,C4SQ,EP5LN,S1SQ,S2SQ,S3SQ,S4SQ,TOL,
  U1,U2,U3,U4,X1,X2,X3,X4,Y1,Y2,Y3,Y4
  1 (C2SQ,C1SQ),(A3SQ,B2SQ),(B3SQ,ALSQ),
  2 EQUIVALENCE

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IQH 40
 IQH 41
 IQH 42
 IQH 43
 IQH 44
 IQH 45
 IQH 46
 IQH 47
 IQH 48
 IQH 49
 IQH 50
 IQH 51
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 IQH 53
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 IQH 82
 IQH 83
 IQH 84
 IQH 85
 IQH 86
 IQH 87
 IQH 88

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      DATA
      TOL = .710543E-14
      EPSLN = TOL*100.0
      X1 = X(I1)
      Y1 = Y(I1)
      X2 = X(I2)
      Y2 = Y(I2)
      X3 = X(I3)
      Y3 = Y(I3)
      X4 = X(I4)
      Y4 = Y(I4)
      IDX = 0
      U3 = (Y2-Y3)*(X1-X3)-(X2-X3)*(Y1-Y3)
      U4 = (Y1-Y4)*(X2-X4)-(X1-X4)*(Y2-Y4)
      IF (U3+U4.LE.O.O) GO TO 5
      U1 = (Y3-Y1)*(X4-X1)-(X3-X1)*(Y4-Y1)
      U2 = (Y4-Y2)*(X3-X2)-(X4-X2)*(Y3-Y2)
      A1S0 = (X1-X3)**2+(Y1-Y3)**2
      B1S0 = (X4-X1)**2+(Y4-Y1)**2
      C1S0 = (X3-X4)**2+(Y3-Y4)**2
      A2S0 = (X2-X4)**2+(Y2-Y4)**2
      B2S0 = (X3-X2)**2+(Y3-Y2)**2
      C3S0 = (X2-X1)**2+(Y2-Y1)**2
      S1S0 = U1*(A1S0+AMAX1(A1S0,B1S0))
      S2S0 = U2*(C2S0+AMAX1(A2S0,B2S0))
      S3S0 = U3*(C3S0+AMAX1(A3S0,B3S0))
      S4S0 = U4*(C4S0+AMAX1(A4S0,B4S0))
      IF ((AMIN1(S3S0,S4S0)-AMIN1(S1S0,S2S0)).GT.EPSLN) IDX = 1
      IQHSD = IDX
      RETURN
END
SUBROUTINE IQHSE (NDP,XD,YD,ZD,NT,IPT,PD,WK)
INTEGER NDP,NT,IPT(1)
REAL XO(1),YO(1),ZO(1),PD(1),WK(1)
INTEGER IDP,IPTI(3),IT,IV,JPDO,JPDMX,JPD,JPTO,JPT,NDPO,
1 NTO
REAL DX1,OYZ,DY1,OY2,DZ1,DZ2,DZX1,DZX2,DZY1,DZY2,
1 EPSLN,TOL,XV(3),VPXX,VPXY,VPX,VPLY,VPY,
2 VPZMN,VPZ,YV(3),ZV(3),ZXV(3),ZYV(3)
TOL=1641400000000000000008/
DATA
TOL = .710543E-14
EPSLN = TOL*100.0
NDPO = NDP
NTO = NT
JPDMX = 5*NDPO
DO 5 JPD=1,JPDMX
PD(JPD) = O.O
5 CONTINUE

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```

      DO 10 IOP=1,NOP
      WK(IDP) = 0.0
10  CONTINUE
      DO 25 IT=1,NT0
      JPT0 = 3*(IT-1)
      DO 15 IV=1,3
      JPT = JPT0+IV
      IDP = IPT(JPT)
      IPTI(IV) = IDP
      XV(IV) = XD(IDP)
      YV(IV) = YD(IDP)
      ZV(IV) = ZD(IDP)
15  CONTINUE
      DX1 = XV(2)-XV(1)
      DY1 = YV(2)-YV(1)
      DZ1 = ZV(2)-ZV(1)
      DX2 = XV(3)-XV(1)
      DY2 = YV(3)-YV(1)
      DZ2 = ZV(3)-ZV(1)
      VPX = DY1*DZ2-DZ1*DY2
      VPY = DZ1*DX2-DX1*DZ2
      VPZ = DX1*DY2-DY1*DX2
      VPZMN = ABS(DX1*DX2+DY1*DY2)+EPSLN
      IF (ABS(VPZ).LE.VPZMN) GO TO 25
      DO 20 IV=1,3
      IDP = IPTI(IV)
      JPDO = 5*(IDP-1)+1
      PD(JPDO) = PD(JPDO)+VPX
      PD(JPDO+1) = PD(JPDO+1)+VPY
      WK(IDP) = WK(IDP)+VPZ
20  CONTINUE
25  CONTINUE
      DO 30 IOP=1,NOP0
      JPDO = 5*(IDP-1)+1
      PD(JPDO) = -PD(JPDO)/WK(IDP)
      PD(JPDO+1) = -PD(JPDO+1)/WK(IDP)
30  CONTINUE
      DO 45 IT=1,NT0
      JPT0 = 3*(IT-1)
      DO 35 IV=1,3
      JPT = JPT0+IV
      IDP = IPT(JPT)
      IPTI(IV) = IDP
      XV(IV) = XD(IDP)
      YV(IV) = YD(IDP)
      JPDO = 5*(IDP-1)+1
      ZXV(IV) = PD(JPDO)
      ZYV(IV) = PD(JPDO+1)
35  CONTINUE

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IQH 138
 IQH 139
 IQH 140
 IQH 141
 IQH 142
 IQH 143
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 IQH 184
 IQH 185
 IQH 186

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DX1 = XV(2)-XV(1)
DY1 = YV(2)-YV(1)
DZX1 = ZXV(2)-ZXV(1)
DZY1 = ZYV(2)-ZYV(1)
DX2 = XV(3)-XV(1)
DY2 = YV(3)-YV(1)
DZX2 = ZXV(3)-ZXV(1)
DZY2 = ZYV(3)-ZYV(1)
VPXX = DY1+DZX2-DZX1+DY2
VPXY = DZX1+DX2-DX1+DZX2
VPYX = DY1+DZY2-DZY1+DY2
VPYY = DZY1+DX2-DX1+DZY2
VPZ = DX1+DY2-DY1+DX2
VPZMN = ABS(DX1+DX2+DY1+DY2)*EPSLN
IF (ABS(VPZ).LE.VPZMN) GO TO 45
DO 40 IV=1,3
  IDP = IPTI(IV)
  JPDO = 5*(IDP-1)+3
  PD(JPDO) = PD(JPDO)+VPXX
  PD(JPDO+1) = PD(JPDO+1)+VPXY+VPYX
  PD(JPDO+2) = PD(JPDO+2)+VPYY
40 CONTINUE
45 CONTINUE
DO 50 IDP=1,NDPO
  JPDO = 5*(IDP-1)+3
  PD(JPDO) = -PD(JPDO)/WK(IDP)
  PD(JPDO+1) = -PD(JPDO+1)/(2.0*WK(IDP))
  PD(JPDO+2) = -PD(JPDO+2)/WK(IDP)
50 CONTINUE
RETURN
END
SUBROUTINE IQHSF (XD,YD,ZD,NT,IPT,NL,IPL,PDD,ITI,XII,YII,ZII)
  INTEGER
  REAL
  INTEGER
  INTEGER
  REAL
  1
  2
  REAL
  1
  2
  3
  4
  COMMON /IBCOPT/
  1
  2
  EQUIVALENCE
  ITO = ITI
  (XD,YD,ZD,NT,IPT,NL,IPL,PDD,ITI,XII,YII,ZII)
  NT,NL,ITI,IPT(1),IPL(1)
  XD(1),YD(1),ZD(1),PDD(1),XII,YII,ZII
  IDP,ILI,IL2,ITO,I,J,IPL,JIPT,JPDD,JPD,KPD,NL
  ITPV
  XO,YO,AP,AP,CP,DP,P00,P10,P20,P30,P40,P50,P5,
  P01,P11,P21,P31,P41,P02,P12,P22,P32,P03,P13,
  P23,P04,P14,P05
  AA,AB,ACT2,AC,ADBC,AD,A,88,8C,8DT2,B,CC,CD,
  CSUV,C,DD,DLT,DX,DY,D,G1,G2,H1,H2,H3,LU,
  LV,X(3),PO,P1,P2,P3,P4,PD(15),THSV,THUV,
  THXU,U,V,Y(3),ZO,ZUU(3),ZUV(3),ZU(3),ZVV(3),
  ZV(3),Z(3)
  XO,YO,AP,AP,CP,DP,P00,P10,P20,P30,P40,P50,
  P01,P11,P21,P31,P41,P02,P12,P22,P32,P03,P13,
  P23,P04,P14,P05,ITPV
  (P5,P50)

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```

NTL = NT*NL
IF (ITO.LE.NTL) GO TO 5
IL1 = ITO/NTL
IL2 = ITO-IL1*NTL
IF (IL1.EQ.IL2) GO TO 30
GO TO 55
5 IF (ITO.EQ.ITPV) GO TO 25
JIPT = 3*(ITO-1)
JPD = 0
DO 15 I=1,3
  JIPT = JIPT+1
  IDP = IPT(JIPT)
  X(I) = XD(IDP)
  Y(I) = YD(IDP)
  Z(I) = ZD(IDP)
  JPDD = 5*(IDP-1)
  DO 10 KPD=1,5
    JPD = JPD+1
    JPDD = JPDD+1
    PD(JPD) = PDD(JPDD)
10 CONTINUE
15 CONTINUE
X0 = X(1)
Y0 = Y(1)
A = X(2)-X0
B = X(3)-X0
C = Y(2)-Y0
D = Y(3)-Y0
AD = A*D
BC = B*C
DLT = AD-BC
AP = D/DLT
BP = -B/DLT
CP = -C/DLT
DP = A/DLT
AA = A*A
ACT2 = 2.0*A*C
CC = C*C
AB = A*B
ADBC = AD+BC
CD = C*D
BB = B*B
BDT2 = 2.0*B*D
DD = D*D
DO 20 I=1,3
  JPD = 5*I
  ZU(I) = A*PD(JPD-4)+C*PD(JPD-3)
  ZV(I) = B*PD(JPD-4)+D*PD(JPD-3)
  ZUU(I) = AA*PD(JPD-2)+ACT2*PD(JPD-1)+CC*PD(JPD)

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 IQH 283
 IQH 284


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      ZUV(1) = AB*PD(JPD-2)+ADBC*PD(JPD-1)+CD*PD(JPD)
      ZVV(1) = 8B*PD(JPD-2)+8DT2*PD(JPD-1)+DD*PD(JPD)
20 CONTINUE
      P00 = Z(1)
      P10 = ZU(1)
      P01 = ZV(1)
      P20 = 0.5*ZUU(1)
      P11 = ZUV(1)
      P02 = 0.5*ZVV(1)
      H1 = Z(2)-P00-P10-P20
      H2 = ZU(2)-P10-ZUU(1)
      H3 = ZUU(2)-ZUV(1)
      P30 = 10.0*H1-4.0*H2+0.5*H3
      P40 = -15.0*H1+7.0*H2-H3
      P50 = 6.0*H1-3.0*H2+0.5*H3
      H1 = Z(3)-P00-P01-P02
      H2 = ZV(3)-P01-ZVV(1)
      H3 = ZVV(3)-ZVV(1)
      P03 = 10.0*H1-4.0*H2+0.5*H3
      P04 = -15.0*H1+7.0*H2-H3
      P05 = 6.0*H1-3.0*H2+0.5*H3
      LU = SORT(AA+CC)
      LV = SORT(BB+DD)
      THXU = ATAN2(C,A)
      THUV = ATAN2(D,B)-THXU
      CSUV = COS(THUV)
      P41 = 5.0*LV*CSUV/LV*P50
      P14 = 5.0*LU*CSUV/LV*P05
      H1 = ZV(2)-P01-P11-P41
      H2 = ZUV(2)-P11-4.0*P41
      P21 = 3.0*H1-H2
      P31 = -2.0*H1+H2
      H1 = ZU(3)-P10-P11-P14
      H2 = ZUV(3)-P11-4.0*P14
      P12 = 3.0*H1-H2
      P13 = -2.0*H1+H2
      THUS = ATAN2(D-C,B-A)-THXU
      THSV = THUV-THUS
      AA = SIN(THSV)/LU
      BB = -COS(THSV)/LV
      CC = SIN(THUS)/LV
      DD = COS(THUS)/LV
      AC = AA*CC
      AD = AA*DD
      BC = BB*CC
      G1 = AA*AC*(3.0*BC+2.0*AD)
      G2 = CC*AC*(3.0*AD+2.0*BC)
      H1 = -AA*AA*AA*(5.0*AA*BB*P50+(4.0*BC+AD)*P41)-CC*CC*CC*(5.0*CC
      I*DD*P05+(4.0*AD+BC)*P14)

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 IQH 332
 IQH 333

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H2 = 0.5*ZVV(2)-P02-P12
H3 = 0.5*ZUU(3)-P20-P21
P22 = (G1*H2+G2*H3-H1)/(G1+G2)
P32 = H2-P22
P23 = H3-P22
ITPV = ITO
25 DX = XII-X0
DY = YII-Y0
U = AP*DX+BP*DY
V = CP*DX+DP*DY
P0 = P00+V*(P01+V*(P02+V*(P03+V*(P04+V*(P05))))
P1 = P10+V*(P11+V*(P12+V*(P13+V*(P14))))
P2 = P20+V*(P21+V*(P22+V*(P23)))
P3 = P30+V*(P31+V*(P32))
P4 = P40+V*(P41)
ZII = P0+U*(P1+U*(P2+U*(P3+U*(P4+U*(P5))))
RETURN
30 IF (ITO.EQ.ITPV) GO TO 50
JIPL = 3*(ILI-1)
JPD = 0
DO 40 I=1,2
  JIPL = JIPL+1
  IOP = IPL(JIPL)
  X(I) = XD(IOP)
  Y(I) = YD(IOP)
  Z(I) = ZD(IOP)
  JPD = 5*(IOP-1)
DO 35 KPD=1,5
  JPD = JPD+1
  JPDD = JPDD+1
  PD(JPD) = PDD(JPDD)
35 CONTINUE
40 CONTINUE
X0 = X(1)
Y0 = Y(1)
A = Y(2)-Y(1)
B = X(2)-X(1)
C = -B
D = A
AD = A*D
BC = B*C
DLT = AD-BC
AP = 0/DLT
BP = -B/DLT
CP = -B
DP = AP
AA = A*A
ACT2 = 2.0*A*C
CC = C*C
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AB = A*B
AD8C = AD+BC
CD = C*D
88 = 8*8
8DT2 = 2.0*B*D
DD = D*D
DD 45 I=1,2
      JPD = 5*I
      ZU(I) = A*PD(JPD-4)+C*PD(JPD-3)
      ZV(I) = 8*PD(JPD-4)+D*PD(JPD-3)
      ZUU(I) = AA*PD(JPD-2)+ACT2*PD(JPD-1)+CC*PD(JPD)
      ZUV(I) = AB*PD(JPD-2)+AD8C*PD(JPD-1)+CD*PD(JPD)
      ZVV(I) = 88*PD(JPD-2)+8DT2*PD(JPD-1)+DD*PD(JPD)
45 CONTINUE
      P00 = Z(1)
      P10 = ZU(1)
      P01 = ZV(1)
      P20 = 0.5*ZUU(1)
      P11 = ZUV(1)
      P02 = 0.5*ZVV(1)
      H1 = Z(2)-P00-P01-P02
      H2 = ZV(2)-P01-ZVV(1)
      H3 = ZVV(2)-ZVV(1)
      P03 = 10.0*H1-4.0*H2+0.5*H3
      P04 = -15.0*H1+7.0*H2-H3
      P05 = 6.0*H1-3.0*H2+0.5*H3
      H1 = ZU(2)-P10-P11
      H2 = ZUV(2)-P11
      P12 = 3.0*H1-H2
      P13 = -2.0*H1+H2
      P21 = 0.0
      P23 = -ZUU(2)+ZUU(1)
      P22 = -1.5*P23
      ITPV = ITO
50 DX = XII-XO
   DY = YII-YO
   U = AP*DX+BP*OY
   V = CP*DX+DP*OY
   P0 = P00+V*(P01+V*(P02+V*(P03+V*(P04+V*(P05))))
   P1 = P10+V*(P11+V*(P12+V*(P13)))
   P2 = P20+V*(P21+V*(P22+V*(P23)))
   ZII = P0+U*(P1+U*P2)
   RETURN
55 IF (ITO.EQ.ITPV) GO TO 65
   JIPL = 3*IL2-2
   IDP = IPL(JIPL)
   XO = XD(IDP)
   YO = YD(IDP)
   ZO = ZD(IDP)

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DSQI = DSQF(X1,Y1,XD(IP2),YD(IP2))
IF (DSQI.EQ.0.0) GO TO 160
IF (DSQI.GE.DSQMN) GO TO 5
DSQMN = DSQI
IPMN1 = IP1
IPMN2 = IP2
5 CONTINUE
10 CONTINUE
XOMP = (XD(IPMN1)+XD(IPMN2))/2.0
YOMP = (YD(IPMN1)+YD(IPMN2))/2.0
JP1 = 2
DO 15 IP1=1,NDPO
IF (IP1.EQ.IPMN1.OR.IP1.EQ.IPMN2) GO TO 15
JP1 = JP1+1
IWP(JP1) = IP1
WK(JP1) = DSQF(XOMP,YOMP,XD(IP1),YD(IP1))
15 CONTINUE
DO 25 JP1=3,NDPM1
DSQMN = WK(JP1)
JPMN = JP1
DO 20 JP2=JP1,NDPO
IF (WK(JP2).GE.DSQMN) GO TO 20
DSQMN = WK(JP2)
JPMN = JP2
20 CONTINUE
ITS = IWP(JP1)
IWP(JP1) = IWP(JPMN)
IWP(JPMN) = ITS
WK(JPMN) = WK(JP1)
25 CONTINUE
X1 = XD(IPMN1)
Y1 = YD(IPMN1)
X2 = XD(IPMN2)
Y2 = YD(IPMN2)
DO 30 JP=3,NDPO
IP = IWP(JP)
SP = SPDT(XD(IP),YD(IP),X1,Y1,X2,Y2)
VP = VPDT(XD(IP),YD(IP),X1,Y1,X2,Y2)
IF (ABS(VP).GT.(ABS(SP)+EPSLN)) GO TO 35
30 CONTINUE
GO TO 165
35 IF (JP.EQ.3) GO TO 45
JPMX = JP
DO 40 JPC=4,JPMX
JP = JPMX+4-JPC
IWP(JP) = IWP(JP-1)
40 CONTINUE
IWP(3) = IP
45 IP1 = IPMN1

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IP2 = IPMN2
IP3 = IMP(3)
IF (VPDT(XD(IP1),YD(IP1),XD(IP2),YD(IP2),XD(IP3),YD(IP3)).GE.0.0)
  GO TO 50
IP1 = IPMN2
IP2 = IPMN1
50 NTO = 1
50 NIT3 = 3
IPT(1) = IP1
IPT(2) = IP2
IPT(3) = IP3
NLO = 3
NLT3 = 9
IPL(1) = IP1
IPL(2) = IP2
IPL(3) = 1
IPL(4) = IP2
IPL(5) = IP3
IPL(6) = 1
IPL(7) = IP3
IPL(8) = IP1
IPL(9) = 1
DO 150 JPL=4,NDPO
  IP1 = IMP(JPL)
  X1 = XD(IP1)
  Y1 = YD(IP1)
  DO 65 IL=1,NLO
    IP2 = IPL(3*IL-2)
    IP3 = IPL(3*IL-1)
    X2 = XD(IP2)
    Y2 = YD(IP2)
    X3 = XD(IP3)
    Y3 = YD(IP3)
    SP = SPDT(X1,Y1,X2,Y2,X3,Y3)
    VP = VPDT(X1,Y1,X2,Y2,X3,Y3)
    IF (IL.NE.1) GO TO 55
    IXVS = 0
    IF (VP.LE.(ABS(SP)*(-EPSLN))) IXVS = 1
    ILIV = 1
    ILVS = 1
    GO TO 65
  55 IXVSPV = IXVS
    IF (VP.GT.(ABS(SP)*(-EPSLN))) GO TO 60
    IXVS = 1
    IF (IXVSPV.EQ.1) GO TO 65
    ILVS = IL
    IF (ILIV.NE.1) GO TO 70
    GO TO 65
  60 IXVS = 0

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        IF (IXVSPV.EQ.0) GO TO 65
        ILIV = IL
        IF (ILVS.NE.1) GO TO 70
        CONTINUE
        IF (ILIV.EQ.1.AND.ILVS.EQ.1) ILVS = NLO
        IF (ILVS.LT.ILIV) ILVS = ILVS+NLO
        IF (ILIV.EQ.1) GO TO 85
        NLSH = ILIV-1
        NLSHT3 = NLSH*3
        DO 75 J1=1,NLSHT3
            J12 = J1+NLT3
            IPL(J12) = IPL(J11)
        CONTINUE
        DO 80 J11=1,NLT3
            J12 = J11+NLSHT3
            IPL(J11) = IPL(J12)
        CONTINUE
        ILVS = ILVS-NLSH
        JWL = 0
        DO 105 IL=ILVS,NLO
            ILT3 = IL*3
            IPL1 = IPL(ILT3-2)
            IPL2 = IPL(ILT3-1)
            IT = IPL(ILT3)
            NTO = NTO+1
            NTT3 = NTT3+3
            IPT(NTT3-2) = IPL2
            IPT(NTT3-1) = IPL1
            IPT(NTT3) = IP1
            IF (IL.NE.ILVS) GO TO 90
            IPL(ILT3-1) = IP1
            IPL(ILT3) = NTO
            IF (IL.NE.NLO) GO TO 95
            NLN = ILVS+1
            NLNT3 = NLN*3
            IPL(NLNT3-2) = IP1
            IPL(NLNT3-1) = IPL(1)
            IPL(NLNT3) = NTO
            ITT3 = IT*3
            IPT1 = IPT(ITT3-2)
            IF (IPT1.NE.IPL1.AND.IPT1.NE.IPL2) GO TO 100
            IPT1 = IPT(ITT3-1)
            IF (IPT1.NE.IPL1.AND.IPT1.NE.IPL2) GO TO 100
            IPT1 = IPT(ITT3)
            100 IF (IQHSUXD.YD,IPL,IPT1,IPL1,IPL2).EQ.0) GO TO 105
            IPT(ITT3-2) = IPT1
            IPT(ITT3-1) = IPL1
            IPT(ITT3) = IP1
            IPT(NTT3-1) = IPT1

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105      IF (IL.EQ.ILVS) IPL(ILT3) = IT
      IF (IL.EQ.NLO.AND.IPL(3).EQ.IT) IPL(3) = NTO
      JWL = JWL+4
      IWL(JWL-3) = IPL1
      IWL(JWL-2) = IPT1
      IWL(JWL-1) = IPT1
      IWL(JWL) = IPL2
      CONTINUE
      NLO = NLN
      NLNT3 = NLNT3
      NLF = JWL/2
      IF (NLF.EQ.0) GO TO 150
      NTT3P3 = NTT3+3
      DO 145 IREP=1,NREP
      DO 135 ILF=1,NLF
      IPL1 = IWL(2*ILF-1)
      IPL2 = IWL(2*ILF)
      NTF = 0
      DO 110 ITT3=3,NTT3+3
      ITT3 = NTT3P3-ITT3R
      IPT1 = IPT(ITT3-2)
      IPT2 = IPT(ITT3-1)
      IPT3 = IPT(ITT3)
      IF (IPL1.NE.IPT1.AND.IPL1.NE.IPT2.AND.IPL1.NE.IPT3) GO
        TO 110
      IF (IPL2.NE.IPT1.AND.IPL2.NE.IPT2.AND.IPL2.NE.IPT3) GO
        TO 110
      NTF = NTF+1
      ITF(NTF) = ITT3/3
      IF (NTF.EQ.2) GO TO 115
      CONTINUE
      IF (NTF.LT.2) GO TO 135
      ITT3 = ITF(1)*3
      IPT11 = IPT(ITT3-2)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3-1)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3)
      IT2T3 = ITF(2)*3
      IPT12 = IPT(ITT3-2)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3-1)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3)
      IF (IOWSD(XD,YD,IPT11,IPT12,IPL1,IPL2).EQ.0) GO TO 135
      IPT(ITT3-2) = IPT11
      IPT(ITT3-1) = IPT12
      IPT(ITT3) = IPL1
      IPT(ITT3-2) = IPT12
110      IF (NTF.LT.2) GO TO 135
115      ITT3 = ITF(1)*3
      IPT11 = IPT(ITT3-2)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3-1)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3)
      IT2T3 = ITF(2)*3
      IPT12 = IPT(ITT3-2)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3-1)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3)
      IF (IOWSD(XD,YD,IPT11,IPT12,IPL1,IPL2).EQ.0) GO TO 135
      IPT(ITT3-2) = IPT11
      IPT(ITT3-1) = IPT12
      IPT(ITT3) = IPL1
      IPT(ITT3-2) = IPT12
120      IF (NTF.LT.2) GO TO 135
125      ITT3 = ITF(1)*3
      IPT11 = IPT(ITT3-2)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3-1)
      IF (IPT11.NE.IPL1.AND.IPT11.NE.IPL2) GO TO 120
      IPT11 = IPT(ITT3)
      IT2T3 = ITF(2)*3
      IPT12 = IPT(ITT3-2)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3-1)
      IF (IPT12.NE.IPL1.AND.IPT12.NE.IPL2) GO TO 125
      IPT12 = IPT(ITT3)
      IF (IOWSD(XD,YD,IPT11,IPT12,IPL1,IPL2).EQ.0) GO TO 135
      IPT(ITT3-2) = IPT11
      IPT(ITT3-1) = IPT12
      IPT(ITT3) = IPL1
      IPT(ITT3-2) = IPT12

```



```

130 IPT(IT2T3-1) = IPTI1
135 IPT(IT2T3) = IPL2
    JWL = JWL+8
    IWL(JWL-7) = IPL1
    IWL(JWL-6) = IPTI1
    IWL(JWL-5) = IPTI1
    IWL(JWL-4) = IPL2
    IWL(JWL-3) = IPL2
    IWL(JWL-2) = IPTI2
    IWL(JWL-1) = IPTI2
    IWL(JWL) = IPL1
    DO 130 JLT3=3,NLT3,3
        IPLJ1 = IPL(JLT3-2)
        IPLJ2 = IPL(JLT3-1)
        IF ((IPLJ1.EQ.IPL1.AND.IPLJ2.EQ.IPTI2).OR.(IPLJ2.EQ.IPTI1
            L1.AND.IPLJ1.EQ.IPTI2)) IPL(JLT3)
            = ITF(1)
        IF ((IPLJ1.EQ.IPL2.AND.IPLJ2.EQ.IPTI1).OR.(IPLJ2.EQ.IPTI1
            L2.AND.IPLJ1.EQ.IPTI1)) IPL(JLT3)
            = ITF(2)
        CONTINUE
    CONTINUE
    NLFC = NLF
    NLF = JWL/2
    IF (NLF.EQ.NLFC) GO TO 150
    JWLIMN = 2*NLFC+1
    NLFT2 = NLF+2
    DO 140 JWL1=JWLIMN,NLFT2
        JWL = JWL1+1-JWLIMN
        IWL(JWL) = IWL(JWL1)
    CONTINUE
    NLF = JWL/2
140 CONTINUE
145 CONTINUE
150 CONTINUE
    DO 155 ITT3=3,NTT3,3
        IP1 = IPT(ITT3-2)
        IP2 = IPT(ITT3-1)
        IP3 = IPT(ITT3)
        IF (VPOT(AD(IP1),YD(IP1),XD(IP2),YD(IP2),XD(IP3),YD(IP3)).GE.0.
            1 0) GO TO 155
        IPT(ITT3-2) = IP2
        IPT(ITT3-1) = IP1
155 CONTINUE
        NT = NTO
        NL = NLO
        RETURN
160 IER = 131
165 IER = 130

```

IQH 677
 IQH 678
 IQH 679
 IQH 680
 IQH 681
 IQH 682
 IQH 683
 IQH 684
 IQH 685
 IQH 686
 IQH 687
 IQH 688
 IQH 689
 IQH 690
 IQH 691
 IQH 692
 IQH 693
 IQH 694
 IQH 695
 IQH 696
 IQH 697
 IQH 698
 IQH 699
 IQH 700
 IQH 701
 IQH 702
 IQH 703
 IQH 704
 IQH 705
 IQH 706
 IQH 707
 IQH 708
 IQH 709
 IQH 710
 IQH 711
 IQH 712
 IQH 713
 IQH 714
 IQH 715
 IQH 716
 IQH 717
 IQH 718
 IQH 719
 IQH 720
 IQH 721
 IQH 722
 IQH 723
 IQH 724
 IQH 725

```

RETURN
END
SUBROUTINE IQHSH (XO,YO,NT,IPT,NL,IPL,NXI,NYI,XI,YI,NGP,IGP)
  INTEGER
  REAL
  INTEGER
  1
  2
  3
  REAL
  1
  REAL
  SPDT(U1,V1,U2,V2,U3,V3) = (U1-U2)*(U3-U2)+(V1-V2)*(V3-V2)
  VPDT(U1,V1,U2,V2,U3,V3) = (U1-U3)*(V2-V3)-(V1-V3)*(U2-U3)
  NTO = NT
  NLO = NL
  NXIO = NXI
  NYIO = NYI
  NXINYI = NXIO*NYIO
  XIMN = AMIN1(XI(1),XI(NXIO))
  XIMX = AMAX1(XI(1),XI(NXIO))
  YIMN = AMIN1(YI(1),YI(NYIO))
  YIMX = AMAX1(YI(1),YI(NYIO))
  JNGPO = 0
  JNGPI = 2*(NTO+2*NLO)+1
  JIGPO = 0
  JIGPI = NXINYI+1
  DO 80 ITO=1,NTO
    NGPO = 0
    NGPI = 0
    ITO3 = ITO*3
    IP1 = IPI(ITOT3-2)
    IP2 = IPI(ITOT3-1)
    IP3 = IPI(ITOT3)
    X1 = XD(IP1)
    Y1 = YD(IP1)
    X2 = XD(IP2)
    Y2 = YD(IP2)
    X3 = XD(IP3)
    Y3 = YD(IP3)
    XMN = AMIN1(X1,X2,X3)
    XMX = AMAX1(X1,X2,X3)
    YMN = AMIN1(Y1,Y2,Y3)
    YMX = AMAX1(Y1,Y2,Y3)
    INSD = 0
    DO 10 IXI=1,NXIO
      IF (XI(IXI).GE.XMN.AND.XI(IXI).LE.XMX) GO TO 5
      IF (INSD.EQ.0) GO TO 10
      IXIMX = IXI-1

```

IQH 726
 IQH 727
 IQH 728
 IQH 729
 IQH 730
 IQH 731
 IQH 732
 IQH 733
 IQH 734
 IQH 735
 IQH 736
 IQH 737
 IQH 738
 IQH 739
 IQH 740
 IQH 741
 IQH 742
 IQH 743
 IQH 744
 IQH 745
 IQH 746
 IQH 747
 IQH 748
 IQH 749
 IQH 750
 IQH 751
 IQH 752
 IQH 753
 IQH 754
 IQH 755
 IQH 756
 IQH 757
 IQH 758
 IQH 759
 IQH 760
 IQH 761
 IQH 762
 IQH 763
 IQH 764
 IQH 765
 IQH 766
 IQH 767
 IQH 768
 IQH 769
 IQH 770
 IQH 771
 IQH 772
 IQH 773
 IQH 774

```

5      GO TO 13
      IF (INSD.EQ.1) GO TO 10
      INSD = 1
      IXIMN = IXI
10     CONTINUE
      IF (INSD.EQ.0) GO TO 75
      IXIMX = NXIO
15     DO 70 IYI=1,NYIO
        YII = YI(IYI)
        IF (YII.LT.YMN.OR.YII.GT.YMX) GO TO 70
        DO 65 IXI=IXIMN,IXIMX
          XII = XI(IXI)
          L = 0
          IF (VPOT(XI,YI,X2,Y2,XII,YII)) 65,20,25
          L = 1
          IF (VPOT(X2,Y2,X3,Y3,XII,YII)) 65,30,35
          L = 1
          IF (VPOT(X3,Y3,X1,Y1,XII,YII)) 65,40,45
          L = 1
          IZI = NXIO*(IYI-1)+IXI
          IF (L.EQ.1) GO TO 50
          NGPO = NGPO+1
          JIGPO = JIGPO+1
          IGP(JIGPO) = IZI
          GO TO 65
50      IF (JIGPI.GT.NXINYI) GO TO 60
          DO 55 JIGPI=JIGPI,NXINYI
            IF (IZI.EQ.IGP(JIGPI)) GO TO 65
            CONTINUE
            NGPI = NGPI+1
            JIGPI = JIGPI+1
            IGP(JIGPI) = IZI
            CONTINUE
65      CONTINUE
70      JNGPO = JNGPO+1
75      NGP(JNGPO) = NGPO
          JNGPI = JNGPI+1
          NGP(JNGPI) = NGPI
80     CONTINUE
          DO 225 ILO=1,NLO
            NGPO = 0
            NGPI = 0
            ILO3 = ILO*3
            IPL = IPL(ILO3-2)
            IP2 = IPL(ILO3-1)
            X1 = XD(IPL)
            Y1 = YD(IPL)
            X2 = XD(IP2)
            Y2 = YD(IP2)

```

IQH 775
 IQH 776
 IQH 777
 IQH 778
 IQH 779
 IQH 780
 IQH 781
 IQH 782
 IQH 783
 IQH 784
 IQH 785
 IQH 786
 IQH 787
 IQH 788
 IQH 789
 IQH 790
 IQH 791
 IQH 792
 IQH 793
 IQH 794
 IQH 795
 IQH 796
 IQH 797
 IQH 798
 IQH 799
 IQH 800
 IQH 801
 IQH 802
 IQH 803
 IQH 804
 IQH 805
 IQH 806
 IQH 807
 IQH 808
 IQH 809
 IQH 810
 IQH 811
 IQH 812
 IQH 813
 IQH 814
 IQH 815
 IQH 816
 IQH 817
 IQH 818
 IQH 819
 IQH 820
 IQH 821
 IQH 822
 IQH 823

```

XMN = XIMN
XMX = XIMX
YMN = YIMN
YMX = YIMX
IF (Y2.GE.Y1) XMN = AMIN1(X1,X2)
IF (Y2.LE.Y1) XMX = AMAX1(X1,X2)
IF (X2.LE.X1) YMN = AMIN1(Y1,Y2)
IF (X2.GE.X1) YMX = AMAX1(Y1,Y2)
INSD = 0
DO 90 IXI=1,NXIO
  IF (X1(IXI).GE.XMN.AND.XI(IXI).LE.XMX) GO TO 85
  IF (INSD.EQ.0) GO TO 90
  IXIMX = IXI-1
  GO TO 95
85  IF (INSD.EQ.1) GO TO 90
  INSD = 1
  IXIMN = IXI
  CONTINUE
90  IF (INSD.EQ.0) GO TO 155
  IXIMX = NXIO
  DO 150 IYI=1,NYIO
    YII = YI(IYI)
    IF (YII.LT.YMN.OR.YII.GT.YMX) GO TO 150
    DO 145 IXI=IXIMN,IXIMX
      XII = XI(IXI)
      L = 0
      IF (VPDT(X1,Y1,X2,Y2,XII,YII)) 105,100,145
      L = 1
      IF (SPDT(X2,Y2,X1,Y1,XII,YII)) 145,110,115
      L = 1
      IF (SPDT(X1,Y1,X2,Y2,XII,YII)) 145,120,125
      L = 1
      IZI = NXIO*(IYI-1)+IXI
      IF (L.EQ.1) GO TO 130
      NGPO = NGPO+1
      JIGPO = JIGPO+1
      IGP(JIGPO) = IZI
      GO TO 145
130  IF (JIGPL.GT.NXINYI) GO TO 140
      DO 135 JIGPI=JIGPL,NXINYI
        IF (IZI.EQ.IGP(JIGPI)) GO TO 145
      CONTINUE
135  NGPI = NGPI+1
140  JIGPI = JIGPI-1
      IGP(JIGPI) = IZI
      CONTINUE
145  CONTINUE
150  JNGPO = JNGPO+1
155  NGP(JNGPO) = NGPO

```

IQH 824
 IQH 825
 IQH 826
 IQH 827
 IQH 828
 IQH 829
 IQH 830
 IQH 831
 IQH 832
 IQH 833
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 IQH 835
 IQH 836
 IQH 837
 IQH 838
 IQH 839
 IQH 840
 IQH 841
 IQH 842
 IQH 843
 IQH 844
 IQH 845
 IQH 846
 IQH 847
 IQH 848
 IQH 849
 IQH 850
 IQH 851
 IQH 852
 IQH 853
 IQH 854
 IQH 855
 IQH 856
 IQH 857
 IQH 858
 IQH 859
 IQH 860
 IQH 861
 IQH 862
 IQH 863
 IQH 864
 IQH 865
 IQH 866
 IQH 867
 IQH 868
 IQH 869
 IQH 870
 IQH 871
 IQH 872

```

JNGP1 = JNGP1-1
NGP(JNGP1) = NGP1
NGPO = 0
NGP1 = 0
ILP1 = MOD(ILO,NLO)+1
ILP1T3 = ILP1+3
IP3 = IPL(ILP1T3-1)
X3 = XD(IP3)
Y3 = YD(IP3)
XMN = XIMN
XMX = XIMX
YMN = YIMN
YMX = YIMX
IF (Y3.GE.Y2.AND.Y2.GE.Y1) XMN = X2
IF (Y3.LE.Y2.AND.Y2.LE.Y1) XMX = X2
IF (X3.LE.X2.AND.X2.LE.X1) YMN = Y2
IF (X3.GE.X2.AND.X2.GE.X1) YMX = Y2
INSD = 0
DO 165 IXI=1,NXIO
  IF (XI(IXI).GE.XMN.AND.XI(IXI).LE.XMX) GO TO 160
  IF (INSD.EQ.0) GO TO 165
  IXIMX = IXI-1
  GO TO 170
  IF (INSD.EQ.1) GO TO 165
  INSD = 1
  IXIMN = IXI
  CONTINUE
  IF (INSD.EQ.0) GO TO 220
  IXIMX = NXIO
  DO 215 IYI=1,NYIO
    YII = YI(IYI)
    IF (YII.LT.YMN.OR.YII.GT.YMX) GO TO 215
    DO 210 IXI=IXIMN,IXIMX
      XII = XI(IXI)
      L = 0
      IF (SPOT(XI,YI,X2,Y2,XII,YII)) 180,175,210
      L = 1
      IF (SPOT(X3,Y3,X2,Y2,XII,YII)) 190,185,210
      L = 1
      IZI = NXIO*(IYI-1)+IXI
      IF (L.EQ.1) GO TO 195
      NGPO = NGPO+1
      JIGPO = JIGPO+1
      IGP(JIGPO) = IZI
      GO TO 210
    IF (JIGP1.GT.NXINYI) GO TO 205
    DO 200 JIGP1=JIGP1,NXINYI
      IF (IZI.EQ.IGP(JIGP1)) GO TO 210
    CONTINUE
  160
  165
  170
  175
  180
  185
  190
  195
  200

```

IOH 873
 IOH 874
 IOH 875
 IOH 876
 IOH 877
 IOH 878
 IOH 879
 IOH 880
 IOH 881
 IOH 882
 IOH 883
 IOH 884
 IOH 885
 IOH 886
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 IOH 899
 IOH 900
 IOH 901
 IOH 902
 IOH 903
 IOH 904
 IOH 905
 IOH 906
 IOH 907
 IOH 908
 IOH 909
 IOH 910
 IOH 911
 IOH 912
 IOH 913
 IOH 914
 IOH 915
 IOH 916
 IOH 917
 IOH 918
 IOH 919
 IOH 920
 IOH 921

```

205      NGP1 = NGP1+1
      JIGP1 = JIGP1-1
      IGP(JIGP1) = IZI
      CONTINUE
210      CONTINUE
215      JNGPO = JNGPO+1
220      NGP(JNGPO) = NGPO
      JNGP1 = JNGP1-1
      NGP(JNGP1) = NGP1
225      CONTINUE
      RETURN
END
SUBROUTINE UERIST (IER,NAME)
  INTEGER IER,NAME
  INTEGER NAMESET,NAMEQ
  DATA NAMESET/6HURSET/
  DATA NAMEQ/6H
  DATA LEVEL/4/,IFQDF/0/,IEQ/1H=/
  IF (IER.GT.999) GO TO 25
  IF (IER.LT.-32) GO TO 55
  IF (IER.LE.128) GO TO 5
  IF (LEVEL.LT.1) GO TO 30
  CALL UGETIO(1,NIN,IOUNIT)
  IF (IEQDF.EQ.1) WRITE(IOUNIT,35) IER,NAMEQ,IEQ,NAME
  IF (IEQDF.EQ.0) WRITE(IOUNIT,35) IER,NAME
  GO TO 30
5  IF (IER.LE.64) GO TO 10
  IF (LEVEL.LT.2) GO TO 30
  CALL UGETIO(1,NIN,IOUNIT)
  IF (IEQDF.EQ.1) WRITE(IOUNIT,40) IER,NAMEQ,IEQ,NAME
  IF (IEQDF.EQ.0) WRITE(IOUNIT,40) IER,NAME
  GO TO 30
10 IF (IER.LE.32) GO TO 15
  IF (LEVEL.LT.3) GO TO 30
  CALL UGETIO(1,NIN,IOUNIT)
  IF (IEQDF.EQ.1) WRITE(IOUNIT,45) IER,NAMEQ,IEQ,NAME
  IF (IEQDF.EQ.0) WRITE(IOUNIT,45) IER,NAME
  GO TO 30
15 CONTINUE
  IF (NAME.NE.NAMESET) GO TO 25
  LEVOLD = LFVEL
  LEVEL = IER
  IER = LEVOLD
  IF (LEVEL.LT.0) LEVEL = 4
  IF (LEVEL.GT.4) LEVEL = 4
  GO TO 30
25 CONTINUE
  IF (LEVEL.LT.4) GO TO 30
  CALL UGETIO(1,NIN,IOUNIT)

```

IQH 922
 IQH 923
 IQH 924
 IQH 925
 IQH 926
 IQH 927
 IQH 928
 IQH 929
 IQH 930
 IQH 931
 IQH 932
 IQH 933
 IQH 934
 IQH 935
 IQH 936
 IQH 937
 IQH 938
 IQH 939
 IQH 940
 IQH 941
 IQH 942
 IQH 943
 IQH 944
 IQH 945
 IQH 946
 IQH 947
 IQH 948
 IQH 949
 IQH 950
 IQH 951
 IQH 952
 IQH 953
 IQH 954
 IQH 955
 IQH 956
 IQH 957
 IQH 958
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 IQH 960
 IQH 961
 IQH 962
 IQH 963
 IQH 964
 IQH 965
 IQH 966
 IQH 967
 IQH 968
 IQH 969
 IQH 970

```

IF (IEQDF.EQ.1) WRITE(IOUNIT,50) IER,NAMEQ,IEQ,NAME
IF (IEQDF.EQ.0) WRITE(IOUNIT,50) IER,NAME
30 IEQDF = 0
RETURN
35 FORMAT(19H *** TERMINAL ERROR,10X,7H(IER = ,I3,
& 15H) FROM ROUTINE ,A6,A1,A6)
40 FORMAT(36H *** WARNING WITH FIX ERROR (IER = ,I3,
& 15H) FROM ROUTINE ,A6,A1,A6)
45 FORMAT(18H *** WARNING ERROR,11X,7H(IER = ,I3,
& 15H) FROM ROUTINE ,A6,A1,A6)
50 FORMAT(20H *** UNDEFINED ERROR,9X,7H(IER = ,I5,
& 15H) FROM ROUTINE, A6,A1,A6)
55 IEQDF = 1
NAMEQ = NAME
65 RETURN
END
SUBROUTINE UGETIO(IOPT,NIN,NOUT)
INTEGER IOPT,NIN,NOUT
INTEGER NIND,NOUTD
DATA NIND/5LINPUT/,NOUTD/6LOUTPUT/
IF (IOPT.EQ.3) GO TO 10
IF (IOPT.EQ.2) GO TO 5
IF (IOPT.NE.1) GO TO 9005
NIN = NIND
NOUT = NOUTD
GO TO 9005
5 NIND = NIN
GO TO 9005
10 NOUTD = NOUT
9005 RETURN
END
SUBROUTINE SUTS(N,X,Y,XL,XU,SIGMA,IENDSW,END,IW,PROXIN,WK,IERR)
D1.11
C*****
C*
C*
C* PURPOSE - TO COMPUTE THE INTEGRAL OF A CUBIC SPLINE
C* INTERPOLATION UNDER TENSION TO A SET OF
C* DISCRETE NODES OF INDEPENDENT AND DEPENDENT
C* REAL VARIABLES.
C*
C* USE - CALL SUTS(N,X,Y,XL,XU,SIGMA,IENDSW,END,IW,
C* PROXIN,WK,IERR)
C*
C* PARAMETERS N AN INPUT INTEGER SPECIFYING THE NUMBER OF
C* OF NODES OR VALUES FOR BOTH THE INPUT
C* INDEPENDENT AND DEPENDENT VARIABLES. N > 1.
C* X A ONE-DIMENSIONAL INPUT REAL ARRAY OF LENGTH

```



```

C*          SUTDER (04.3), OR SUTS (01.11).          *SUTS 69
C*          OUTPUT - THE FIRST N PLACES OF WK CONTAIN *SUTS 70
C*          DERIVATIVE INFORMATION NECESSARY ON *SUTS 71
C*          SUCCEEDING CALLS TO STIUNI, *SUTS 72
C*          SUTDER, OR SUTS USING THE SAME *SUTS 73
C*          INPUT DATA. *SUTS 74
C*          WORK - THE REST OF WK IS WORK STORAGE. *SUTS 75
C*          IERR - AN OUTPUT INTEGER ERROR CODE. *SUTS 76
C*          * 0 NORMAL RETURN. *SUTS 77
C*          * 1 N < 2. *SUTS 78
C*          * 2 X IS NOT AN INCREASING ARRAY. *SUTS 79
C*          *SUTS 80
C*          PRECISION *SUTS 81
C*          *SUTS 82
C*          REQUIRED ROUTINES - CEEZ,CURVI,CURVIN,CURV12,INTRVL,SNHCSH,TERMS. *SUTS 83
C*          *SUTS 84
C*          DATE RELEASED - MARCH 1, 1979. *SUTS 85
C*          *SUTS 86
C*          LANGUAGE - FORTRAN *SUTS 87
C*          *SUTS 88
C*          LATEST REVISION - NONE *SUTS 89
C*          *SUTS 90
C*          *SUTS 91
C*          *SUTS 92
C*          *****
C          FORMAL PARAMETERS
C          INTEGER IENDSW(2),IERR,IM,N
C          REAL END(2),PROXIN,SIGMA,WK(1),X(1),XL,XU,Y(1)
C          INTERNAL VARIABLES
C          REAL CURVI,DELX,SIGMAP,ZERO
C          CONSTRUCT THE CONSTANT ZERO
C          DATA ZERO/O.OEO/
C          DATA IUNISW /O/
C          ISLPSW=((IENDSW(2)+2*IENDSW(1))
C          IUNISW=1
C          INITIALIZE IERR AND CHECK ERROR RETURNS
C          IERR = 1
C          IF (N .LT. 2) GO TO 9000
C          IERR = 2
C          DELX = X(N) - X(1)
C          IF (DELX .LE. ZERO) GO TO 9000

```

```

C* SUTS 118
C* SUTS 119
C* SUTS 120
C* SUTS 121
C* SUTS 122
C* SUTS 123
C* SUTS 124
C* SUTS 125
C* SUTS 126
C* SUTS 127
C* SUTS 128
C* SUTS 129
C* SUTS 130
C* SUTS 131
C* SUTS 132
C* SUTS 133
C* SUTS 134
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C* SUTS 136
C* SUTS 137
C* SUTS 138
C* SUTS 139
C* SUTS 140
C* SUTS 141
C* SUTS 142
C* SUTS 143
C* SUTS 144
C* SUTS 145
C* SUTS 146
C* SUTS 147
C* SUTS 148
C* SUTS 149
C* SUTS 150
C* SUTS 151
C* SUTS 152
C* SUTS 153
C* SUTS 154
C* SUTS 155
C* SUTS 156
C* SUTS 157
C* SUTS 158
C* SUTS 159
C* SUTS 160
C* SUTS 161
C* SUTS 162
C* SUTS 163
C* SUTS 164
C* SUTS 165
C* SUTS 166

      IERR = 0
      DENORMALIZE THE TENSION FACTOR  SIGMA
      SIGMAP = ABS(SIGMA) * (N - 1)/DELX
      CALL CURV1 IF NECESSARY
      IF (IW .EQ. 1) GO TO 10
      IF (N .GT. 2) CALL CURVIN(N,X,Y,IENDSW,END,SIGMAP,WK,WK(N+1),
      IERR)
      IF (N .EQ. 2) CALL CURV12(DELX,Y,IENDSW,END,SIGMAP,WK)
      IF (IERR .NE. 0) GO TO 9000
      SET IW = 1 FOR OUTPUT
      IW = 1
      COMPUTE THE INTEGRAL  PROXIN
      10 PROXIN = CURVI(XL,XU,N,X,Y,WK,SIGMAP)
      9000 RETURN
      END
      SUBROUTINE CURVIN(N,X,Y,IENDSW,END,SIGMAP,YP,TEMP,IERR)
C*****
C* PURPOSE
C* - COMPUTE SECOND DERIVATIVES NECESSARY TO BE
C* ABLE TO INTERPOLATE POINTS ON A SPLINE
C* FUNCTION UNDER TENSION PASSING THROUGH AT
C* LEAST THREE NODES.  SEE STIUNI (E1.X).
C* USE
C* - CALL CURVIN(N,X,Y,IENDSW,END,SIGMAP,YP,TEMP,
C* IERR)
C* PARAMETERS  N
C* - AN INPUT INTEGER SPECIFYING THE NUMBER OF
C* NODES.  N MUST BE AT LEAST 3.
C* X
C* - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH*SUTS 158
C* N CONTAINING THE X-COORDINATES OF THE NODES.*SUTS 159
C* Y
C* - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH*SUTS 160
C* N CONTAINING THE Y-COORDINATES OF THE NODES.*SUTS 161
C* IENDSW
C* - AN INPUT ONE-DIMENSIONAL INTEGER ARRAY OF
C* LENGTH 2 SPECIFYING THE OPTIONS CHOSEN AT THE*SUTS 163
C* TWO ENDPPOINTS OF THE SPLINE CURVE.  IENDSW(1)*SUTS 164
C* REFERS TO (X(1),Y(1)) AND IENDSW(2)
C* REFERS TO (X(N),Y(N)) .
C*

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C*      IENDSW(I) = 0  THE USER HAS INPUT VALUES FOR *SUTS 167
C*      Y, AT THE APPROPRIATE ENDPOINT *SUTS 168
C*      IN END(I) . *SUTS 169
C*      = 1  Y, WILL BE ESTIMATED *SUTS 170
C*      INTERNALLY AT THE APPROPRIATE *SUTS 171
C*      ENDPOINT. *SUTS 172
C*      = 2  THE NATURAL CONDITION (Y* = 0) *SUTS 173
C*      WILL BE IMPOSED ON THE *SUTS 174
C*      APPROPRIATE ENDPOINT. *SUTS 175
C*      - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH *SUTS 176
C*      2. END(I) CONTAINS THE DERIVATIVE OF THE *SUTS 177
C*      SPLINE FUNCTION AT (X(1),Y(1)) . END(2) *SUTS 178
C*      CONTAINS THE DERIVATIVE OF THE SPLINE *SUTS 179
C*      FUNCTION AT (X(N),Y(N)) . IF IENDSW(I) IS *SUTS 180
C*      NON-ZERO, THEN END(I) NEED NOT BE DEFINED. *SUTS 181
C*      IF BOTH IENDSW(1) AND IENDSW(2) ARE NON- *SUTS 182
C*      ZERO, THEN END MAY BE A DUMMY PARAMETER. *SUTS 183
C*      SIGNAP - AN INPUT REAL NUMBER SPECIFYING THE *SUTS 184
C*      DENORMALIZED TENSION FACTOR. *SUTS 185
C*      YP - AN OUTPUT ONE-DIMENSIONAL REAL ARRAY OF *SUTS 186
C*      LENGTH N CONTAINING SECOND DERIVATIVE *SUTS 187
C*      INFORMATION NECESSARY TO INTERPOLATE THE *SUTS 188
C*      SPLINE FUNCTION. *SUTS 189
C*      TEMP - A WORK ONE-DIMENSIONAL REAL ARRAY OF LENGTH *SUTS 190
C*      N - 1. *SUTS 191
C*      IERR - AN OUTPUT INTEGER SPECIFYING THE ERROR *SUTS 192
C*      RETURN CODE. *SUTS 193
C*      = 0  NORMAL RETURN. *SUTS 194
C*      = 2  TWO CONSECUTIVE X-VALUES ARE THE SAME. *SUTS 195
C*      *SUTS 196
C*      PRECISION - SINGLE. *SUTS 197
C*      *SUTS 198
C*      REQUIRED ROUTINES - CEEZ,SNHCSH,TERMS. *SUTS 199
C*      *SUTS 200
C*      DATE RELEASED - MARCH 1, 1979. *SUTS 201
C*      *SUTS 202
C*      LANGUAGE - FORTRAN. *SUTS 203
C*      *SUTS 204
C*      SOURCE - A. K. CLINE AND R. J. RENKA *SUTS 205
C*      UNIVERSITY OF TEXAS AT AUSTIN *SUTS 206
C*      NON-NATURAL OPTIONS ONLY *SUTS 207
C*      *SUTS 208
C*      LATEST REVISION - NONE. *SUTS 209
C*      *SUTS 210
C*      *SUTS 211
C***** *SUTS 212
C      *SUTS 213
C      *SUTS 214
C      *SUTS 215
C      FORMAL PARAMETERS

```

```

C      INTEGER IENDSW(2),IERR,N
C      REAL END(2),SIGMAP,TEMP(1),X(N),Y(N),YP(N)
C
C      INTERNAL VARIABLES
C
C      INTEGER I,IBAK,NM1,NP1
C
C      REAL C1,C2,C3,DELXN,DELXNM,DELX1,DELX2,DIAG,DIAG1,DIAG2,DX1,DX2
C      REAL SDIAG1,SDIAG2,SLPN,SLP1,ZERO
C
C      DEFINE THE CONSTANT ZERO
C
C      DATA ZERO/0.0E0/
C
C      INITIALIZE VARIABLES
C
C      NM1 = N - 1
C      NP1 = N + 1
C      DELX1 = X(2) - X(1)
C      IERR = 2
C
C      APPROXIMATE THE LEFT END SLOPE IF NEEDED
C
C      IF (DELX1 .LE. ZERO) GO TO 9000
C      IF (IENDSW(1) .EQ. 0) SLP1 = END(1)
C      IF (IENDSW(1) .NE. 1) GO TO 10
C      DELX2 = X(3) - X(1)
C      IF (DELX2 .LE. DELX1) GO TO 9000
C      CALL CEEZ (DELX1,DELX2,SIGMAP,C1,C2,C3,N)
C      SLP1 = C1*Y(1) + C2*Y(2) + C3*Y(3)
C
C      APPROXIMATE THE RIGHT END SLOPE IF NEEDED
C
C      10 IF (IENDSW(2) .EQ. 0) SLPN = END(2)
C      IF (IENDSW(2) .NE. 1) GO TO 20
C      DELXN = X(N) - X(NM1)
C      DELXNM = X(N) - X(N-2)
C      IF (DELXN .LE. ZERO .OR. DELXNM .LE. DELXN) GO TO 9000
C      CALL CEEZ (-DELXN,-DELXNM,SIGMAP,C1,C2,C3,N)
C      SLPN = C1*Y(N) + C2*Y(NM1) + C3*Y(N-2)
C
C      SET UP RIGHT HAND SIDE AND TRIANGONAL SYSTEM FOR YP AND
C      PERFORM FORWARD ELIMINATION
C
C      20 DX1 = (Y(2) - Y(1)) / DELX1
C      CALL TERMS (DIAG1,SDIAG1,SIGMAP,DELX1)
C
C      TEST WHETHER THE LEFT ENDPOINT IS NATURAL

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SUTS 216
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 SUTS 262
 SUTS 263
 SUTS 264


```

80 CONTINUE
IERR = 0
9000 RETURN
C
END
SUBROUTINE CURV12(DX,Y,IENDSW,END,SIGMAP,YP)
C*****
C* PURPOSE
C* - COMPUTE SECOND DERIVATIVES NECESSARY TO BE
C*   ABLE TO INTERPOLATE POINTS ON A SPLINE
C*   FUNCTION UNDER TENSION PASSING THROUGH
C*   EXACTLY TWO NODES. SEE STIUNI (E1.X).
C*
C* USE
C* - CALL CURV12(DX,Y,IENDSW,END,SIGMAP,YP)
C*
C* PARAMETERS
C* DX      - AN INPUT REAL NUMBER = X(2) - X(1) WHERE
C*           THE X'S ARE THE X-COORDINATES OF THE NODES.
C* Y        - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH
C*           2 CONTAINING THE Y-COORDINATES OF THE NODES.
C* IENDSW   - AN INPUT ONE-DIMENSIONAL INTEGER ARRAY OF
C*           LENGTH 2 SPECIFYING THE OPTIONS CHOSEN AT THE
C*           TWO ENDPOINTS OF THE SPLINE CURVE.
C* IENDSW(I) = 0 THE USER HAS INPUT VALUES FOR
C*              (X(I),Y(I)) IN END(I).
C*              = 1 (X(I),Y(I)) WILL BE
C*                  ESTIMATED INTERNALLY.
C*              = 2 THE NATURAL CONDITION
C*                  ( (X(I),Y(I)) = 0 ) WILL BE
C*                  IMPOSED.
C*
C* END
C* - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH
C* 2. END(I) CONTAINS THE DERIVATIVE OF THE
C*   SPLINE FUNCTION AT (X(I),Y(I)). IF
C*   IENDSW(I) IS NON-ZERO, THEN END(I) NEED
C*   NOT BE DEFINED. IF BOTH IENDSW(1) AND
C*   IENDSW(2) ARE NON-ZERO, THEN END MAY BE A
C*   DUMMY PARAMETER.
C*
C* SIGMAP - AN INPUT REAL NUMBER SPECIFYING THE
C*           DENORMALIZED TENSION FACTOR.
C*
C* YP      - AN OUTPUT ONE-DIMENSIONAL REAL ARRAY OF
C*           LENGTH 2 CONTAINING SECOND DERIVATIVE
C*           INFORMATION NECESSARY TO INTERPOLATE THE
C*           SPLINE FUNCTION.
C*
C* PRECISION
C* - SINGLE.
C*
C* REQUIRED ROUTINES
C* - SNHCSH,TERMS.
C*****
SUTS 314
SUTS 315
SUTS 316
SUTS 317
SUTS 318
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SUTS 362

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C*      DATE RELEASED      - MARCH 1, 1979.
C*
C*      LANGUAGE          - FORTRAN.
C*
C*      SOURCE            - A. K. CLINE AND R. J. RENKA
C*                        UNIVERSITY OF TEXAS AT AUSTIN
C*                        NON-NATURAL OPTIONS ONLY
C*
C*      LATEST REVISION    - NONE.
C*
C*****
C*      FORMAL PARAMETERS
C
C      INTEGER IENDSW(2)
C
C      REAL DX,END(2),SIGMAP,Y(2),YP(2)
C
C      INTERNAL VARIABLES
C
C      REAL DIAG,DIAG1,DIAG1,DX1,ONE,SDIAG1,SLP1,SLPN,TEMP,ZERO
C
C      INITIALIZE CONSTANTS
C
C      DATA ZERO,ONE/0.0E0,1.0E0/
C
C      INITIALIZE YP
C
C      YP(1) = ZERO
C      YP(2) = ZERO
C
C      CHECK TO SEE IF EACH ENDPOINT IS EITHER NATURAL OR
C      HAS A COMPUTER-CONSTRUCTED DERIVATIVE
C
C      IF (IENDSW(1)*IENDSW(2) .NE. 0) GO TO 9000
C
C      AT LEAST ONE ENDPOINT HAS A USER-DEFINED SLOPE
C      COMPUTE SOME NECESSARY CONSTANTS
C
C      DX1 = (Y(2) - Y(1))/DX
C      CALL TEMPS(DIAG1,SDIAG1,SIGMAP,DX)
C      DIAG1 = ONE/DIAG1
C
C      CHECK TO SEE IF THE LEFT ENDPOINT IS NATURAL
C
C      IF (IENDSW(1) .NE. 2) GO TO 10
C      YP(2) = (END(2) - DX1)*DIAG1

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*SUTS 363
 *SUTS 364
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 *SUTS 410
 *SUTS 411


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C*          DENORMALIZED TENSION FACTOR.
C*
C*    PRECISION          - SINGLE.
C*
C*    REQUIRED ROUTINES   - INTRVL,SNHCSH.
C*
C*    DATE RELEASED      - MARCH 1, 1979.
C*
C*    LANGUAGE           - FORTRAN.
C*
C*    SOURCE              - A. K. CLINE AND R. J. RENKA
C*                        UNIVERSITY OF TEXAS AT AUSTIN
C*
C*    LATEST REVISION    - NONE.
C*
C*****
C*    FORMAL PARAMETERS
C*
C*    INTEGER N
C*
C*    REAL SIGMAP,X(N),XL,XU,Y(N),YP(N)
C*
C*    INTERNAL VARIABLES
C*
C*    INTEGER I,IL,ILM1,ILP1,IM1,IU,IUM1
C*
C*    REAL CL1,CL2,CS,CU1,CU2,C1,C2,DELI,DELL1,DELL2,DELS,DELSI,DELU1
C*    REAL DELU2,DEL1,DEL2,DUMMY,HALF,ONE,SIGCUBE,SIGDELS,SIXTH,SS
C*    REAL SSIGN,SUM,THIRD,TWELFTH,T1,T2,XXL,XXU,ZERO
C*
C*    INITIALIZE CONSTANTS
C*
C*    DATA ZERO,ONE,HALF,0.0EO,1.0EO,5.0E-1/
C*    DATA THIRD,SIXTH,TWELFTH/
C*    $1716525252525252538,1715525252525252538,1714525252525252538/
C*    THIRD = 1./3.
C*    SIXTH = 1./6.
C*    TWELFTH = 1./12.
C*
C*    CONSTRUCT THE STATEMENT FUNCTION TERM
C*    TERM COMPUTES COEFFICIENTS FOR ANTI-DERIVATIVES
C*
C*    TERM(CHM1,CHM2,T) = (CHM1 - CHM2 - SIGMAP*T*SS) /
C*    $ (SIGCUBE*(SS + SIGDELS))
C*    SIGCUBE = SIGMAP*SIGMAP*SIGMAP
C*
C*    DETERMINE ACTUAL UPPER AND LOWER LIMITS.

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*SUTS 461
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C      XXL = XL
C      XXU = XU
C      SSIGN = ONE
C      IF (XXL.LT. XXU) GO TO 10
C      XXL = XXU
C      XL = XL
C      SSIGN = -ONE
C      IF (XXU.GT. XXL) GO TO 10
C
C      RETURN ZERO IF XL = XU
C
C      CURVI = ZERO
C      GO TO 9000
C
C      SEARCH FOR THE PROPER SUBINTERVALS
C      X(ILM1) .LE. XXL .LE. X(IL)
C
C      10 ILM1 = INTRVL (XXL,X,N)
C      IL = ILM1 + 1
C
C      X(IUM1) .LE. XXU .LE. X(IU)
C
C      IUM1 = INTRVL (XXU,X,N)
C      IU = IUM1 + 1
C
C      BRANCH IF XXU AND XXL ARE IN THE SAME SUBINTERVAL
C
C      IF (IL.EQ. IU) GO TO 70
C      SUM = ZEPO
C
C      BRANCH IF XXL = X(IL)
C
C      IF (XXL.EQ. X(IL)) GO TO 20
C
C      INTEGRATE FROM XXL TO X(IL)
C      FOR THIS CODE XXL .LT. XX(IL) .LE. XX(IUM1)
C
C      DEL1 = XXL - X(ILM1)
C      DEL2 = X(IL) - XXL
C      DELS = DEL1 + DEL2
C      DELSI = HALF/DELS
C      T1 = (DEL1 + DELS)*DEL2*DELSI
C      T2 = DEL2*DEL2*DELSI
C      SUM = T1*Y(IL) + T2*Y(IUM1)
C      IF (SIGMAP.EQ. ZERO) SUM = SUM - T1*T1*DELS*YP(IL)*SIXTH
C      - T2 * (DEL1*(DEL2 + DELS) + DELS*DELS) * YP(ILM1)*THELFTH
C      IF (SIGMAP.EQ. ZERO) GO TO 20
C
C

```

```

C
C      THIS CODE IS FOR A SPLINE UNDER TENSION
C
C      SIGDELS = SIGMAP*DELS
C      CALL SNHCSDUMMY,C1,SIGMAP*DELL,2)
C      CALL SNHCSDUMMY,C2,SIGMAP*DEL2,2)
C      CALL SNHCSDUMMY,C3,SIGDELS,3)
C      SUM = SUM + TERM(CS,C1,Y1)*YP(IL) + TERM(C2,ZERO,T2)*YP(ILM1)
C
C      BRANCH IF X(IL) = X(IUM1)
C
C      20 IF (IL .EQ. IUM1) GO TO 60
C
C      INTEGRATE OVER THE INTERIOR INTERVALS WITH A LOOP
C      FOR THIS CODE X(IL) < X(IUM1)
C
C      ILP1 = IL + 1
C      IF (SIGMAP .NE. ZERO) GO TO 40
C
C      THIS CODE IS FOR A CUBIC SPLINE
C
C      DO 30 I=ILP1,IUM1
C      IM1 = I - 1
C      DELS = (X(I) - X(IM1))*HALF
C      SUM = SUM + (Y(I) + Y(IM1))*DELS - (YP(I) + YP(IM1))*
C      DELS*DELS*DELS*THIRD
C
C      30 CONTINUE
C      GO TO 60
C
C      THIS CODE IS FOR A SPLINE UNDER TENSION
C
C      40 DO 50 I=ILP1,IUM1
C      IM1 = I - 1
C      DELS = X(I) - X(IM1)
C      SIGDELS = SIGMAP*DELS
C      CALL SNHCSDUMMY,C3,SIGDELS,3)
C      SUM = SUM + (Y(I) + Y(IM1))*DELS*HALF + (YP(I) + YP(IM1))*
C      (CS - SS*SIGDELS*HALF)/(SIGCUBE*(SS + SIGDELS))
C
C      50 CONTINUE
C
C      BRANCH IF X(IU-1) = XXU
C
C      60 IF (XXU .EQ. X(IUM1)) GO TO 80
C
C      INTEGRATE FROM X(IU-1) TO XXU
C      FOR THIS CODE X(IU-1) < XXU
C
C      DEL1 = XXU - X(IUM1)
C      DEL2 = X(IU) - XXU
C      DELS = X(IU) - X(IUM1)

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 SUTS 607


```
C          REAL COSHM1,DEL,DENOM,ONE,SIN,TWO,ZERO
C
C          INITIALIZE CONSTANTS
C
C      DATA ZERO,ONE,TWO/O.OEO,1.OEO,2.OEO/
C
C      TEST WHETHER N = 2
C      IF (N .EQ. 2) GO TO 20
C
C      INITIALIZE THE VARIABLE DEL
C
C      DEL = DEL2 - DEL1
C
C      TEST WHETHER SIGMA = ZERO
C
C      IF (SIGMA .NE. ZERO) GO TO 10
C
C      THIS CODE IS FOR A STANDARD CUBIC SPLINE
C
C          C1 = -(DEL1+DEL2)/(DEL1*DEL2)
C          C2 = DEL2/(DEL1*DEL)
C          C3 = -DEL1/(DEL2*DEL)
C          GO TO 9000
C
C      THIS CODE IS FOR A SPLINE UNDER TENSION
C
C      10    CALL SNHCSH(DUMMY,COSHM1,SIGMA*DEL1,1)
C           CALL SNHCSH(DUMMY,COSH2,SIGMA*DEL2,1)
C           SIN = TWO * SINH(SIGMA*(DEL2+DEL1)/TWO) + SINH(SIGMA*DEL/TWO)
C           DENOM = ONE / (COSHM1*DEL - DEL1*SIN)
C           C1 = SIN * DENOM
C           C2 = -COSH2 * DENOM
C           C3 = COSHM1 * DENOM
C           GO TO 9000
C
C      TWO COEFFICIENTS
C
C      20    C1 = -ONE/DEL1
C           C2 = -C1
C           9000 RETURN
C
C      END
C      SUBROUTINE TERMS(DIAG,SDIAG,SIGMA,DEL)
C*****
C*
C*   - COMPUTE THE DIAGONAL AND SUPERDIAGONAL TERMS
C* OF THE TRIDIAGONAL LINEAR SYSTEM ASSOCIATED
C*
```

```

C*          WITH SPLINE UNDER TENSION INTERPOLATION.          *SUTS 755
C*          *SUTS 756
C*          *SUTS 757
C*          *SUTS 758
C*          *SUTS 759
C*          *SUTS 760
C*          *SUTS 761
C*          *SUTS 762
C*          *SUTS 763
C*          *SUTS 764
C*          *SUTS 765
C*          *SUTS 766
C*          *SUTS 767
C*          *SUTS 768
C*          *SUTS 769
C*          *SUTS 770
C*          *SUTS 771
C*          *SUTS 772
C*          *SUTS 773
C*          *SUTS 774
C*          *SUTS 775
C*          *SUTS 776
C*          *SUTS 777
C*          *SUTS 778
C*          *SUTS 779
C*          *SUTS 780
C*          *SUTS 781
C*          *SUTS 782
C*          *SUTS 783
C*          *SUTS 784
C*          *SUTS 785
C*          *SUTS 786
C*          *SUTS 787
C*          *SUTS 788
C*          *SUTS 789
C*          *SUTS 790
C*          *SUTS 791
C*          *SUTS 792
C*          *SUTS 793
C*          *SUTS 794
C*          *SUTS 795
C*          *SUTS 796
C*          *SUTS 797
C*          *SUTS 798
C*          *SUTS 799
C*          *SUTS 800
C*          *SUTS 801
C*          *SUTS 802
C*          *SUTS 803

C*          USE          - CALL TERMS(DIAG,SDIAG,SIGMA,DEL)
C*          PARAMETERS  DIAG  - AN OUTPUT REAL NUMBER CONTAINING THE DIAGONAL
C*          *SUTS 758          TERM.
C*          *SUTS 759          SDIAG  - AN OUTPUT REAL NUMBER CONTAINING THE
C*          *SUTS 760          SUPERDIAGONAL TERM.
C*          *SUTS 761          SIGMA  - AN INPUT REAL NUMBER SPECIFYING THE TENSION
C*          *SUTS 762          FACTOR.
C*          *SUTS 763          DEL    - AN INPUT REAL NUMBER SPECIFYING THE STEP
C*          *SUTS 764          SIZE.
C*          *SUTS 765          PRECISION  - SINGLE.
C*          *SUTS 766          *SUTS 767
C*          *SUTS 768          *SUTS 769
C*          *SUTS 769          *SUTS 770
C*          *SUTS 770          *SUTS 771
C*          *SUTS 771          *SUTS 772
C*          *SUTS 772          *SUTS 773
C*          *SUTS 773          *SUTS 774
C*          *SUTS 774          *SUTS 775
C*          *SUTS 775          *SUTS 776
C*          *SUTS 776          *SUTS 777
C*          *SUTS 777          *SUTS 778
C*          *SUTS 778          *SUTS 779
C*          *SUTS 779          *SUTS 780
C*          *SUTS 780          *SUTS 781
C*          *SUTS 781          *SUTS 782
C*          *SUTS 782          *SUTS 783
C*          *SUTS 783          *SUTS 784
C*          *SUTS 784          *SUTS 785
C*          *SUTS 785          *SUTS 786
C*          *SUTS 786          *SUTS 787
C*          *SUTS 787          *SUTS 788
C*          *SUTS 788          *SUTS 789
C*          *SUTS 789          *SUTS 790
C*          *SUTS 790          *SUTS 791
C*          *SUTS 791          *SUTS 792
C*          *SUTS 792          *SUTS 793
C*          *SUTS 793          *SUTS 794
C*          *SUTS 794          *SUTS 795
C*          *SUTS 795          *SUTS 796
C*          *SUTS 796          *SUTS 797
C*          *SUTS 797          *SUTS 798
C*          *SUTS 798          *SUTS 799
C*          *SUTS 799          *SUTS 800
C*          *SUTS 800          *SUTS 801
C*          *SUTS 801          *SUTS 802
C*          *SUTS 802          *SUTS 803

C*          FORMAL PARAMETERS
C*          REAL DEL,DIAG,SDIAG,SIGMA
C*          INTERNAL VARIABLES
C*          REAL COSHM,DENOM,SIGDEL,SINHM,SIXTH,THIRO
C*          INITIALIZE CONSTANTS
C*          DATA ZERO,SIXTH,THIRO/
C*          $0.0EO,171525252525252538,171652525252525253B/
C*          SIXTH = 1./6.
C*          TWELFTH = 1./12.
C*          TEST WHETHER SIGMA IS ZERO
C*          IF (SIGMA .NE. ZERO) GO TO 10
C*          THIS CODE IS FOR THE STANDARD CUBIC SPLINE

```

```

DIAG = DEL*THIRD
SDIAG = DEL*SIXTH
GO TO 9000

      THIS CODE IS FOR A SPLINE UNDER NON-ZERO TENSION

C
C
C
      10 SIGDEL = SIGMA*DEL
      CALL SNHCSSH(SINHM,COSHM,SIGDEL,0)
      DENOM = DEL / ((SINHM + SIGDEL) * (SIGDEL + SIGDEL))
      DIAG = DENOM * (SIGDEL*COSHM - SINHM)
      SDIAG = DENOM * SINHM

C
      9000 RETURN
      END
      SUBROUTINE SNHCSSH(SINHM,COSHM,X,ISW)
C
C*****
C*
C*
C* PURPOSE
C*
C* - APPROXIMATE  $\text{SINH}(X) = \text{SINH}(X) - X$ 
C*  $\text{COSH}(X) = \text{COSH}(X) - 1$ 
C* AND
C*  $\text{COSH}(X) = \text{COSH}(X) - 1 - X^2/2$ 
C* WITH RELATIVE ERROR < 3.42E-14
C*
C*
C* USE
C*
C* - CALL SNHCSSH(SINHM,COSHM,X,ISW)
C*
C*
C* PARAMETERS
C* SINHM
C*
C* IF ISW = -1, 0, OR 3. OTHERWISE SINHM IS
C* UNCHANGED UPON RETURN.
C*
C* COSHM
C*
C* - AN OUTPUT REAL NUMBER CONTAINING  $\text{COSH}(X)$ 
C* IF ISW = 0 OR 1 AND CONTAINING  $\text{COSH}(X)$ 
C* ISW = 2 OR 3. OTHERWISE COSHM IS UNCHANGED
C* UPON RETURN.
C*
C* X
C*
C* - AN INPUT REAL NUMBER CONTAINING THE
C* INDEPENDENT VARIABLE.
C*
C* ISW
C*
C* - AN INPUT INTEGER SPECIFYING THE FUNCTION
C* DESIRED.
C*
C* = -1 (ONLY) SINHM IS DESIRED.
C* = 0 SINHM AND COSHM ARE DESIRED.
C* = 1 (ONLY) COSHM IS DESIRED.
C* = 2 (ONLY) COSHM IS DESIRED.
C* = 3 SINHM AND COSHM ARE DESIRED.
C*
C*
C* PRECISION
C*
C* - SINGLE.
C*
C*
C* REQUIRED ROUTINES
C*
C* - NONE.
C*
C*
C* DATE RELEASED
C*
C* - MARCH 1, 1979.
C*
C*****
SUTS 804
SUTS 805
SUTS 806
SUTS 807
SUTS 808
SUTS 809
SUTS 810
SUTS 811
SUTS 812
SUTS 813
SUTS 814
SUTS 815
SUTS 816
SUTS 817
SUTS 818
SUTS 819
SUTS 820
SUTS 821
SUTS 822
SUTS 823
SUTS 824
SUTS 825
SUTS 826
SUTS 827
SUTS 828
SUTS 829
SUTS 830
SUTS 831
SUTS 832
SUTS 833
SUTS 834
SUTS 835
SUTS 836
SUTS 837
SUTS 838
SUTS 839
SUTS 840
SUTS 841
SUTS 842
SUTS 843
SUTS 844
SUTS 845
SUTS 846
SUTS 847
SUTS 848
SUTS 849
SUTS 850
SUTS 851
SUTS 852

```



```

C* LANGUAGE          -  FORTRAN.
C*
C* SOURCE            -  A. K. CLINE AND R. J. RENKA
C*                   UNIVERSITY OF TEXAS AT AUSTIN
C*
C* LATEST REVISION   -  NONE.
C*
C*
C*****
C*                   FORMAL PARAMETERS
C*
C* INTEGER ISW
C*
C* REAL COSHM,SINHM,X
C*
C*                   INTERNAL PARAMETERS
C*
C* REAL AX,CP1,CP2,CP3,CP4,COL,EXPX,SP1,SP2,SP3,SP4,SQL,XS,XX,ZP1,ZP2,SUTS 871
C* REAL ZP3,ZQ1,ZQ2,ZQ3,ZQ4
C*
C* DATA SP4/4.50217693381333E-08/,
C*       SP3/8.95278544216390E-06/,
C*       SP2/8.72048976791502E-04/,
C*       SP1/4.36314556981690E-02/,
C*       SQ1/-6.36854430175110E-03/
C*
C* DATA CP4/1.78419567490190E-07/,
C*       CP3/2.8727722979904E-05/,
C*       CP2/2.15151519902028E-03/,
C*       CP1/7.58181822756256E-02/,
C*       COL/-7.3151510567967E-03/
C*
C* DATA ZP3/5.59297115264720E-07/,
C*       ZP2/1.77943493030894E-04/,
C*       ZP1/1.69900461694792E-02/,
C*       ZQ4/1.33412535492375E-09/,
C*       ZQ3/-5.80858944138663E-07/,
C*       ZQ2/1.27814964403863E-04/,
C*       ZQ1/-1.63532871439181E-02/
C*
C* XX = X
C* AX = ABS(XX)
C* XS = XX*XX
C* IF ((AX .GE. 2.70) .OR. (AX .GE. 1.15 .AND.
C*     ISW .NE. 2)) EXPX = EXP(AX)
C*
C*
C*                   APPROXIMATE SINHM
C*
C* IF (ISW .EQ. 1 .OR. ISW .EQ. 2) GO TO 2
C* IF (AX .GE. 1.15) GO TO 1

```

```

      SINHM = (((SP4*XS+SP3)*XS+SP2)*XS+SP1)*XS+1.)*XS*XX)
      *
      GO TO 2
      /((SQ1*XS+1.)*6.)
1 SINHM = -(((1./EXPX+AX)+AX)-EXPX)/2.
  IF (XX.LT. 0.) SINHM = -SINHM
C
C      APPROXIMATE COSHM
C
2 IF (ISW.NE. 0 .AND. ISW.NE. 1) GO TO 4
  IF (AX.GE. 1.15) GO TO 3
  COSHM = (((CP4*XS+CP3)*XS+CP2)*XS+CP1)*XS+1.)*XS)
  *
  GO TO 4
  /((CQ1*XS+1.)*2.)
3 COSHM = ((1./EXPX-2.)*EXPX)/2.
C
C      APPROXIMATE COSHM
C
4 IF (ISW.LE. 1) RETURN
  IF (AX.GE. 2.70) GO TO 5
  COSHM = (((ZP3*XS+ZP2)*XS+ZP1)*XS+1.)*XS*XS)/((((ZQ4
  *XS+ZQ3)*XS+ZQ2)*XS+ZQ1)*XS+1.)*24.)
  RETURN
5 COSHM = ((1./EXPX-2.)*XS)+EXPX)/2.
  RETURN
FND
FUNCTION INTRVL(T,X,N)
C*****
C*
C*
C* PURPOSE
C* - DETERMINE THE INDEX OF THE INTERVAL
C* (DETERMINED BY A GIVEN INCREASING SEQUENCE)
C* IN WHICH A GIVEN VALUE LIES.
C*
C* USE
C* - I = INTRVL(T,X,N)
C*
C* PARAMETERS T - AN INPUT REAL NUMBER SPECIFYING THE
C* GIVEN VALUE.
C* X - AN INPUT ONE-DIMENSIONAL REAL ARRAY OF LENGTH*
C* N SPECIFYING THE INCREASING SEQUENCE. X
C* MUST BE STRICTLY INCREASING.
C* N - AN INPUT INTEGER SPECIFYING THE LENGTH OF
C* X. N > 1.
C*
C* OUTPUT I - IF T.LE. X(2), I = 1.
C* IF T.GE. X(N-1), I = N - 1.
C* OTHERWISE, X(I) .LE. T .LE. X(I+1)
C*
C* PRECISION - SINGLE.
C*****
SUTS 902
SUTS 903
SUTS 904
SUTS 905
SUTS 906
SUTS 907
SUTS 908
SUTS 909
SUTS 910
SUTS 911
SUTS 912
SUTS 913
SUTS 914
SUTS 915
SUTS 916
SUTS 917
SUTS 918
SUTS 919
SUTS 920
SUTS 921
SUTS 922
SUTS 923
SUTS 924
SUTS 925
SUTS 926
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SUTS 935
SUTS 936
SUTS 937
SUTS 938
SUTS 939
SUTS 940
SUTS 941
SUTS 942
SUTS 943
SUTS 944
SUTS 945
SUTS 946
SUTS 947
SUTS 948
SUTS 949
SUTS 950

```

```
*SUTS 951
*SUTS 952
*SUTS 953
*SUTS 954
*SUTS 955
*SUTS 956
*SUTS 957
*SUTS 958
*SUTS 959
*SUTS 960
*SUTS 961
*****SUTS 962
SUTS 963
SUTS 964
SUTS 965
SUTS 966
SUTS 967
SUTS 968
SUTS 969
SUTS 970
SUTS 971
SUTS 972
SUTS 973
SUTS 974
SUTS 975
SUTS 976
SUTS 977
SUTS 978
SUTS 979
SUTS 980
SUTS 981
SUTS 982
SUTS 983
SUTS 984
SUTS 985
SUTS 986
SUTS 987
SUTS 988
SUTS 989
SUTS 990
SUTS 991
SUTS 992
SUTS 993
SUTS 994
SUTS 995
SUTS 996
SUTS 997
SUTS 998
SUTS 999
```

```
C* REQUIRED ROUTINES - NONE.
C*
C* DATE RELEASED - MARCH 1, 1979.
C*
C* SOURCE - A. K. CLINE AND R. J. RENKA
C* UNIVERSITY OF TEXAS AT AUSTIN
C*
C* LATEST REVISION - NONE.
C*
C*****
C C
C C FORMAL PARAMETERS
C C
C C
C C INTEGER N
C C
C C REAL T,X(N)
C C
C C INTERNAL VARIABLES
C C
C C INTEGER IH,IL
C C
C C REAL TT
C C
C C TT = T
C C IF (TT .LE. X(2)) GO TO 4
C C IF (TT .GE. X(N-1)) GO TO 5
C C IL = 2
C C IH = N-1
C C
C C INTERPOLATE LINEARLY FOR I
C C
C C 1 I = IL+IFIX(FLOAT(IH-IL)*(TT-X(IL))/(X(IH)-X(IL)))
C C IF (TT .LT. X(I)) GO TO 2
C C IF (TT .LE. X(I+1)) GO TO 3
C C
C C I IS TOO SMALL - ADJUST AND TRY AGAIN
C C
C C IL = I+1
C C GO TO 1
C C
C C 2 IH = I
C C GO TO 1
C C
C C 3 INTRVL = I
```

C C I IS JUST RIGHT - RETURN

C C RETURN

C C LEFT END

C C 4 INTRVL = 1
RETURN

C C RIGHT END

C C 5 INTRVL = N-1
RETURN

C END

SUTS1000
SUTS1001
SUTS1002
SUTS1003
SUTS1004
SUTS1005
SUTS1006
SUTS1007
SUTS1008
SUTS1009
SUTS1010
SUTS1011
SUTS1012
SUTS1013
SUTS1014

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*
*      SEGMENTATION DIRECTIVES FOR THE
*      VORTEX LATTICE FORTRAN PROGRAM
*
*      H. E. HERBERT
*      COMPUTER SCIENCES CORPORATION
*      HAMPTON, VA.
*      JULY, 1981
*
*
*      ROOT TREE WINGAL-(GEOMTRY,MATXSOL,AERODYN,CDRAGNF,TIPSUCT,VORTEX,CNLONG
*,)
*
*      INCLUDE  WINGAL,INFSUB,LOADING,FTLUP,READIN
*      GLOBAL   ALL,TOHREE,THREFOR,ONETHRE,MAINONE,CCRRDD,INSUB23
*
*
*      GEOMTRY  INCLUDE  GEOMTRY,PLANPLT
*
*      MATXSOL  INCLUDE  MATXSOL,GIVENS,BLOCKR,TRIANG,SOLVER,BUFFIN
*
*      AERODYN  INCLUDE  AERODYN,FLOWFL,CDICLS,HEAPSRT,SIFT
*
*      CDRAGNF  INCLUDE  CDRAGNF
*
*      TIPSUCT  INCLUDE  TIPSUCT,WRTANS
*
*      VORTEX   INCLUDE  VORTEX
*
*      CNLONG   INCLUDE  CNLONG,INTERP,IQHSCV,IQHSD,IQHSE,IQHSF,IQHSG
*      CNLONG   INCLUDE  IQHSH,UERTST,UGETIO,SUTS,CURVIN,CURV12,CURVI
*      CNLONG   INCLUDE  CEEZ,TERMS,SNHCSH,INTRVL
*
*
*      END

```

FIGURE 1 - SEGMENTATION DIRECTIVES



1. Report No. NASA TM 83304		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle Production Version of the Extended NASA-Langley Vortex Lattice FORTRAN Computer Program. Vol. II Source Code				5. Report Date April 1982	
				6. Performing Organization Code	
7. Author(s) Henry E. Herbert John E. Lamar				8. Performing Organization Report No. 505-31-43-03	
9. Performing Organization Name and Address Langley Research Center Hampton, VA 23665				10. Work Unit No.	
				11. Contract or Grant No.	
				13. Type of Report and Period Covered	
12. Sponsoring Agency Name and Address National Aeronautics and Space Administration Washington, DC 20546				14. Sponsoring Agency Code	
15. Supplementary Notes Henry E. Herbert, Computer Science Corporation, Hampton, VA 23665 John E. Lamar, Langley Research Center					
16. Abstract This document presents the source code for the latest production version, MARK IV, of the NASA-Langley Vortex Lattice Computer Program. All viable subcritical aerodynamic features of previous versions have been retained. This version extends the previously documented program capabilities to four planforms, 400 panels, and enables the user to obtain vortex-flow aerodynamics on cambered planforms, flow-field properties off the configuration in attached flow, and planform longitudinal load distributions.					
17. Key Words (Suggested by Author(s)) Vortex Lattice Method Segmentation Directives			18. Distribution Statement Unclassified [REDACTED] Subject Category 61		
19. Security Classif. (of this report) Unclassified	20. Security Classif. (of this page) Unclassified	21. No. of Pages 150	22. Price A07		